# ST.ANNE'S COLLEGE OF ENGINEERING AND TECHNOLOGY 

DEPARTMENT OF MECHANICAL<br>ENGINEERING<br>V - SEMESTER (MECHANICAL)<br>THERMAL ENGINEERING LABORATORY - I MANUAL



# THERMAL ENGINEERING LABORATORY IV SEMESTER MECHANICAL ENGINEERING 

## 1. IC ENGINES LAB AND FUELS LABORATORY

## LIST OF EXPERIMENTS

1. STUDY OF IC ENGINES, COMPONENTS AND LOADING DEVICES
2. VALVE TIMING AND PORT TIMING DIAGRAMS
3. PERFORMANCE TEST ON A 4 STROKE DIESEL ENGINE
4. HEAT BALANCE TEST ON A 4 STROKE DIESEL ENGINE.
5. MORSE TEST ON MULTI CYLINDER DIESEL ENGINE.
6. RETARDATION TEST TO FIND FRICTIONAL POWER OF A DIESEL ENGINE
7. DETERMINATION OF VISCOSITY -RED WOOD VISCOMETER
8. DETERMINATION OF FLASH POINT AND FIRE POINT

## 2.STEAM LABORATORY

1.STUDY OF STEAM GENERATORS
2.PERFORMANCE AND ENERGY BALANCE TEST ON STEAM GENERATORS
3.PERFORMANCE AND ENERGY BALANCE TEST ON STEAM TURBINE.

| INDEX |  |  |  |  |
| :---: | :--- | :---: | :---: | :---: |
| S <br> NO | EXPERIMENT | PAGE NO. |  |  |
| INTERNAL COMBUSTION LABORATORY |  |  |  |  |
| 1 | STUDY OF IC ENGINE, COMPONENTS AND LOADING <br> DEVICES |  |  |  |
| 2 | VALVE TIMING DIAGRAM OF A 4 STROKE ENGINE |  |  |  |
| 3 | PORT TIMING DIAGRAM OF A 2 STROKE ENGINE |  |  |  |
| 4 | PERFORMANCE TEST ON A 4 STROKE DIESEL ENGINE |  |  |  |
| 5 | HEAT BALANCE SHEET FOR A 4 STROKE DIESEL <br> ENGINE |  |  |  |
| 6 | MORSE TEST ON MULTICYLINDER DIESEL ENGINE |  |  |  |
| 7 | RETARDATION TEST TO FIND THE FRICTIONAL <br> POWER OF A DIESEL ENGINE |  |  |  |
| 8 | DETERMINATION OF VISCOSITY BY REDWOOD <br> VISCOMETER |  |  |  |
|  | STEAM LABORATORY |  |  |  |
|  | STUDY OF STEAM GENERATORS AND STEAM <br> TURBINES |  |  |  |
| 2 | PERFORMANCE TEST ON STEAM GENERATORS |  |  |  |
| 3 | PERFORMANCE TEST ON STEAM TURBINES |  |  |  |

## EXPERIMENT 1:

## STUDY OF IC ENGINES, COMPONENTS AND LOADING DEVICES

AIM:
To identify and study the different parts of an I.C. Engine and the loading mechanisms used in it.

Apparatus required:
Cut model of an I.C. Engine.
Classifications of an I.C. Engine:
There are a number of ways by which an I.C. Engine is classified. Some of the classifications are given below:

- Based on the type of combustion as:

Spark ignition \& Compression ignition engines

- Based on the no. of power strokes per cycle as:

4 -stroke \& 2-stroke engines

- Based on the mounting of the engine as:

Horizontally mounted \& Vertically mounted engines

- Based on the type cooling as:

Air cooled \& water cooled engines

- Based on the type of operation as:

Reciprocating \& Rotary Engines
Cross sectional view of a Compression Ignition Engine:


FIGURE 1

## Engine Components

Fig1 shows the cross section of an engine. The major components are listed below:

* Cylinder: It is a cylindrical sleeve in to which a close fitting piston reciprocates. The cylinder is supported by the cylinder block.
* Piston: A cylindrical component that is fitted snugly in the cylinder providing an air tight space along with the piston rings. The piston provides the combustion space inside the cylinder.
* Piston rings: These are rings that are fitted in to the slots around the piston. They provide a pressure tight seal between the piston \& the cylinder wall, thus preventing the leakage of gases in to the crank case.
* Gudgeon pin: The pin that connects the piston to the connecting rod.
* Inlet Manifold: The pipe which connects the intake system to the inlet valve of the engine. It provides the passage for air fuel mixture (in SI Engines) or air alone (in C I Engines).
* Exhaust Manifold: The pipe which connects the exhaust valve of the engine to the tail pipe of the automobile. It provides the passage for the exhaust gases to be thrown out of the engine in to the atmosphere.
* Connecting rod: It interconnects the piston and the crank shaft and transmits the gas force from the piston to the crankshaft.
* Crank shaft: It converts the reciprocatory motion of the piston in to the rotary motion of the out put shaft. The crank shaft of the engine is provided with crank arms \& balance weight for static \& dynamic balancing.
* Inlet \& Exhaust Valves: The valves used are mushroom shaped poppet type. They are provided on either side of the cylinder head for regulating the charge coming in to the cylinder and for discharging the products of combustion from the cylinder.
* Cam shaft: The shaft used to control the valve movement. It receives power from the crank shaft through a gear arrangement and a speed ratio of 1:2 is maintained.
* Fly wheel: It is a device which is used to provide a uniform torque to the out put shaft. During the operation of an engine, the net torque produced fluctuates between a minimum and maximum value. This torque cannot be put to proper use, so in order to damn out the fluctuations; an inertia mass in the form of a fly wheel is attached to the out put shaft.
* Cooling System: During the operation of an engine, enormous heat is developed which if not properly dissipated, can lead to thermal cracks along the walls of the engine resulting in its break down. In order to dissipate this heat, the engine is provided with a cooling system. It can be classified in to air cooling \& water cooling systems.

Air cooling: When the load on the engine is very less, the amount of heat produced during the operation is also less. In such cases, air cooling is preferred. Around the engine, extended surfaces called fins (as shown in Fig 2) are fixed. Due to increase in the surface area, the heat transfer rate increases resulting in higher dissipation of heat as the atmospheric air blows past the engine


FIGURE 2
Water cooling: Air cooling can be used only in engines where the amount of heat removed is less. When a large amount of heat has to be dissipated, water cooling is always preferred to air cooling. Here water is used as the heat transfer medium. A pump is used to circulate the water around the engine through the water jackets. A heat exchanger called a radiator is used to transfer heat from the hot water to the cool air. A schematic of the water cooling system is shown in Fig 3.


FIGURE 3

Lubrication System: An I.C, Engine has reciprocating parts in it. Due to frequent reciprocation of piston inside the cylinder, friction takes place which results in the wear and tear of engine parts. In order to prevent friction parts from getting worn out due to friction, lubricant oil is used. The system which circulates the lube oil through out the engine parts is called as lubrication system. The lubrication system can be classified as follows:
$>$ Petroil or Mist lubrication
$>$ Wet sump lubrication

- Splash lubrication
- Splash \& pressure lubrication
- Pressure lubrication
> Dry sump lubrication
A schematic of the Splash Lubrication System is shown in Fig 4.


Figure : 4
General Nomenclatures used to specify an engine:
1 Cylinder bore: The nominal inner diameter of the cylinder. It is expressed in mm.
2 Stroke: The nominal distance through which a piston moves between successive reversals of its direction of motion. It is expressed in mm .

3 Piston area: The area of a circle whose diameter is equal to the cylinder bore. It is expressed in $\mathrm{cm}^{2}$.

4 Dead Centre: The position of the piston at the moment when the direction of the piston motion is reversed at either end of the stroke. There are two dead centres:
a. Top dead centre (TDC): The dead centre when the piston is farthest from the crank shaft. It is also called Inner Dead Centre (IDC).
b. Bottom dead centre (BDC): The dead centre when the piston is nearest to the crank shaft. It is also called Outer Dead Centre (ODC).

5 Swept Volume: The nominal volume swept by the piston when traveling from one dead centre to the other. It is also called displacement. It is expressed in cubic centimeters (cc).

Swept volume $\left(V_{s}\right)=\left(\pi * d^{2} * L\right) / 4$ cc
Where
d is the bore in cm
L is the stroke in cm
6 Clearance Volume $\left(\mathrm{V}_{\mathrm{c}}\right)$ : The nominal volume of the combustion chamber when the piston is at the top dead centre.

7 Compression Ratio (r): Ratio of the total volume when the piston is at BDC and the clearance volume (when the piston is at TDC).

$$
\text { Compression ratio }(\mathrm{r})=\left(\mathrm{V}_{\mathrm{s}}+\mathrm{V}_{\mathrm{c}}\right) / \mathrm{V}_{\mathrm{c}}
$$

## TERMS RELATED WITH IC ENGINES



## 1.BORE

The inside diameter of the cylinder is called bore.
2.STROKE

As the piston reciprocates inside the engine cylinder, it has got limiting upper and lower positions beyond which it cannot move and reversal of motion takes place at these limiting positions.
3.TOP DEAD CENTRE(TDC)

The top most position of the piston towards the cover end side of the cylinder is called top dead centre.In case of horizontal engines, it is known as Inner Dead Centre.

## 4.BOTTOM DEAD CENTRE (BDC)

The lowest position of the piston towards the crank end side of the cylinder is called bottom dead centre.In case of horizontal engines, it is called the outer dead centre.

## 5.CLEARANCE VOLUME

The volume contained in the cylinder above the top of the piston when the piston is at top dead centre, is called the clearance volume.
6. SWEPT VOLUME

The volume swept through by the piston is moving between top dead centre and bottom dead centre, is called swept volume or piston displacement.
7. COMPRESSION RATIO (r)
$r=\frac{V_{s}+V_{c}}{V_{c}}$
Where, $\mathrm{V}_{\mathrm{s}}=$ swept volume, and $\mathrm{V}_{\mathrm{c}}=$ Clearance volume
It is ratio of total cylinder volume to clearance volume.

## 8. PISTON SPEED

The average speed of the piston is called piston speed.
Piston speed, $=2 \mathrm{~L} N$
Where, $\mathrm{L}=$ Length of the stroke and $\mathrm{N}=$ Speed of the engine in rpm.

## LOADING DEVICES

These devices are also known as Dynamometers, which are used to find the Torque of the engine output shaft. After the determination of the torque, and the angular speed, of the engine, the brake power can be measured .

## CLASSIFICATION OF DYNAMOMETER

Dynamometers can be broadly classified as
(i)Absorption Dynamometers

These Dynamometer measure and absorb the power output of the engine to which they are coupled.The power absorbed is usually dissipated as heat by some means. Examples of such dynamometers are prony brake ,rope brake ,hydraulic ,eddy current dynamometers, etc.
(ii) Transmission Dynamometer:

In transmission dynamometers the power is transmitted to the load coupled to the engine after it is indicated on some types of scale. These are also called torque meters.

## PRONY BRAKE

$>$ This is one of the simple methods of measuring power output of an engine.
$>$ Here , an attempt to stop the engine by means of a mechanical brake on the flywheel and measure the weight which an arm attached to the brake will support ,as it tries to rotate with the flywheel. This system is known as the prony brake and form its use the expression brake power has come.

## CONSTRUCITON

$>$ The Prony brake consists of frames with two brake shoes gripping the flywheel.
$>$ The pressure of the brake shoes on the fly wheel can be varied by the spring loaded using nuts on the top of the frame.
> The wooded block when pressed into contact with the rotating drum opposes the engine torque and the power is dissipated in overcoming frictional resistance.
$>$ The power absorbed is converted into heat and hence this type of dynamometer must be cooled.
$>\quad$ The brake power is given by
$\mathrm{BP}=2 \pi \mathrm{NT}$

## ROPE BRAKE DYNAMOMETER

$>$ The rope brake is another simple device for measuring BP of an engine. If consists of a number of turns of rope wound around the rotating drum attached to the output shaft.
$>$ One side of the rope is connected to a spring balance and the other to a loading device.
$>$ The power absorbed is due to friction between the rope and the drum. The drum therefore requires cooling.
$>$ Rope brake is quite cheaper and can be easily fabricated but not very accurate because of changes in the friction coefficient of the rope with temperature.
$>$ The BP is given by
$\mathrm{BP}=\pi \mathrm{DN}(\mathrm{W}-\mathrm{S})$
Where, D is the brake drum diameter
W is the Weight
S is the Spring scale reading

## EXP 2 : VALVE TIMING DIAGRAM OF A 4 STROKE ENGINE

AIM: TO DRAW (i) THE VALVE TIMING DIAGRAM FOR A 4 STROKE ENGINE SHOWING THE RELATIVE CRANK ANGLES CORRESPONDING TO OPENING AND CLOSING OF INLET AND EXHAUST VALVES

APPARATUS:
CUT SECTION MODEL OF 4 STROKE
ENGINE,SCALE,CHALK

FORMULA USED:
Distance on Flywheel from
the nearest dead centre


Circumference of Flywheel

## PROCEDURE

The circumference of the flywheel is measured using the thread and scale.
$>$ The flywheel is measured in clockwise direction.
$>$ The markings corresponding to the opening and closing of inlet and exhaust valves are made on the flywheel rim.
$>$ The distance between these markings and the nearest dead centres are measured (in cms)
$>$ Using the above mentioned formula, the crank angle from the TDC and BDC are found and the Valve timing diagram drawn.
> Mark the valve overlap period.

TABULAR COLUMN:

| VALVE OPENING <br> /CLOSING | DISTANCE WITH RESPECT TO THE <br> NEAREST DEAD CENTRE | CRANK ANGLE |
| :---: | :--- | :--- |
| IVO |  |  |
| IVC |  |  |
| EVO |  |  |
| EVC |  |  |

## RESULT

Thus, the valve timing diagram has been drawn with relative crank angle.
The angle of overlap =

## AIM : TO DRAW THE PORT TIMING DIAGRAM OF A TWO STROKE ENGINE.

## APPARATUS : CUT SECTION MODEL OF A 2 STROKE ENGINE

PROCEDURE
> The experiment is conducted in a reduced scaled model of a 2 -stroke engine. The aim is to represent diagrammatically the sequence of operation of the inlet port, transfer port and the Exhaust port.
$>$ The flywheel of the model is rotated till the piston reaches the BDC.
$>$ The lower most portion of the flywheel and the corresponding portion of the base is properly marked.
> The total length of the circumference of the flywheel outer rim is measured as ' C ' corresponding to 360 degrees of flywheel rotation.
$>$ The flywheel is rotated till the piston reaches TDC for compression stroke.
$>$ The following points are marked on the flywheel rim:
(i)Inlet port opening (IPO) : Fresh charge enters into the crankcase from carburetor.
(ii) Exhaust Port Closes (EPC): After the full closing of the exhaust port and transfer port, Compression of the charge begins.
(iii)Transfer port closes(TPC): When the reaches spark ignition accompanied by the explosion.

TABULAR COLUMN

| PORT OPENING/CLOSING <br> POSITION | DISTANCE MEASURED <br> ALONG THE <br> FLYWHEEL(cm) | ANGLE <br> (Degrees) |
| :--- | :---: | :---: |
| IPO before TDC |  |  |
| IPC after TDC |  |  |
| EPO before BDC |  |  |
| EPC after BDC |  |  |
| TPO before BDC |  |  |
| TPC after BDC |  |  |

RESULT : Hence the port timing diagram has been drawn .

## EXP \# 4 PERFORMANCE TEST ON SINGLE CYLINDER - 4 STROKE DIESEL ENGINE

AIM:
To conduct a performance test on the given single cylinder 4-stroke diesel engine and to draw the following characteristic curves:
a) Brake Power Vs Specific Fuel Consumption
b) Brake Power Vs Mechanical Efficiency
c) Brake Power Vs Brake Thermal Efficiency
d) Brake Power Vs Indicated Thermal Efficiency

## APPARATUS REQUIRED:

Stop watch, Tachnometer

## ENGINE DETAILS:

Type : Single cylinder four stroke Diesel engine Coupled to an eddy current dynamometer.
Power $\quad: 8 \mathrm{HP}(5.9 \mathrm{~kW})$
Speed $: 850 \mathrm{rpm}$.
Orifice diameter $: 20 \mathrm{~mm}$.
Effective radius of the brake drum : 0.2 m

## PROCEDURE:

> The cooling water lines are opened.
> The fuel in the tank and the valve (used to allow fuel from tank) position are checked.
$>$ The engine is started at no load.
$>$ The time taken for 10 cc of fuel consumption is noted by keeping the right side knob in closed position.
> Supply of fuel from the main tank is ensured after taking the above reading.
$>$ The manometric heights $\mathrm{h}_{1}$ and $\mathrm{h}_{2}$ are taken down in cm .
$>$ The engine is loaded by adding weights on the mechanical dynamometer.
$>$ All the above readings are noted down.
$>$ The experiment is repeated for different current settings.
$>$ The readings are tabulated neatly.
FORMULAE REQUIRED :
a) Brake Power (BP ) $=(V \times I) / 1000 \mathrm{~kW}$
b) Fuel consumption $(\mathrm{FC})=\left(10 \times 10^{-3} \mathrm{x}\right.$ sp.gr. $\left.\times 3600\right) / \mathrm{t} \mathrm{kg} / \mathrm{hr}$
where, $\quad \mathrm{t}$ - Time taken for 10 cc of fuel consumption sp. gr. - specific gravity of diesel $=0.86$
c) Specific Fuel Consumption (SFC) $=\mathrm{FC} / \mathrm{BP} \quad \mathrm{kg} / \mathrm{kW} \mathrm{hr}$
d) Heat Input $(\mathrm{HI})=\mathrm{FC} * \mathrm{CV} / 3600 \mathrm{~kJ} / \mathrm{s}$

Where, $\quad$ CV - Calorific Value of diesel $=43500 \mathrm{~kJ} / \mathrm{kg}$
e) Brake thermal efficiency $\left(\eta_{\mathrm{BTE}}\right)=(\mathrm{BP} / \mathrm{HI}) * 100 \quad \%$
f) Mechanical efficiency $\left(\eta_{\text {mech }}\right)=(\mathrm{BP} / \mathrm{IP}) * 100 \quad \%$

Where, BP - Brake Power, kW
IP - Indicated Power, kW $\quad=\mathrm{BP}+\mathrm{FP}$
FP - Frictional Power, kW (To be determined from "BP Vs FC" plot)
g) Indicated thermal efficiency,: $\eta_{\text {ITE }}=(\mathrm{IP} / \mathrm{HI}) \times 100$

RESULT:
A performance test was conducted on the given diesel engine and the following characteristic curves were drawn.
a) Brake Power Vs Specific Fuel Consumption
b) Brake Power Vs Mechanical Efficiency
c) Brake Power Vs Brake Thermal Efficiency
d) Brake Power Vs Indicated Thermal Efficiency
"WILLAN's Line Method" TO Find Frictional Power (Model Graph)


Frictional Power $\leq 1 / 4$ th of rated power


HEAT BALANCE TEST ON 4 STROKE DIESEL ENGINE

AIM :
To conduct heat balance test on the given diesel engine and to draw up a heat balance sheet showing the proportion of useful work and various losses.

APPARATUS REQUIRED :
Stop watch, tachometer

## ENGINE DETAILS:

Type : Single cylinder four stroke Diesel engine coupled to an eddy current dynamometer.
Power $\quad: 8 \mathrm{HP}(5.9 \mathrm{~kW})$
Speed $\quad: 850 \mathrm{rpm}$.
Orifice diameter $: 20 \mathrm{~mm}$.
Effective radius of the brake drum : 0.2 m

PROCEDURE :

- The maximum load (full load) is calculated from the engine ratings.
- The $1 / 4,1 / 2$ and $3 / 4$ of full load are estimated.
- The cooling water lines are opened.
- The fuel in the tank and the valve ( to allow fuel from the tank ) position are checked.
- The engine is started at no load condition.
- The time taken for 10 cc of fuel consumption is noted by keeping the right side knob in closed position.
- Supply of fuel from the main tank is ensured after taking the above reading.
- The following readings are also noted:
* Temperature of cooling water outlet
* Temperature of exhaust gas
* Manometer readings
- The engine is loaded ( with $1 / 4,1 / 2,3 / 4$ and full load) and all the above readings are noted down.
- The readings are tabulated neatly.


## FORMULAE REQUIRED :

a) Maximum load / Full load ( $\mathrm{W}_{\max }$ ) = BP $\mathrm{rated} \times 60000 /\left(2 \pi \mathrm{~N}_{\text {rated }} \times \mathrm{R}_{\text {eff }} \times 9.81\right)$ Where,
$\mathrm{BP}_{\text {rated }}$ - Rated Brake Power, kW
$\mathrm{N}_{\text {rated }}$ - Rated Speed, rpm
$\mathrm{R}_{\text {eff }} \quad$ - Effective Radius of the brake drum
b) Brake Power (BP) = $2 \Pi$ NT/60000 kW
where, N - speed in rpm , T - Torque in $\mathrm{N}-\mathrm{m}=\mathrm{W}_{\text {net }} \times \mathrm{R}_{\text {eff }}$
c) Heat carried away by cooling water $\left(\mathrm{Q}_{\mathrm{cw}}\right)=\mathrm{m}_{\mathrm{w}} \mathrm{c}_{\mathrm{pw}}\left(\mathrm{t}_{\mathrm{w} 2}-\mathrm{t}_{\mathrm{w} 1}\right)$
where, $m_{w}$ - mass flow rate of cooling water $=1 / \mathrm{t}_{2} \mathrm{~kg} / \mathrm{s}$
$\mathrm{t}_{2}$ - time for 10 litres of water collection
$c_{\mathrm{pw}}-$ specific heat of water $=4.187 \mathrm{~kJ} / \mathrm{kg} \mathrm{K}$
$\mathrm{t}_{\mathrm{w} 1}$ - temperature of cooling water at inlet, ${ }^{\circ} \mathrm{C}$
$\mathrm{t}_{\mathrm{w} 2}$ - temperature of cooling water at outlet, ${ }^{\circ} \mathrm{C}$
d) Heat carried away by exhaust gases ( $\mathrm{Q}_{\mathrm{eg}}$ )
$=\mathrm{m}_{\mathrm{g}} \mathrm{c}_{\mathrm{pg}}\left(\mathrm{t}_{\mathrm{go}}-\mathrm{t}_{\mathrm{gi}}\right)$
where,
$\mathrm{m}_{\mathrm{g}}$ - mass flow rate of exhaust gases $=\mathrm{m}_{\mathrm{f}}+\mathrm{m}_{\mathrm{a}}$
$\mathrm{m}_{\mathrm{f}}-$ mass flow rate of fuel $=10 \times 10^{-3} \times \mathrm{sp} . \mathrm{gr} / \mathrm{t}_{1}$
sp. gr. - specific gravity of diesel $=0.86$
$t_{1}$ - time taken for 10 cc of fuel consumption
$m_{a^{-}}$mass flow rate of air $=\rho_{a} C_{d} A_{o} \sqrt{ } 2 g h_{a}$
where, $\mathrm{C}_{\mathrm{d}}$ - Co efficient of discharge of orifice meter $=0.62$
$\mathrm{A}_{0}$ - Area of orifice meter $=\pi \mathrm{d}^{2} / 4$
d - diameter of orifice
g $-9.81 \mathrm{~m} / \mathrm{s}^{2}$
$\mathrm{h}_{\mathrm{a}}=\left(\mathrm{h}_{1}-\mathrm{h}_{2}\right)^{*} \rho_{\mathrm{w}} /\left(\rho_{\mathrm{a}} * 100\right)$
where,
$\mathrm{h}_{1} \mathrm{~h}_{2}$ - manometer readings cm
$\rho_{\mathrm{w}}$ - density of water $=1000 \mathrm{~kg} / \mathrm{m}^{3}$
$\rho_{\mathrm{a}}$ - density of air at room temperature
$=\left(\rho_{\mathrm{a}}\right.$ at STP $) * 273 /\left(273+\mathrm{t}_{\mathrm{R}}\right)$
where,
$\rho_{\mathrm{a}}$ at STP $=1.18 \mathrm{~kg} / \mathrm{m}^{3}$
$\mathrm{t}_{\mathrm{R}}$ - Room Temperature, ${ }^{\circ} \mathrm{C}$
$\mathrm{c}_{\mathrm{pg}}-$ specific heat of exhaust gases $=1.005 \mathrm{~kJ} / \mathrm{kg} \mathrm{K}$
$\mathrm{t}_{\mathrm{go}}$ - temperature of exhaust gases, ${ }^{\circ} \mathrm{C}$
$\mathrm{t}_{\mathrm{gi}}$ - temperature of atmospheric air, ${ }^{\circ} \mathrm{C}$
e) Heat Input ( HI ) $=\mathrm{m}_{\mathrm{f}} \times$ C.V.
where, $\mathrm{m}_{\mathrm{f}}$ - mass flow rate of fuel, $\mathrm{kg} / \mathrm{s}$
C.V. - Calorific Value of diesel $=40,500 \mathrm{~kJ} / \mathrm{kg}$
f) Unaccounted Heat loss $\left(\mathrm{Q}_{\mathrm{un}}\right)=\mathrm{HI}-\left(\mathrm{BP}+\mathrm{Q}_{\mathrm{cw}}+\mathrm{Q}_{\mathrm{eg}}\right) \mathrm{kJ} / \mathrm{s}$

## GRAPH:

Brake power( x axis) Vs Thermal Energy distribution. (Y axis )
Heat input (x axis ) Vs Thermal Energy Distribution (Y axis)
RESULT: A heat balance test was conducted on the given diesel engine and a heat balance sheet has been drawn up for different load conditions.


A: Radiation \&
Unaccounted loss
B: Exhaust loss

C: coolant
D: Brake Power
Brake Power, BP

## EXPERIMENT \# 6

## MORSE TEST ON A 4 CYLINDER DIESEL ENGINE COUPLED WITH HYDRAULIC DYNAMOMETER

AIM : To conduct a morse test on a multicylinder engine coupled with hydraulic dynamometer and to find the frictional power.

## SPECIFICATIONS:

1. Number of cylinders $=4$.
2. Power $=7.36 \mathrm{~kW} / 10 \mathrm{HP}$
3. $\quad$ Speed $=1500 \mathrm{rpm}$
4. Type $=4$ Cylinder 4 stroke vertical- Diesel engine
5. $\mathrm{MAKE}=$ Ambassador
6. Loading $=$ Hydraulic Dynamometer

## APPARATUS REQUIRED:

1. Dynamometer.
2. Tachometer.

## DEFINITIONS:

1. Brake Power(BP): The net power available at the output shaft of the internal combustion engine is called brake power. It is the power available after some power is utilized for overcoming friction.
2. Indicated Power(IP): The horse power developed inside the cylinder is called Indicated Power.
3. Friction Power(FP): The difference between indicated power and brake power is called Friction Power.
4. Mechanical Efficiency: The ratio of brake power to the indicated power expressed as a percentage is called Mechanical Efficiency.

CALCULATION OF THE MAXIMUM LOAD THAT CAN BE APPLIED TO THE ENGINE :
$\mathrm{BP}_{\text {rated }}=\left(\mathrm{W}_{\text {max }} \times \mathrm{N}_{\text {rated }}\right) / 2720$
$\mathrm{W}_{\text {max }}=7.36 \times 2720 / 1500=13.35 \mathrm{~kg}$
$\mathrm{Bp}_{\text {rated }}=7.36 \mathrm{~kW}$
$\mathrm{N}_{\text {rated }}=1500 \mathrm{rpm}$
$\mathrm{W}_{\text {max }}$ is in kg

## FORMULAE:

1. Brake Power $\left(\mathrm{BP}_{\mathrm{T}}\right)=(\mathrm{W} \times \mathrm{N}) / 2720 \mathrm{~kW}$
where, $\mathrm{W}=$ Load applied in kg ,
$\mathrm{N}=$ Speed in rpm,
2720 = Dynamometer constant.
2. Total Indicated Power $\left(\mathrm{IP}_{\mathrm{T}}\right)=\left[\mathrm{IP}_{1}+\mathrm{IP}_{2}+\mathrm{IP}_{3}+\mathrm{IP}_{4}\right] \mathrm{kW}$ where, 1 - I cylinder is cut off.

2 - II cylinder is cut off.
3 - III cylinder is cut off.
4 - IV cylinder is cut off.
3. If , BP : Brake Power developed when all the cylinders are working, then $\mathrm{IP}_{1}=\mathrm{BP}-\mathrm{BP}_{1}$. $\mathrm{IP}_{2}=\mathrm{BP}-\mathrm{BP}_{2}$.
$\mathrm{IP}_{3}=\mathrm{BP}-\mathrm{BP}_{3}$.
$\mathrm{IP}_{4}=\mathrm{BP}-\mathrm{BP}_{4}$.
4. $\mathrm{BP}_{1}=\mathrm{W}_{1} \times \mathrm{N} / 2720$
$\mathrm{BP}_{1}$ : Brake power of the engine when the first cylinder is cut off .
$\mathrm{BP}_{2}$ : Brake power of the engine when the second cylinder is cut off.
$\mathrm{BP}_{3}$ : Brake power of the engine when the third cylinder is cut off.
$\mathrm{BP}_{4}$ : Brake power of the engine when the fourth cylinder is cut off.
$\mathrm{W}_{1}$ :Load on the dynamometer when the first cylinder if cut off and the speed is maintained to be N rpm.
5. Mechanical Efficiency $=\left(\mathrm{BP}_{\mathrm{T}} / \mathrm{IP}_{\mathrm{T}}\right) * 100 \%$.

## PROCEDURE:

1. The engine is first started by checking the fuel, lubricating oil, cooling water etc., with all 4 cylinders working.
2. The speed of the engine is adjusted to a particular value say 1500 rpm and the BP of the engine at that speed is calculated.
3. Now cylinder 1 is cut-off and the speed is adjusted to 1500 rpm byn decreasing the load and the BP is found out.
4. The difference of BP of all the 4 cylinders and that of the cut off cylinder gives the IP of the cylinder of the cut-off cylinder.
5. In this way each cylinder is cut-off and the BP is found out.
6. From the value of BP the IP of all the cylinders is found out.
7. The sum of the indicated powers of all the cylinders gives the indicated power of the entire engine.
8. The Mechanical Efficiency is then calculated at that particular speed.

| CONDITIONS | LOAD | BRAKE POWER <br> B.P=[(W x N $) / 2720] \mathrm{kW}$ | INDICATED <br> POWER | MECHANICAL <br> EFFICIENCY |
| :--- | :---: | :---: | :---: | :---: |
|  | W <br> kg | kW | Watts | $\%$ |
| SPEED , N, <br> (rpm) | USUALLY, 1200 RPM, 1300 RPM, OR 1500 RPM |  |  |  |
| All cylinders <br> working |  |  |  |  |
| First cylinder is <br> cut off |  |  |  |  |
| Second cylinder <br> is cut off |  |  |  |  |
| Third cylinder is <br> cut off |  |  |  |  |
| Fourth cylinder <br> is cut off |  |  |  |  |

RESULT:
Thus the Morse Test on the 4-stroke, 4-cylinder diesel engine is carried out and the mechanical efficiency is determined.

## PRECAUTION

1.Lubricating oil level should be checked.
2.Cooling water should be supplied to engine and dynamometer.
3. Water is to be supplied to dynamometer and water pressure is to be $0.5 \mathrm{~kg} / \mathrm{cm}^{2}$.

GRAPH :
SPEED
(i) vs Brake Power (BP)
(ii)vs Indicated power (IP)
(iii)Frictional power (FP)
(iv)Mechanical Efficiency ( $\eta_{\text {mech }}$ )


## EXPERIMENT \# 7: RETARDATION TEST TO FIND THE FRICTIONAL POWER OF A 4-STROKE DIESEL ENGINE

AIM:
To conduct the retardation test on the given single cylinder 4-stroke diesel engine and to draw the graph between the Drop in speed and the time taken.

## APPARATUS REQUIRED:

Stop watch, Tachometer

## ENGINE DETAILS:

Type : Single cylinder four stroke Slow speed Diesel engine, Loaded with a brake drum dynamometer.
Power : $8 \mathrm{HP}(5.9 \mathrm{~kW})$
Speed : 850 rpm

## PROCEDURE:

> The cooling water lines are opened.
$>$ The fuel in the tank and the valve (used to allow fuel from tank) position are checked.
> The engine is started at no load and allowed to run at the rated speed for about 5 minutes..
$>$ After the engine has attained the rated speed, the fuel supply is cut off. The time taken for the speed to decrease through 100 rpm from the rated speed is noted.
$>$ As soon as the reading is taken, the fuel is again allowed to flow and the maximum speed is reached to achieve
> Time for 200 , 300 and 400 rpm drop of speed from the maximum speed is taken.
$>$ The engine is now loaded to $50 \%$ of the full load and is allowed to attain the rated speed.
$>$ After attaining the rated speed, the same procedure is followed and the time for the various speed reductions is noted.
> The readings are tabulated neatly.
Calculation of the Maximum load, Wmax
$>$ Brake Power (BP)
$=\left(2 \pi \mathrm{~N}_{\mathrm{rated}} \mathrm{T}\right) /(60 \times 1000) \mathrm{kW}$
> Brake Torque ( T $=($ BP x 60000$) /\left(2 \pi \mathrm{~N}_{\text {rated }}\right) \quad \mathrm{Nm}$.

$$
=(5.9 \times 60 \times 1000) /(2 \pi \times 850)
$$

$$
\mathrm{T}=66.32 \mathrm{Nm}
$$

$$
\begin{aligned}
& \mathrm{T}=\mathrm{W}_{\max } \times \mathrm{R}_{\text {eff }} \\
& \begin{aligned}
\mathrm{W}_{\max } & =\mathrm{T} / \mathrm{R}_{\text {eff }} \\
& =66.32 / 0.2=331.58 \mathrm{~N}=331.58 / \mathrm{g}=331.58 / 9.81=33.8 \mathrm{~kg}
\end{aligned}
\end{aligned}
$$

Hence the maximum load that can be applied to the engine is 33.8 kg .
$\mathrm{R}_{\text {eff }}=$ Effective radius of the brake drum $=0.2 \mathrm{~m}$

FORMULAE USED FOR CALCULATIONS :
$>$ Brake Power (BP) $\quad=(2 \pi$ NT $) /(60 \times 1000) \mathrm{kW}$
N : Speed values of the engine from the tabular column.
T : Brake torque $=\mathrm{W}_{\text {net }} \times \mathrm{R}_{\text {eff }}$.
$\mathrm{W}_{\text {net }}=\mathrm{W}_{1}+\mathrm{w}-\mathrm{W}_{2}$.
$\mathrm{W}_{1}$ :Load added on the weight hanger.
w : Weight of the hanger. $=2 \mathrm{~kg}$
$\mathrm{W}_{2}$ :Spring balance reading, kg
$>$ Torque at $1 / 2 \operatorname{load}\left(\mathrm{~T}_{1 / 2}\right)=\mathrm{T} / 2 \quad \mathrm{Nm}=66.32 / 2=33.16 \mathrm{Nm}$
$>$ Frictional que $\left(\mathrm{T}_{\mathrm{f}}\right) \quad=\left(\mathrm{T}_{1 / 2} * \mathrm{t}_{1}\right) /\left(\mathrm{t}_{2}-\mathrm{t}_{1}\right)$
Where $t_{i}$ is the time for a drop of 100 rpm at no load $\mathrm{t}_{2}$ is the time taken for a drop of 100 rpm at $50 \%$ load.

Frictional Power $(\mathrm{FP})=2 \Pi \mathrm{NT}_{\mathrm{f}} / 60000$.
TABULAR COLUMN :

| S. No. | Speed Variation <br> "rpm" | Drop in speed <br> N <br> "rpm" | Time for the <br> speed drop at no <br> load <br> $\mathrm{t}_{1}, \mathrm{sec}$ | Time for the <br> speed drop at <br> $50 \%$ load <br> $\mathrm{t}_{2}$, sec |
| :--- | :--- | :--- | :--- | :--- |
| 1 | $715-615$ |  |  |  |
| 2 | $715-515$ |  |  |  |
| 3 | $715-415$ |  |  |  |
| 4 | $715-315$ |  |  |  |

MODEL GRAPH :


RESULT:
Thus the retardation test on the single cylinder 4- Stroke diesel engine was conducted and the frictional power was found to be $\qquad$

```
EXPERIMENT # 8
    DETERMINATION OF VISCOSITY USING REDWOOD VISCOMETER
```

AIM
To determine the viscosity of the given oil at various temperatures and study the variation of viscosity with temperature.

## APPARATUS REQUIRED

Thermometer and stop watch

## DESCRIPTION

- The given apparatus consists of two metal cups with holes at the bottom.
- The oil in the cup receives heat from the surrounding water which in turn is heated by a coil.
- Provisions are made to measure the temperatures of water and oil.
- To ensure uniform heating an electric stirring arrangement is provided.


## PROCEDURE

- The oil cup is cleaned and the oil is poured.
- Sufficient water is poured in the space surrounding the cup.
- The time taken for collection of 60 ml of oil is noted at room temp.
- The heater and stirrer are switched on.
- The time taken for collection of 60 ml of oil is noted at different temperatures from $40^{\circ} \mathrm{C}$ to $60^{\circ} \mathrm{C}$ with an interval of $10^{\circ} \mathrm{C}$.
- The values of kinematic and absolute(dynamic) viscosities are calculated using Redwood seconds.

FORMULAE REQUIRED

1. Kinematic Viscositym, $v=[\mathrm{At}-\mathrm{B} / \mathrm{t}]$ stokes where,
$\mathrm{A}=0.0026 \quad \& \quad \mathrm{~B}=1.88$
2. Density $\quad \rho=\rho_{\mathrm{r}}-0.000657\left(\mathrm{~T}-\mathrm{T}_{\mathrm{r}}\right) \quad \mathrm{kg} / \mathrm{lt}$
where, $\quad \rho_{\mathrm{r}}$. density at room temperature $=0.84 \mathrm{~kg} / \mathrm{lt}$
$T$ - temperature at which viscosity is determined $T_{r}$-room temperature
3. Absolute viscosity or dynamic viscosity $\mu=v^{*} \rho \quad$ poise
note:
1 stoke $=1 \mathrm{~cm}^{2} / \mathrm{s} ; 10$ poise $=1 \mathrm{~N} \mathrm{~s} / \mathrm{m}^{2}$

TABULAR COLUMN

| Sl.No. | temp. <br> of <br> water | temp. of <br> oil | time for 60ml <br> of oil <br> collection | density | kinematic <br> viscosity | absolute or <br> dynamic <br> viscosity |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
|  | ${ }^{\circ} \mathrm{C}$ | ${ }^{\circ} \mathrm{C}$ | sec | $\mathrm{kg} / \mathrm{lt}$ | stokes or $\mathrm{cm}^{2}$ <br> $/ \mathrm{s}$ | Poise |
|  |  |  |  |  |  |  |

RESULT
The viscosity of the given oil was found out at various temperatures and following graphs were plotted.
a) temperature Vs density
b) temperature Vs kinematic viscosity
c) temperature Vs absolute viscosity
d) Temp Vs Redwood Seconds

## EXPERIMENT \# 8

## DETERMINATION OF FLASH \& FIRE POINT OF LIQUID FUEL

AIM:

To determine the flash and fire points of the given oil using Pensky-Mortein's apparatus.

## APPARATUS REQUIRED:

Thermometer

Flash and Fire points:
Flash point is the temperature to which the oil must be heated to give off sufficient vapour to form an inflammable mixture with air.

Fire point is the lowest temperature at which the production of combustible gas from the oil is enough to maintain a steady flame after ignition.

## DESCRIPTION:

- The apparatus consists of a brass cup with a filling mark inside.
- The cup is surrounded by electric heating elements.
- The brass cup is closed with a cover where provisions are made to insert thermometer, to introduce test flame and to connect the stirrer.


## PROCEDURE:

- The oil cup is cleaned and dried.
- The oil is poured in the cup up to the filling mark.
- The flash and fire points are noted in the open conditions.
- The cup is covered with the given lid.
- The heater is connected to the mains and the rate of heating is adjusted.
- As the oil temperature increases, the test flame is applied at an interval of $20^{\circ} \mathrm{C}$ to find the flash and fire points.

TABULAR COLUMN:

| sample oil | condition | Sl.No. | Flash point | Fire point |
| :--- | :--- | :--- | :--- | :--- |
|  | open cup | 1 |  |  |
|  |  | 2 |  |  |
|  | closed cup | 1 |  |  |
|  |  | 2 |  |  |

## RESULT:

a) The flash point of the oil

1 ) in open condition is $\qquad$ ${ }^{\circ} \mathrm{C}$
2) in closed condition is $\qquad$ ${ }^{\circ} \mathrm{C}$
b) The fire point of the oil

1) in open condition is $\qquad$ ${ }^{\circ} \mathrm{C}$
2) in closed condition is $\qquad$ ${ }^{\circ} \mathrm{C}$

## STEAM LABORATORY

## EXPERIMENT \# 1 <br> STUDY OF STEAM BOILER AND TURBINES

AIM : To study and understand the principle of steam boilers and turbines.

## THEORY:

DEFINITION:
Boiler, also called steam generator is the engineering device which generates steam at constant pressure .It is a closed vessel, generally made of steel in which vaporization of water takes place.

Heat required for vaporization may be provided by the combustion of fuel in furnace ,electricity ,nuclear reactor, hot exhaust gases, solar radiations. etc.

TYPES OF BOILERS:
Boilers are of many types. Depending upon their features they can be classified as given under:
(a)Based upon the orientation /axis of the shell.
(i)Vertical boiler has a vertical shell..
(ii)Horizontal boiler has its shell horizontal.
(iii)Inclined boiler has its shell inclined .
(b)Based upon utility of boiler.
(i)Stationary boiler
(ii)Portable boiler
(c)Based on the type of firing employed.
(i)externally fired boiler; e.g : Lancahsire boiler, Locomotive boiler
(ii)Internally fired boiler ., e.g .Cochran boiler, Babcock and Wilcox boiler
(d)Based on the tube content.
(i)Fire tube boiler e.g: Cornish, Cochran, Lancashire, Locomotive boiler.

Hot gases flow through the tubes and water surrounds them.
(ii)Water tube boiler

Water flows through the tubes and hot gases surrounds them.
e.g: Babcock and Wilcox boiler, Stirling boiler, La- Mont boiler, Benson boiler .
(e)Based on the type of fuel used.
(i)solid fuel boiler :e.g : coal fired boilers.
(ii)Liquid Fuel boiler, e.g: Oil fired boilers.
(iii)Gas fired boiler : e.g: Natural gas fired boilers.
(f)Based on circulation

According to the flow of the water/steam is caused by the density difference which is due to the temperature variation.
(i)Natural circulation

Here, the circulation of water/steam is caused by the density difference which is due to the steam temperature.
(ii)Forced Circulation boilers.

Here, the circulation of water/steam is caused by a pump, or fan.
One Water tube boiler is discussed below.

## BABCOCK AND WILCOX BOILER :



SV: Safety valve
MH :Man hole
PG :Pressure Gauge
BP :Baffle Plate
WLI: Water Level Indicator
APP :Anti priming pipe
SH:Super heater
H:Header
D: Door
BDV: Blow down valve
WT: Water Tubes
FV: Feed Valve
MC: Mud collector
FD :Fire Door
G:Grate

## Definition

Babcock and Wilcox boiler is an example of water tube boiler which is horizontal straight tube boiler and may be designed for stationary or marine purposes.
The particulars relating to this boiler are

| Diameter of the drum | 1.22 to 1.83 m |
| :--- | :--- |
| Length | 6.096 to 9.144 m |
| Size of the water tubes | 7.62 to 10.16 cm |
| Size of the super heater tubes | 3.84 to 5.71 cm |
| Working pressure | $40 \mathrm{bar}(\mathrm{max})$ |
| Steaming capacity | $40000 \mathrm{~kg} / \mathrm{hr}(\mathrm{max})$ |
| Efficiency | 60 to $80 \%$ |

The Babcock and Wilcox Boiler consist of a drum connected to a series of front end and rear end header by short riser tubes. To these headers are connected a series of inclined water tubes of solid drawn mild steel.

The angle of inclination of the water tubes to the horizontal is about $15^{\circ}$ or more. A hand hole is provided in the header in front of each tube fro cleaning and inspection of tubes. A feed valve is provided to fill the drum and inclined tubes with water the level of which is indicated by the water level indicator. Through the fire door the fuel is supplied to grate where it is burnt. The water form the drum flows through the inclined tubes via down take header and goes back into the shell in the form of water and steam via uptake header. The steam gets collected in the steam space of the drum. The steam then enters through the anti-priming pipe and flows in the super-heater tubes where it is further heated and is finally taken out through stop valve and supplied to the engine when needed. At the lowest point of the boiler is provided the mud collector to remove the mud particles through a blow down clock.

## BOILER MOUNTINGS

## 1.WATER LEVEL INDICATOR

It is used for knowing the level of water in the boiler.

## 2.SAFETY VALVE

Its function is to prevent the steam pressure from exceeding a limiting maximum pressure value. Safety valve should operate automatically by releasing excess steam and bring pressure down within safe limits.

## 3.FUSIBLE PLUG

It is a safety device used for preventing the level of water from going down below a critical point and thus avoid overheating. It is mounted at crown plate of combustion chamber.

## 4.PRESSURE GAUGE

It is mounted at front top. Pressure is continuously monitored so as to avoid occurrence of over shooting of boiler pressure.

## 5.STOP VALVE

It regulates the flow of steam from the boiler .This is generally mounted on highest part of boiler shell and performs function of regulating the flow of steam from boiler.
6.FEED CHECK VALVE

It is a non return valve at the end of delivery pipe from feed water pump and is placed on boiler shell slightly below normal water level. It is used to fill the boiler with water.

## 7.BLOW OFF COCK

It is used for cleaning by discharging the water and sediments from bottom of boiler. Blow off cock also helps in regulating the salt concentration as frequent draining helps in throwing out salt deposited over period of time. Opening blow off cock removes deposited sediments in boiler.

## 8.MANHOLE AND MUD BOX

Manhole provides the opening for cleaning ,inspection and maintenance purpose.
Mud box is a collection chamber for collecting the mud.

## ACCESSORIES

## 1.SUPER HEATER

Its purpose is to super heat steam and is a type of heat exchanger in which steam flows inside tubes and hot gases surround it.

## 2.ECONOMISER

An economizer is a device used to heat feed water by utilizing the heat in the exhaust flue gases before leaving through the chimney.

## 3.AIR PREHEATER

An air preheater is used to recover the heat from the exhaust flue gases. It is installed between the economizer an the chimney. The air required for the purpose of combustion is drawn through the air preheater where its temperature is raised. It is then passed through the ducts to the furnace. The air is passed through the tubes of the heater internally while hot flue gases are passed over the outside of the tubes.

## STEAM TURBINES

A steam turbine is a prime mover in which rotary motion is obtained by the gradual change of momentum of the steam.
In general a steam turbine consists of the following
1.The nozzle in which the heat energy of high pressure steam is converted to kinetic energy, so that the steam issues from the nozzle with a very high velocity.
2.The blades change the direction of steam issuing form the nozzle, so that a force acts on the blades due to the change of momentum and propel them.

## PRINCIPLE

The principle of steam turbine is the generation of high velocity steam jet by the expansion of high pressure steam and then conversion of kinetic energy, so obtained into mechanical work on rotor blades.

## CLASSIFICATION OF STEAM TURBINE

1.According to the mode of steam action
(i)Impulse turbine (ii)Reaction turbine
2.According to the direction of steam flow
(i)Axial flow turbine (ii)Radial flow turbine
3.According to the exhaust condition of steam
(i)Condensing turbine (ii)Non condensing turbine
4.According to the pressure of steam
(i)High pressure turbine (ii)Medium Pressure turbine (iii)high pressure turbine
5.According to the number of stages
(i)Single stage turbine (ii)Multi stage turbine

## IMPULSE TURBINE

An impulse turbine is a turbine which runs by the impulse of steam jet of this turbine, the steam is first made to flow through the nozzle. Then the steam jet impinges on the turbine blades (which are curves like blades) and are mounted on the circumference of the wheel. The steam jet after impinging glides over the concave surface of the blades and finally leave the turbine.

## REACTION TURBINE

In a reaction turbine, the steam enters the wheel under pressure and flows over the blades.The steam, while gliding ,propels the blades and make them to move.As a matter of fact, the turbine runner is rotated by the reactive forces of steam jets.The backward motion of the blades is similar to the recoil of a gun.

DIFFERENCE BETWEEN THE STEAM AND TURBINE AND A STEAM ENGINE

| Sr <br> No | Criteria | Steam Turbine | Steam Engine |
| :--- | :--- | :--- | :--- |
| 1 | Thermal Efficiency | High | Less |
| 2 | Balancing | As there is no reciprocating <br> parts in steam turbine, perfect <br> balancing is possible. | Hard to achieve perfect balancing <br> and needs heavy foundation. |
| 3 | Speed | Higher and greater range of <br> speed is possible | Lesser range of speed. |
| 4 | Lubrication | simple | Complicated |
| 5 | Need of Flywheel | As the power generation here <br> is at a uniform rate, a <br> flywheel is not required. | A flywheel is needed. |
| 6 | Steam consumption | Less | More |
| 7 | Compactness | More compact | Bulky |
| 8 | Construction | Lighter construction and <br> easier maintenance | Complex construction |
| 9 | Cost | Affordable cost than the <br> steam engine | Costly than the steam turbine. |

RESULT : Hence, the steam generator and the steam turbine are studied.

AIM : To conduct a test on an oil fired steam generator and
(i) To Calculate the Boiler Thermal Efficiency
(ii) To Draw the heat balance test.

## SPECIFICATIONS :

TYPE : Non IBR - Automatic, three pass, Oil fired Boiler with Economiser.
CAPACITY : $200 \mathrm{Kg} / \mathrm{hr}$
OPERATING PRESSURE : $12 \mathrm{~kg} / \mathrm{cm}^{2}$.
FUEL : High Speed Diesel
CALORIFIC VALUE : $44500 \mathrm{~kJ} / \mathrm{kg}$
APPARATUS :
Steam generator Test Rig, and accessories.

## PRECAUTIONS

1.Check the water level in the feed water tank.
2.Check the fuel level in the diesel tank.
3.Switch on the mains.
4.Switch on the side switch , and check the panel./
5.Start the water pump after opening the blow down valve.
6.Wait for two minutes.
7.Ensure that water flow is there and close the blow down valve.
8.Start the burner.

9 .Allow the pressure to rise to $10 \mathrm{~kg} / \mathrm{cm}^{2}$ in the boiler.
10.Close the steam valve to build the pressure.
11.The steam bypass valve is opened to enable the steam to be exhausted to the turbine, maintaining the pressure.

The following observations are recorded in the observation table.(TABLE \#1)
OBSERVATIONS :

1. Clock time ( in minutes)
2.Steam pressure, p
3.Dryness fraction , $\mathrm{x}=0.9$ (assumed)
2. Initial and final level of water in the feed water tank, $\mathrm{W}_{\mathrm{h} 1}$ and $\mathrm{W}_{\mathrm{h} 2}$.
5.Temperature of feed water, $\mathrm{T}_{\mathrm{w}}$.
6.Initial and final level of fuel in the Diesel feed tank, $D_{h 1}$ an $D_{h 2}$.
7.Flue gas temperature, $\mathrm{T}_{\mathrm{g}}$.
8.Room temperature, $\mathrm{T}_{\mathrm{R}}$.
9.Air flow meter manometer differential reading $\Delta \mathrm{A}_{\mathrm{h}}$, in mm.
10.Repeat the above measurements three times every ten minutes.

## TABULAR COLUMN

OBSERVATION TABLE :

| TIME | Steam <br> Pr | Feed Water <br> Tank Level |  | Feed <br> Water <br> Temp. | Oil level in Diesel <br> Tank |  | Flue <br> Gas <br> Temp | Room <br> Temp |  | Manometer reading |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| t | P | Initial <br> Wh1 | Final <br> Wh2 | Tw | Initial <br> Dh1 | Final <br> Dh2 | Tg | $\mathrm{T}_{\mathrm{R}}$ | Initial <br> Ah1 | Final <br> Ah2 |  |
| MIN | $\mathrm{Kg} / \mathrm{cm}^{2}$ | cm | cm | ${ }^{\circ} \mathrm{C}$ | cm | cm | ${ }^{\circ} \mathrm{C}$ | ${ }^{\circ} \mathrm{C}$ | cm | cm |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |

FORMULAE AND CALCULATIONS: (for performance test )
1.TOTAL DURATION OF TEST , $\mathrm{t}_{\mathrm{d}}$.
$\mathrm{t}_{\mathrm{d}}=\left(\mathrm{t}_{4}-\mathrm{t}_{1}\right)$ minutes.
2.AVERAGE STEAM PRESSURE, pavg.
$\mathrm{p}_{\text {avg }}=\left[\frac{p_{1}+p_{2}+p_{3}+p_{4}}{4}\right] \mathrm{kg} / \mathrm{cm}^{2}$.
3.ABSOLUTE STEAM PRESSURE, P.
$\mathrm{P}=0.9807 \mathrm{X}(\mathrm{pavg}+1.032) \mathrm{kg} / \mathrm{cm}^{2}$.
4.AVERAGE FEED WATER TEMPERATURE, Tw.
$\mathrm{Tw}=\left[\frac{T_{W 1}+T_{W 2}+T_{W 3}+T_{W 4}}{4}\right]{ }^{\circ} \mathrm{C}$
5.AVERAGE ROOM TEMPERATURE, $\mathrm{T}_{\mathrm{R} . \mathrm{R}}=\left[\frac{T_{R 1}+T_{R 2}+T_{R 3}+T_{R 4}}{4}\right]{ }^{\circ} \mathrm{C}$
6.AVERAGE FLUE GAS TEMPERATURE , Tg
$\operatorname{Tg}=\left[\frac{T_{g 1}+T_{g_{2}}+T_{g 3}+T_{g 4}}{4}\right]{ }^{\circ} \mathrm{C}$
7.MASS OF STEAM GENERATED BY THE BOILER IN ONE HOUR , $\mathrm{m}_{\mathrm{s}}$.
$\mathrm{m}_{\mathrm{s}}=$ Fall in water level x Area of feed $\times 1000 \times \underline{60}$ in td minutes water tank $\left(\mathrm{m}^{2}\right) \quad \mathrm{t}_{\mathrm{d}}$

$$
=\frac{\Delta W h}{1000} \times(1.252 \times 0.602) \times 1000 \times \frac{60}{t_{d}} \mathrm{~kg} / \mathrm{hr}
$$

Here,
$\Delta \mathrm{W}_{\mathrm{h}}=($ Final level - Initial level $)$ in water tank in mm.
Area of feed water tank $=(1.252 \times 0.602) \mathrm{m}^{2}$.
8.Mass of fuel oil supplied in one hour, $\mathrm{m}_{\mathrm{f}}$.
$m_{f}=\begin{gathered}\text { Fall in fuel level } \\ \text { in td minutes }\end{gathered}$
$x$ $\begin{gathered}\text { Area of fuel } \\ \operatorname{tank}\left(\mathrm{m}^{2}\right)\end{gathered} \quad \mathrm{x} 950 \quad \mathrm{x} \underline{\mathrm{t}_{\mathrm{d}}}$
where,
$\Delta \mathrm{D}_{\mathrm{h}}=($ Final level - Initial level $)$ in fuel tank in mm.
Area of fuel oil tank $=(0.4 \times 0.4) \mathrm{mm}^{2}$.
(a) Enthalpy, ,h of steam generated at pressure , p
$\mathrm{h}=\mathrm{h}_{\mathrm{f}}+\mathrm{h}_{\mathrm{fg}}$, for dry saturated steam , $\mathrm{kJ} / \mathrm{kg}$
$h=h_{f}+\mathrm{x}_{\mathrm{hg}}$, for wet steam
where $\mathrm{x}=0.9$ (assumed)
$\mathrm{h}_{\mathrm{f}}$ and $\mathrm{h}_{\mathrm{fg}}$, are read from steam tables for absolute steam pressure, P .
(b) Enthalpy, $\mathrm{h}_{\mathrm{w}}$ of water at feed water temperature, $\mathrm{T}_{\mathrm{w}}$.
$h_{w}$ is read from steam tables for $T_{w}$ in $\mathrm{kJ} / \mathrm{kg}$.
10.BOILER THERMAL EFFICIENCY , $\eta_{\text {THERM }}$
$\eta_{\text {THERM }}=\frac{m s(h-h w)}{m f \times 44500} \times 100 \quad \%$

RESULT :
Hence the Boiler performance test has been conducted and the efficiency is found out to be $\qquad$ $\%$.

## CALCULATIONS FOR ENERGY BALANCE TEST OF STEAM GENERATOR

1.MASS OF AIR SUPPLIED IN ONE HOUR, $\mathrm{m}_{\mathrm{a}}$.

$$
\begin{aligned}
& \mathrm{m}_{\mathrm{a}}=1.23 \times 0.62 \times 0.00785 \times\left(2 \times g \times 814 \times \frac{\Sigma \Delta A_{h}}{4}\right)^{0.5} \times 3600 \\
& m_{a}=1362.5 \times\left(\Sigma \Delta A_{h}\right)^{0.5} \mathrm{~kg} / \mathrm{hr} \\
& \Sigma \Delta A_{h}=\Delta A_{h 1}+\Delta A_{h 2}+\Delta A_{h 3}+\Delta A_{h 4}
\end{aligned}
$$

$\Delta A_{h}$ is the water level difference in manometer.
Density of Air $=1.23 \mathrm{~kg} / \mathrm{m} 3$.
Coefficient of orifice $=0.62$
Diameter of orifice $=0.1 \mathrm{~m}$
Area of orifice $=0.00785 \mathrm{~m}^{3}$.
Water to air density ratio $=814$
2.HEAT GAINED BY THE STEAM FROM THE BOILER PER HOUR,
( Qs).
$\mathrm{Q}_{\mathrm{s}}=\mathrm{m}_{\mathrm{s}} \mathrm{X}\left(\mathrm{h}-\mathrm{h}_{\mathrm{w}}\right) \mathrm{Kj} / \mathrm{hr}$
3.EVAPORATION RATE , m

$$
m=\frac{m_{s}}{m_{f}} \mathrm{~kg} \text { steam } / \mathrm{kg} \text { fuel }
$$

4.EQUIVALENT EVAPORATION, $\mathrm{m}_{\mathrm{e}}$ :

$$
m_{e}=m_{s} \times \frac{h-h_{w}}{2260} \mathrm{~kg} / \mathrm{hr}
$$

2260 : Latent heat of steam at 1 bar
5.HEAT GENERATED BY BURNT FUEL OIL, $\mathrm{Q}_{\mathrm{f}}$ :
$\mathrm{Q}_{\mathrm{f}}=\mathrm{m}_{\mathrm{f}} \times \mathrm{CV} \mathrm{kJ} / \mathrm{hr}$
CV : Calorific Value of fuel oil $=44500 \mathrm{~kJ} / \mathrm{kg}$
6.HEAT LOST TO FLUE GAS , Qg:
$\mathrm{Q}_{\mathrm{g}}=\mathrm{mg} \times \mathrm{C}_{\mathrm{pg}} \mathrm{x}\left(\mathrm{T}_{\mathrm{g}}-\mathrm{T}_{\mathrm{R}}\right)$
$m g=\left(m_{a}+m_{f}\right) \mathrm{kg}$
8.HEAT BALANCE SHEET
a) HEAT INPUT, Qf
b) useful heat, Qs =
c)Heat in exhaust gas , $\mathrm{Qg}=$
d) Radiation and other unaccounted losses, Qun =

HEAT BALANCE SHEET ON HOUR BASIS :

| HEAT SUPPLIED | kJ | Heat utilization , kJ | \% |
| :---: | :---: | :---: | :---: |
|  | $\mathrm{Qf}=\mathrm{mf} \times \mathrm{CV}$ | Heat used to generate steam $\mathrm{Qs}=$ | (Qs/Qf) x 100 |
| HEAT SUPPLIED BY FUEL |  | Heat carried by dry flue gases, $\mathrm{Qg}=$ | (Qg/Qf) x 100 |
|  |  | Heat unaccounted for , Qun= | ( Qun x Qf) x 100 |

## GRAPH :

A Graph is drawn between heat balance in \% of heat input and \% of heat utilized.
\% QunEXPERIMENT \# 1
STUDY OF STEAM BOILER AND TURBINESEXPERIMENT \#2
PERFORMANCE AND ENERGY BALANCE TEST ON A STEAM GENERATOR


RESULT: Hence A test is conducted on the boiler and the equivalent evaporation capacity and its thermal efficiency are determined.Also a heat balance sheet is drawn.

## EXPERIMENT \#3 <br> PERFORMANCE AND ENERGY BALANCE TEST ON A STEAM TURBINE

AIM : To conduct a test on the impulse steam turbine and to draw the performance curves and draw a heat balance sheet.

SPECIFICATION :
TURBINE: IMPULSE
NO. OF NOZZLES : 3
INLET PRESSURE : $10 \mathrm{~kg} / \mathrm{cm}^{2}$.
STEAM FLOW RATE : $0.01 \mathrm{~kg} / \mathrm{s} /$ Nozzle
SPEED OF TURBINE : 3000 RPM

## STARTING AND RUNNING PROCEDURE :

1.Start the boiler and maintain a pressure of $10 \mathrm{~kg} / \mathrm{cm}^{2}$.
2.Switch on the steam turbine panel which is connected to a 3 phase 440 V power source with neutral connection.
3.Swtich on the vacuum reciprocating pump.
4.Mainitch on the turbine panel board is switched on.When sufficient pressure ( 10 kg ) $\mathrm{cm}^{2}$ to $12 \mathrm{~kg} / \mathrm{cm}^{2}$ ) is built in the boiler, allow cooling water to circulate through the condenser by switching on the condenser water supply pump. Slowly open the boiler main valve a little and simultaneously open the main line valve's bypass valve in the steam line and the ball valve provided under the turbine.
5.When all the condensed water in the steam line and the turbine is driven out close the main line bypass valve and the valve under the turbine.
6.Adjust the handle of dynamometer spring balance screw so that the balance is just loaded to 0 kg .
7.Open the steam valve and adjust so that a pressure of $10 \mathrm{~kg} / \mathrm{cm}^{2}$, is maintained on the turbine pressure gauge.
8. When the turbine speed reaches about 2000rpm, load the turbine by switching on the individual switches to the bulbs.
9.Enter the observations.
10.Switch on another bulb on the loading device and after few minutes note down all the observations.
11.While doing the experiment control the pressure so that the turbine inlet pressure remains at $10 \mathrm{~kg} / \mathrm{cm}^{2}$ and the sufficient water is in the boiler.

| Turbine Speed | $\begin{aligned} & \text { Steam } \\ & \text { Pr } \end{aligned}$ |  | Steam Temp |  | Time <br> Taken <br> For <br> 4 cm <br> rise <br> in <br> conden <br> sate <br> tank | Spring balance reading | Cooling Water |  | TSC | Turbine Output Power | Cooling <br> Water <br> Flow meter <br> Mano <br> Meter <br> Reading |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\begin{aligned} & \text { In } \\ & \text { let } \end{aligned}$ | Out let | $\begin{aligned} & \hline \text { In } \\ & \text { let } \end{aligned}$ | $\begin{aligned} & \hline \mathrm{Ou} \\ & \text { tlet } \end{aligned}$ |  |  | $\begin{aligned} & \text { In } \\ & \text { let } \end{aligned}$ | Out let |  |  | Hm1 | Hm2 | $\Delta \mathrm{Hm}$ |
| n | P1 | P2 | T1 | T2 | tc | $\mathrm{W}_{\mathrm{D}}$ |  |  |  |  |  |  |  |
| rpm |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  |

CALCULATIONS :
1.Converstion of $\mathrm{kg} / \mathrm{cm}^{2}$ to bar.
$=\left[0.9807 \times\left\{\mathrm{p}_{1}\left(\mathrm{~kg} / \mathrm{cm}^{2}\right)+1.032\right\}\right]$ bar
p 1 : Steam pressure at turbine inlet.
2.Total steam consumption , TSC.

TSC + ( $1271 / \mathrm{tc}) \mathrm{kg} / \mathrm{hr}$
tc : Time taken for 4 cm rise in water level of condensate tank in sec.
3.Specific Steam consumtion, SSC
$\mathrm{SSC}=\mathrm{TSC} / \mathrm{Pd} \mathrm{kg} / \mathrm{kW}$
Pd : Turbine Output power

$$
P_{d}=\frac{1.294 \times n \times W_{D}}{10000} k W
$$

n : Speed of the turbine in rpm.
$\mathrm{W}_{\mathrm{D}}$ : Dynamometer spring balance reading, kg .
4.Rankine Efficiency , $\eta$ RE:

$$
\eta_{\mathrm{RE}}=\frac{\left(\mathrm{T}_{1}-\mathrm{T}_{2}\right)\left(1+\frac{\mathrm{h}_{\mathrm{fg} 1}}{\mathrm{~T}_{2}}\right)-T_{2} \ln \left(\frac{T_{1}}{T_{2}}\right)}{\left(\mathrm{T}_{1}-\mathrm{T}_{2}\right)+h_{f g_{2}}}
$$

$\mathrm{T}_{1}$ : Temperature of steam at turbine inlet in ${ }^{\circ} \mathrm{C}$
$\mathrm{T}_{2}$ : Temperature of steam at turbine outlet in ${ }^{\circ} \mathrm{C}$
$\mathrm{h}_{\mathrm{fg} 1}$ : Enthalpy of vaporization from saturated steam tables for $\mathrm{T}_{1}{ }^{\circ} \mathrm{C}$
$h_{\mathrm{fg} 2}$ : Enthalpy of vaporization from saturated steam tables for $\mathrm{T}_{2}{ }^{\circ} \mathrm{C}$
5.Rankine Engine Efficiency , $\eta$ Ree.

$$
\eta_{\text {REE }}=\frac{h_{1}-h_{2}}{h_{1}-h_{f 2}} \times 100
$$

$\mathrm{h} 1=\mathrm{hf} 1+\mathrm{x} 1 \mathrm{hfg} 1$
S1 = S2 = Sf1 + x1 sfg $11=$ sf $2+\mathrm{x} 2 \mathrm{Sfg} 2$
Sf2 and sfg2 are found from sat steam tables for $\mathrm{T}_{2}{ }^{\circ} \mathrm{C}$.
Find x 2
$\mathrm{h} 2=\mathrm{hf} 2+\mathrm{x} 2 \mathrm{hfg} 2$
hf2 and hfg2 are found for $\mathrm{T} 2{ }^{\circ} \mathrm{C}$
6. Brake Thermal Efficiency, BTE in \%

$$
B T E=\frac{P d \times 100 \times 3600}{T S C\left(h_{1}-h_{2}\right)}
$$

7.Relative Efficiency $=(\mathrm{BTE} / \mathrm{RE}) \times 100 \%$
calculations table :

| St <br> Pr. | St <br> Pr | TSC | Turbine <br> Output <br> Power | SSc | Rankine <br> Eff | Rank. <br> Engine <br> Eff. | BTE | Rel <br> Eff. |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| inlet | Outlet |  | Pd |  | $\eta$ RE | $\eta$ REE |  | $\eta$ Rel |
| P1 | P2 |  |  |  |  |  |  |  |
| bar | bar | $\mathrm{Kg} / \mathrm{hr}$ | kW | $\mathrm{Kg} /$ <br> kW hr | $\%$ | $\%$ | $\%$ | $\%$ |
|  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |



GRAPHS :
1.TSC Vs TURBINE O/P POWER
2.TSC Vs RANKINE EFFICIENCY
3.TSC Vs BRAKE THERMAL EFFICIENCY

## HEAT BALANCE CALCULATIONS :

## 1.HEAT INPUT AT TURBINE INLET, Qi

$\mathrm{Qi}=\mathrm{TSCX} \mathrm{h} \mathrm{h}_{1} \mathrm{~kJ} / \mathrm{s}$

## 2.HEAT CARRIED AWAY BY CONDENSATE, Qc

Qc = TSC X hfc.
Hfc : Enthalpy of condensate liquid for $\mathrm{Tc}^{\circ} \mathrm{C}$
3.HEAT EQUIVALENT OF WORK DONE , Pd, kJ / s

## 4.HEAT CARRIED AWAY BY CONDENSOR COOLING WATER .

$$
Q w=m_{c}\left(h_{f o}-h_{f i}\right)
$$

$\mathrm{m}_{\mathrm{c}}=$ Cooling water mass flow rate measured from flow meter manometer in $\mathrm{kg} / \mathrm{s}$.

$$
m_{c}=K_{M} \sqrt{\Delta H_{m}} \quad \mathrm{~kg} / \mathrm{s}
$$

$\Delta \mathrm{Hm}=$ cooling water flow meter manometer reading in mm .
$\mathrm{K}_{\mathrm{M}}=2.081 \times 10^{-3} \quad$ constant
$\mathrm{h}_{\mathrm{fi}}$ : Enthalpy of water at inlet temperature of cooling wateer, $\mathrm{T}_{\mathrm{w} 1}{ }^{\circ} \mathrm{C}$
$\mathrm{h}_{\mathrm{fo}}$ : Enthalpy of water at outlet temperature of cooling wateer, $\mathrm{T}_{\mathrm{w} 2}{ }^{\circ} \mathrm{C}$
5.Unaccounted losses, Qun $=1-(2+3+4) \mathrm{kJ} / \mathrm{s}$

All the above are calculated for each of the 4 or 5 sets of readings.

| 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| n | Qi | Qd | Qc | Qw | Qun | Refer <br> below <br> $11)$ | Refer <br> below <br> $12)$ | Refer <br> below <br> $13)$ | Refer <br> below ( <br> $14)$ |
| rpm | $\mathrm{kJ} / \mathrm{s}$ | $\mathrm{kJ} / \mathrm{s}$ | $\mathrm{kJ} / \mathrm{s}$ | $\mathrm{kJ} / \mathrm{s}$ | $\mathrm{kJ} / \mathrm{s}$ | $\%$ | $\%$ | $\%$ | $\%$ |
|  |  |  |  |  |  |  |  |  |  |
|  |  |  |  |  |  |  |  |  |  |

1.Turbine Speed
2.Heat input at turbine inlet
3. Heat Equivalent of work done
4. Heat carried away by condensate
5.Heat carried away by condensate cooling water
6.Unaccounted losses, Qun
7.\% of heat used,$=(\mathrm{Qd} / \mathrm{Qi}) \times 100$
8.\% of heat carried away by condensate, (Qc / Qi ) x 100
$9 . \%$ of heat carried away by condenser cooling water ,
( Qw / Qi ) x 100
$10 . \%$ of unaccounted heat loss (Qun / Qi ) x 100
11. ( Qd / Qi ) x 100
12. ( Qc / Qi ) x 100
13. ( Qw / Qi ) x 100
14. (Qun / Qi) x 100

HEAT BALANCE SHEET


Result : Hence the performance of the steam turbine has been worked out and the graphs drawn .The heat balance sheet has also been drawn.

# ST.ANNE'S <br> COLLEGE OF ENGINEERING AND TECHNOLOGY 

DEPARTMENT OF MECHANICAL
THERMAL ENGINEERING LAB

NAME
ROLL NO.
YEAR
BATCH

# THERMAL ENGINEERING LABORATORY 

ANNA UNIVERSITY CHENNAI

## DEPARTMENT MECHANICAL ENGINEERING

## UNIVERSITY PRACTICAL EXAMINATION

## Allotment of Marks

$\begin{array}{ll}\text { Internal Assessment } & =20 \text { marks } \\ \text { Practical Examination } & =80 \text { marks }\end{array}$

## INTERNAL ASSESSMENT [20 Marks]

Staff should maintain the assessment Register and the Head of the Department should monitor it.

## SPLIT UP OF INTERNAL MARKS

| Record Note | $=10$ marks |
| ---: | :--- |
| Model Exam | $=5$ marks |
| Attendance | $=5$ marks |
|  | $=$ |
| Total | $\underline{20 \text { marks }}$ |

## UNIVERSITY EXAMINATION

The examination will be conducted for 100 marks. Then the marks will be calculated to 80 marks.

## Split up of Practical Examination Marks

| Aim and Procedure | $=25$ marks |
| :--- | :--- |
| Tabulation | $=15$ marks |
| Formulae and Calculation | $=30$ marks |
| Graph and Result | $=20$ marks |
| Viva Voce | $=10$ marks |
| $\quad$ Total | $=100$ marks |
|  |  |

## GENERAL INSTRUCTIONS FOR LABORATORY CLASSES

- Enter the Lab with CLOSED FOOTWEAR.
- Boys should "TUCK IN" the shirts.
- Students should wear uniform only.
- LONG HAIR should be protected, let it not be loose especially near ROTATING MACHINERY.
- Any other machines / equipments should not be operated other than the prescribed one for that day.
- POWER SUPPLY to your test table should be obtained only through the LAB TECHNICIAN.
- Do not LEAN and do not be CLOSE to the rotating components.
- TOOLS, APPARATUS and GUAGE sets are to be returned before leaving the lab.
- HEADINGS and DETAILS should be neatly written
i. Aim of the experiment
ii. Apparatus / Tools / Instruments required
iii. Procedure / Theory / Algorithm / Program
iv. Model Calculations
v. Neat Diagram / Flow charts
vi. Specifications / Designs Details
vii. Tabulations
viii. Graph
ix. Result / discussions.
- Before doing the experiment, the student should get the Circuit / Program approval by the FACULTY-IN-CHARGE.
- Experiment date should be written in the appropriate place.
- After completing the experiment, the answer to the viva-voce questions should be neatly written in the work book.
- Be PATIENT, STEADY, SYSTEMATIC AND REGULAR.


## LIST OF EXPERIMENTS

## HEAT TRANSFER

1. Thermal conductivity measurement by guarded plate method.
2. Thermal conductivity of pipe insulation using lagged pipe apparatus.
3. Natural convection heat transfer from a vertical cylinder.
4. Forced convection inside tube.
5. Heat transfer from pin-fin(natural \& forced convection modes)
6. Determination of Stefan-Boltzmann constant.
7. Determination of emissivity of a gray surface.
8. Effectiveness of Parallel/counter flow heat exchanger.

## REFRIGERATION AND AIR CONDITIONING

1. Determination of COP of a refrigeration system.
2. Experiments on air-conditioning system.
3. Performance test on single/two stage reciprocating air compressor.

## LIST OF EXPERIMENTS BEYOND THE SYLLUBUS

1. Thermal Conductivity of insulating powder.
2. Thermal Conductivity of metal rod.
3. Calorific value determination by Junker's gas calorimeter.

## CONTENTS

| Exp. No. | Name of the Experiment | Page No. |
| :---: | :--- | :--- |
| 1 | Thermal conductivity measurement by guarded <br> plate method |  |
| 2 | Thermal conductivity of pipe insulation using <br> Lagged Pipe Apparatus |  |
| 3 | Natural Convection Heat Transfer from a Vertical <br> Cylinder |  |
| 4 | Forced Convection inside tube |  |
| 5 | Heat transfer from pin-fin(natural \& forced <br> convection modes) |  |
| 6 | Determination of Stefan-Boltzmann constant. |  |
| 7 | Determination of Emissivity of a grey surface |  |
| 8 | Effectiveness of Parallel / Counter Flow Heat <br> Exchanger |  |
| 9 | Determination of COP of a Refrigeration system |  |
| 10 | Experiments on air-conditioning system. |  |
| 11 | Performance test on Two stage reciprocating Air <br> Compressor |  |
| 12 | Thermal Conductivity of insulating powder |  |
| 13 | Thermal Conductivity of metal rod. |  |
| 14 | Calorific value determination by Junker's gas <br> calorimeter |  |
|  | Question Bank |  |

## OBSERVATION:

MINIMUM 40 VOLTS [BOTH]

| S. <br> No. | Volts | Amps | Volts | Amps | $\mathrm{T}_{1}$ | $\mathrm{~T}_{2}$ | $\mathrm{~T}_{3}$ | $\mathrm{~T}_{4}$ | $\mathrm{~T}_{5}$ | $\mathrm{~T}_{6}$ | $\mathrm{~T}_{7}$ | $\mathrm{~T}_{8}$ | $\mathrm{~T}_{9}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1. | 39 | 0.18 | 44.1 | 0.14 | 34.2 | 32.3 | 41.3 | 80.9 | 62.5 | 92.1 | 98.4 | 42.0 | 34.6 |
| 2. | 36 | 0.16 | 41 | 0.13 | 22.6 | 34.5 | 44.4 | 85.0 | 66.4 | 95.6 | 102.8 | 44.9 | 37.4 |
| 3. | 32 | 0.14 | 36 | 0.12 | 23.8 | 35.0 | 44.5 | 82.5 | 65.4 | 92.0 | 98.7 | 45.0 | 38.2 |

## CALCULATION:

| Insulating material | : Asbestos sheet [commercial grade] |
| :---: | :---: |
| Specimen diameter | $=150 \mathrm{~mm}=0.15 \mathrm{~m}$. |
| Area of specimen | $=\pi / 4 \times[0.15]^{2}=0.018 \mathrm{~m}^{2}$. |
| Specimen thickness | $=\Delta \mathrm{L}=12 \mathrm{~mm}=0.012 \mathrm{~m}$. |
| Volts | $=32$ |
| Amps | $=0.14$ Main heater |
| Volts | $=36$ |
| Amps | $=0.12$ Ring heater |
| Heat input q | $\begin{aligned} & =\mathrm{q}[\text { main heater }]+\mathrm{q} \text { [guard heater }] \\ & =[32 \times 0.14 \times 0.86]+[36 \times 0.12 \times 0.86] \\ & =3.85+3.71 \\ & =7.56 \mathrm{kca} / / \mathrm{hr} . \end{aligned}$ |

It should be noted that out of this heat input, ideally only a half will pass through each of the specimens [top and bottom].

$$
\begin{aligned}
\text { Hence } \mathrm{q} & =\mathrm{q} / 2=7.56 / 2=3.78 \mathrm{kcal} / \mathrm{hr} . \\
\Delta \mathrm{T} & =\left\{\left[\mathrm{T}_{4}-\mathrm{T}_{3}\right]+\left[\mathrm{T}_{7}-\mathrm{T}_{8}\right]\right\} / 2 \\
& =\{[82.5-44.5]+[98.7-45.0]\} / 2 \\
& =[38+53.7] / 2 \\
& =91.7 / 2=45.85^{\circ} \mathrm{C} .
\end{aligned}
$$

Thermal conductivity of specimen

$$
\begin{aligned}
\mathrm{K} & =\mathrm{q} \Delta \mathrm{~L} / \mathrm{A} \Delta \mathrm{~T} \\
& =3.78 \times 0.012 / 0.018 \times 45.85 \\
& =0.04536 / 0.8253 \\
& =0.05496 \mathrm{kcal} / \mathrm{hr} \mathrm{~m}^{0} \mathrm{C} .
\end{aligned}
$$

## Experiment Number: 1

## Title of the Experiment: Thermal Conductivity Measurement By Guarded Plate Method

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the thermal conductivity of a poor conducting material, say Asbestos sheet.

## RELEVANT THEORY

Thermal conductivity is a specific property of conducting material which is defined below for a homogeneous solid as the quantity of heat conducted across a unit area normal to the flow direction in unit time and for unit temperature gradient along the flow.
$K=q d L / A d T$
Where,
$\mathrm{q}=$ heat conducted in watts
$\mathrm{dL}=$ thickness [m]
A = Area of conduction heat transfer, $\mathrm{m}^{2}$
$\mathrm{dT}=$ temperature difference across the length $\mathrm{dL}\left[{ }^{0} \mathrm{C}\right]$

## MEASUREMENT:

Experimental measurement of thermal conductivities of solids can be accomplished by a variety of methods, all based on the observation of the temperature gradient across a given area of the material conducting heat at a known rate. Each of these methods has certain unique limitations, and the choice of one over another is governed by the general temperature level at which K is measured, by the physical structure of the material in question and by whether the material is a good or poor conductor.

In measuring the thermal conductivity of poor conductors, the specimens are taken in the form of sheets in order that the heat flow path is short and the conducting area large. [low dL , higher A].

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Guarded plate apparatus | 1 |

## SPECIFICATIONS:

Material
Specimen diameter [d]
Specimen thickness dL
Area of specimen
Heat input

$$
\begin{aligned}
& =\text { Asbestos sheet [commercial grade] } \\
& =150 \mathrm{~mm} \text { or } 0.15 \mathrm{~m} . \\
& =12 \mathrm{~mm} \text { or } 0.012 \mathrm{~m} . \\
& =\pi / 4 \times[0.15]^{2} \mathrm{~m}^{2} \\
& =\text { VI watts }[\mathrm{q}]
\end{aligned}
$$

## b] Guarded Hot Plate method [Solids]

The apparatus consists of a Guarded Hot Plate, the arrangement along with thermocouple positions [T3, T4] across the specimen and T5, T6 guarded heater temperature [only for check] [T1, T2] Top and Bottom pad temperatures.

The panel consists of voltmeter, ammeter, temperature indicator [all digital], dimmer controls, voltmeter and ammeter selector [common switch, thermocouple selector switch].

## c] Operation:

a] Connect the three pin plug top to $230 \mathrm{~V}, 50 \mathrm{~Hz}, 5 \mathrm{Amps}$ power supply socket, dimmers in OFF position.
b] Keep the voltmeter and ammeter switch in 1 position. Turn the dimmer in clockwise and adjust the power input to main heater to any desired value by looking at voltmeter and ammeter.
c] Turn the voltmeter and ammeter switch to position marked 2 and check the voltage \& current are same for ring heater.
d] Allow the unit to stabilize [approx 30 minutes].
e] Note down the temperature indicated by the digital temperature indicator by turning the thermocouple selector switch clockwise step by step $[1,2,3,4,5$, and 6$]$.
f] Repeat the experiment for different power inputs to the heater.
g] Tabulate all the readings and calculate for different conditions.
h] After the experiment is over turn all the dimmer knobs anti clockwise, direction to zero.
i] Disconnect the three pin plug top from the mains.

## CAUTION:

The equipment should be operated between 0 and 150 V .

## d] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Supply a small quantity of energy to the source 'H' [the main heater MH]. |
| 2. | Now adjust the input to the guard heaters such that the temperature is same <br> as that of the main heater |
| 3. | Allow water through the cooling circuit slowly. |
| 4. | Allow $30-60$ minutes for the temperatures to stabilize. |
| 5. | Note down all the parameter |
| 6. | Repeat the experiment at different temperature values by adjusting <br> appropriately the input conditions. |

## e] Result:

Thus the thermal conductivity of a poor conducting material [Asbestos sheet] is determined.
$\mathbf{K}=0.05496 \mathrm{Kcal} / \mathrm{hr} \mathrm{m}{ }^{0} \mathrm{C}$.

## VIVA QUESTIONS

1. Define heat transfer.

Heat transfer can be defined as the transmission of energy from one region to another due to temperature difference.
2. What are the modes of heat transfer?

1. Conduction
2. Convection
3. Radiation.

## 3. What is conduction?

Heat conduction is a mechanism of heat transfer from a region of high temperature to a region of low temperature within a medium [solid, liquid or gases] or different medium in direct physical contact.
4. State Fourier's law of conduction.

The rate of heat conduction is proportional to the area measured normal to the direction of heat flow and to the temperature gradient in that direction.
$\mathrm{Q} \alpha-\mathrm{AdT} / \mathrm{dx}$
$\mathrm{Q}=-\mathrm{kA} \mathrm{dT} / \mathrm{dx}$
Where, A - Area in $\mathrm{m}^{2}$.
$\mathrm{dT} / \mathrm{dx}$ - Temperature gradient, K/m
k - Thermal conductivity, W/mK.

## 5. Define Thermal conductivity.

Thermal conductivity is defined as the ability of a substance to conduct heat.

## Experiment Number: 2

## Title of the Experiment: Thermal Conductivity Of Pipe Insulation Using Lagged Pipe Apparatus

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT:

To plot the radial temperature distribution in the composite cylinder and to determine the thermal conductivity of the pipe insulation.

## THEORY

Consider one dimensional radial heat flow through a hollow cylinder, under steady state conditions.

$$
\mathrm{q}=2 \pi \mathrm{KL}\left[\mathrm{~T}_{1}-\mathrm{T}_{2}\right] / \ln \left[\mathrm{r}_{2} / \mathrm{r}_{1}\right]
$$

Where $T_{1}, T_{2}$ are the inner and outer wall temperature $r_{1}$ and $r_{2}$ are the inner and outer radii of the pipe.
$\mathrm{K}=$ Thermal conductivity of the material.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| SI. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Lagged Pipe Apparatus | 1 |

## b] Description of the Apparatus:

The apparatus consists of a metal pipe with two layers of insulation. An electric heating coil wound on a silica rod is placed at the center. The ends are thickly insulated to prevent heat loss so that, heat flow only in a radial direction. Three thermocouples each are placed at different radii to measure the temperature distribution within the cylinder.

## c] Technical Data:

Location of thermocouples 1, 2, 3 at a radius $=25 \mathrm{~mm}$.
Location of thermocouples 4, 5, 6 at a radius $=37.5 \mathrm{~mm}$.
Location of thermocouples 7, 8, 9 at a radius $=50 \mathrm{~mm}$.
Location of thermocouples $10,11,12$ at a radius $=62.5 \mathrm{~mm}$.
Length of the pipe $\mathrm{L}=500 \mathrm{~mm}$.

## TABULATION

| S.No | Heat Input [watts] |  |  | Temp at Radius$\mathbf{R}_{1}{ }^{0} \mathbf{C}$ |  |  | Temp at Radius$\mathbf{R}_{2}{ }^{0} \mathbf{C}$ |  |  | Temp at Radius$\mathbf{R}_{3}{ }^{0} \mathbf{C}$ |  |  | Temp at Radius$\mathbf{R}_{4}{ }^{0} \mathbf{C}$ |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | V | A | q | $\mathrm{T}_{1}$ | $\mathrm{T}_{2}$ | $\mathrm{T}_{3}$ | $\mathrm{T}_{4}$ | $\mathrm{T}_{5}$ | $\mathrm{T}_{6}$ | $\mathrm{T}_{7}$ | $\mathrm{T}_{8}$ | T9 | $\mathrm{T}_{10}$ | $\mathrm{T}_{11}$ | $\mathrm{T}_{12}$ |
| 1. | 75 | 0.44 | 33 | 111.8 | 133.8 | 102.2 | 37.2 | 41.9 | 37.3 | 35.4 | 39.9 | 39.7 | 31.8 | 30.2 | 30.4 |

MODEL CALCULATION:

$$
\begin{aligned}
\text { At }_{1}= & 25 \mathrm{~mm} \\
& \mathrm{~T}_{1}=(111.8+133.8+102.2) / 3=115.93^{\circ} \mathrm{C} . \\
\text { At } \mathrm{r}_{2}= & 37.5 \mathrm{~mm} \\
& \mathrm{~T}_{2}=(37.2+41.9+37.3) / 3=38.8^{\circ} \mathrm{C}
\end{aligned}
$$

At $\mathrm{r}_{3}=50 \mathrm{~mm}$

$$
\mathrm{T}_{3}=(35.4+39.9+39.7) / 3=38.33^{\circ} \mathrm{C} .
$$

$$
\text { At } \mathrm{r}_{4}=62.5 \mathrm{~mm}
$$

$$
\mathrm{T}_{4}=(31.8+30.2+30.4) / 3=30.8^{\circ} \mathrm{C} \text {. }
$$

$\mathrm{q}=2 \pi \mathrm{KL}\left[\mathrm{T}_{2}-\mathrm{T}_{3}\right] / \ln \left[\mathrm{r}_{3} / \mathrm{r}_{2}\right]$
$\mathrm{q}=\mathrm{V} \times \mathrm{I}=75 \times 0.44=33$.
$\mathrm{K}=\mathrm{q} \ln \left[\mathrm{r}_{3} / \mathrm{r}_{2}\right] / 2 \pi \mathrm{~L}\left[\mathrm{~T}_{2}-\mathrm{T}_{3}\right]$
$=33 \ln [0.05 / 0.0375] / 2 \pi \times 0.5[38.8-38.33]$
$\mathrm{K}=0.1948 \mathrm{~W} / \mathrm{mK}$.

## d] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Connect the equipment to a 230V, 5 amps, and 50 HZ electrical source. |
| 2. | Twin the dimmerstat knob clockwise and fix the heat input to a desired <br> wattage [V x I] |
| 3. | Allow the equipment to stabilize and attain steady state. |
| 4. | Turn the thermocouple selector switch knob clockwise and note down <br> temperature $\mathrm{T}_{1}$ to $\mathrm{T}_{12}$. |
| 5. | Repeat the experiment for different heat inputs. |

## d] Formula:

$$
\begin{aligned}
& \mathrm{q}=2 \pi \mathrm{KL}\left[\mathrm{~T}_{2}-\mathrm{T}_{3}\right] / \ln \left[\mathrm{r}_{3} / \mathrm{r}_{2}\right] \\
& \mathrm{T}_{2}=\text { Temperature at radius } \mathrm{r}_{2}{ }^{0} \mathrm{C} \\
& \mathrm{~T}_{3}=\text { Temperature at radius } \mathrm{r}_{3}{ }^{\circ} \mathrm{C} \\
& \mathrm{r}=\text { radius of the pipe ' } \mathrm{m} ' \\
& \mathrm{~K}=\text { Thermal Conductivity }-\mathrm{W} / \mathrm{m} \mathrm{~K} \\
& \mathrm{~L}=\text { Length of the pipe }- \text { ' } \mathrm{m} \text { ' }
\end{aligned}
$$

## e] Result:

Thus the thermal Conductivity of the pipe insulation is determined. $\mathrm{K}=0.1948 \mathrm{~W} / \mathrm{m} \mathrm{K}$.

## VIVA QUESTIONS

1. What is conduction?

Heat conduction is a mechanism of heat transfer from a region of high temperature to a region of low temperature within a medium [solid, liquid or gases] or different medium in direct physical contact.
2. State Fourier's law of conduction.

The rate of heat conduction is proportional to the area measured normal to the direction of heat flow and to the temperature gradient in that direction.
$\mathrm{Q} \alpha-\mathrm{AdT} / \mathrm{dx}$
$\mathrm{Q}=-\mathrm{kA} \mathrm{dT} / \mathrm{dx}$
Where, A - Area in $\mathrm{m}^{2}$.
$\mathrm{dT} / \mathrm{dx}$ - Temperature gradient, K/m
k - Thermal conductivity, W/mK.
3. Define Thermal conductivity.

Thermal conductivity is defined as the ability of a substance to conduct heat.
4. Write down the equation for conduction of heat through a slab or plane wall.

Heat transfer, $\mathrm{Q}=\Delta \mathrm{T}_{\text {overal }} \mathrm{l} / \mathrm{R}$
Where, $\Delta \mathrm{T}=\mathrm{T}_{1}-\mathrm{T}_{2}$
$\mathrm{R}=\mathrm{L} / \mathrm{kA}$ - Thermal resistance of slab
L - Thickness of slab
K - Thermal conductivity of slab
A - Area
5. Write down the equation for conduction of heat through a hollow cylinder.

Heat transfer, $\mathrm{Q}=\Delta \mathrm{T}_{\text {overal }} / \mathrm{R}$
Where
$\Delta \mathrm{T}=\mathrm{T}_{1}-\mathrm{T}_{2}$
$\mathrm{R}=1 / 2 \pi \mathrm{Lk}$ in $\left[\mathrm{r}_{2} / \mathrm{r}_{1}\right]-$ Thermal resistance of slab.
L - Length of cylinder
k - Thermal conductivity
$\mathrm{r}_{2}$ - Outer radius
$\mathrm{r}_{1}$ - Inner radius
6. What are the factors affecting the thermal conductivity?
a. Moisture
b. Density of material
c. Pressure
d. Temperature
e. Structure of material.

TABULATION

| $\begin{gathered} \text { S. } \\ \text { No } \end{gathered}$ | Volt <br> Meter <br> Reading <br> [Volts] | Am- <br> Meter <br> Reading <br> [I] <br> Amps | $\begin{aligned} & \text { In } \\ & \text { Put } \\ & \text { Watts } \end{aligned}$ | Thermocouple Location in ${ }^{0} \mathrm{C}$ |  |  |  |  |  |  | $\mathrm{T}_{\mathrm{s}}{ }^{0} \mathrm{C}$ | $\begin{gathered} {\left[\mathrm{T}_{8}\right]} \\ \mathrm{T}_{\mathrm{a}} \\ { }^{\circ} \mathrm{C} . \end{gathered}$ | $\begin{gathered} \mathrm{T}_{\mathrm{mf}} \\ {\left[\mathrm{~T}_{\mathrm{s}}+\mathrm{T}_{\mathrm{T}}\right] / 2} \\ { }^{0} \mathrm{C} \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | 1 | 2 | 3 | 4 | 5 | 6 | 7 |  |  |  |
| 1. | 78.125 | 0.64 | 50 | 53 | 56 | 62 | 68 | 66 | 71 | 64 | 62.857 | 38 | 50.429 |
| 2. | 84.337 | 0.83 | 70 | 83 | 92 | 101 | 107 | 101 | 110 | 103 | 99.571 | 38 | 68.736 |
| 3. | 91.836 | 0.98 | 90 | 98 | 102 | 121 | 128 | 123 | 133 | 113 | 116.857 | 39 | 77.929 |

## CALCULATION:

[1] Experiment heat transfer co-efficient [Average]

$$
\begin{aligned}
& \mathrm{Q}=\mathrm{hA}[\Delta \mathrm{~T}] \text { Watts. } \\
& \mathrm{Q}=\mathrm{hA}\left[\mathrm{~T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}\right] \text { Watts. } \\
& \mathrm{h}=\mathrm{Q} / \mathrm{A}\left[\mathrm{~T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}\right] \mathrm{W} / \mathrm{m}^{2} \mathrm{~K} . \\
& \mathrm{h}=50 / \pi \times \mathrm{d} \times 1\left[\mathrm{~T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}\right]=50 / \pi \times 0.031 \times 0.5[62.857-38] \\
& \mathrm{h}_{\text {avg }}=41.309 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K} .
\end{aligned}
$$

[2] Local heat transfer co-efficient:
$\mathrm{h}_{\mathrm{exp}}[$ local $]=\mathrm{Q} / \mathrm{A}\left[\mathrm{T}_{\mathrm{x}}-\mathrm{T}_{\mathrm{a}}\right]$.
$\mathrm{h}_{1}=50 / 0.5 \times 0.031 \times \pi \times[53-38]=68.454 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{h}_{2}=50 / 0.5 \times 0.031 \times \pi \times[56-38]=57.045 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{h}_{3}=50 / 0.5 \times 0.031 \times \pi \times[62-38]=42.784 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{h}_{4}=50 / 0.5 \times 0.031 \times \pi \times[68-38]=34.227 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{h}_{5}=50 / 0.5 \times 0.031 \times \pi \times[66-38]=36.672 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{h}_{6}=50 / 0.5 \times 0.031 \times \pi \times[71-38]=31.115 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{h}_{7}=50 / 0.5 \times 0.031 \times \pi \times[64-38]=39.493 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
[3] Theoretical heat transfer co-efficient. $\mathrm{h}_{\text {theo }}$ [Average]

$$
\begin{aligned}
\mathrm{Gr}=[ & {[\mathrm{g} \mathrm{x} \mathrm{~L}} \\
& \times \beta \times \Delta \mathrm{T}] / v^{2} \\
& \beta=1 / \mathrm{T}_{\mathrm{mf}} \text { in } \mathrm{K}=1 /[50.429+273]=3.092 \times 10^{-3}{ }^{0} \mathrm{~K}^{-1} . \\
& \Delta \mathrm{T}=\left[\mathrm{T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}\right]=24.857{ }^{\circ} \mathrm{C} . \\
& \mathrm{T}_{\mathrm{f}}=\left[\mathrm{T}_{\mathrm{s}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[62.857+38] / 2=50.429=50^{\circ} \mathrm{C} .
\end{aligned}
$$

Properties of air at $50^{\circ} \mathrm{C}$

$$
\begin{array}{lll}
\rho=1.093 \mathrm{~kg} / \mathrm{m}^{3} ; & \mathrm{Pr}=0.698 ; & \mathrm{L}=0.5 \mathrm{~m} ; \\
\nu=17.95 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} ; & \mathrm{K}=0.02826 \mathrm{~W} / \mathrm{m} \mathrm{~K} ; & \mathrm{g}=9.81 \mathrm{~m} / \mathrm{sec}^{2} .
\end{array}
$$

$\mathrm{Gr}=\left[\mathrm{g} \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \beta \times \Delta \mathrm{T}\right] / v^{2}=\left\{9.81 \times[0.5]^{3} \times 3.092 \times 10^{-3} \times 24.857\right\} /\left[17.95 \times 10^{-6}\right]^{2}$
$\mathrm{Gr}=0.0942 / 3.222 \times 10^{-10}=2.924 \times 10^{8}$.
$\operatorname{GrPr}=\left[2.924 \times 10^{8}\right] \times 0.698=2.040 \times 10^{8}$.
$\mathrm{Nu}=0.59\left[2.040 \times 10^{8}\right]^{0.25}=70.511$
$70.511=\mathrm{hx} 0.5 / 0.02826$
$\mathrm{h}_{\text {theo }}=3.985 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.

## Experiment Number: 3

## Title of the Experiment: Natural Convection Heat Transfer from a Vertical Cylinder

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine surface heat transfer co-efficient, local heat transfer co-efficient along the length of the tube and also to draw the graph between local heat transfer co-efficient and distance along the length of the tube.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Natural convection-vertical cylinder <br> apparatus | 1 |

## b] Theory

When a hot body is kept in a still air, heat is transferred to the surrounding by natural convection, the fluid layer in contact with the hot surface gets heated, rises up due to decrease in its density and the cold fluid rushes into take its place. The process is continuous and heat transfer takes place due to relative motion of hot and cold fluid. The surface heat transfer co-efficient of a system transferring heat by natural convection depends upon its shape, dimension, orientation and also the temperature difference between the surface and the fluid.

## c] Apparatus Description

The apparatus consists of a Brass tube fitted in a rectangular duct in a vertical fashion. The duct is open at the top and bottom and forms an enclosure and serves the purpose of undisturbed surrounding. One side of the duct is made up of Perspex sheet for visualization. An electric heating element is kept in the vertical tube which in turn heats the tube to the surrounding air by natural convection. The surface temperature of the vertical tube is measured by seven thermocouple wires. The tube surface is polished to minimize the radiation losses. The temperature of the tube measured by a temperature indicator.

## Specification:

[1] Diameter of the tube [d]
$=31 \mathrm{~mm}$.
[2] Length of the tube [1]
$=500 \mathrm{~mm}$.
[3] Duct size
$=200 \mathrm{~mm} \times 200 \mathrm{~mm} \times 750 \mathrm{~mm}$
[4] Theoretical Local heat transfer co-efficient. $\mathrm{h}_{\text {theo }}$ [Local]
Gr [local] $=\left[\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T}\right] / \nu^{2}$
[i] $\mathrm{T}_{\mathrm{mf}}[$ Local $]=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[53+38] / 2=45.5^{\circ} \mathrm{C}$.
Properties of air at $45.5^{\circ} \mathrm{C}$

$$
\begin{array}{lll}
\rho=1.109 \mathrm{~kg} / \mathrm{m}^{3} ; & \mathrm{Pr}=0.685 ; & \mathrm{L}_{\mathrm{x}}=0.01 \mathrm{~m} ; \\
\nu=17.505 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} & \mathrm{~K}=0.02795 \mathrm{~W} / \mathrm{mK} ; & \mathrm{g}=9.81 \mathrm{~m}^{2} / \mathrm{s}
\end{array}
$$

$\mathrm{Gr}=\left[\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T}\right] / v^{2} \quad \beta=1 / 318.5=3.140 \times 10^{-3}$. $=9.81 \times 3.140 \times 10^{-3} \times 0.01^{3} \times[53-38] /\left[17.505 \times 10^{-6}\right]^{2}$
$\mathrm{Gr}=1.508 \times 10^{3}$.
$\mathrm{Gr} \mathrm{Pr}=\left[1.508 \times 10^{3}\right][0.685]$
$\operatorname{Gr} \operatorname{Pr}=1.033 \times 10^{3}$.
$10^{4} \leq \mathrm{Gr} \operatorname{Pr} \leq 10^{9} \quad \mathrm{Nu}=[0.59][\mathrm{Gr} \operatorname{Pr}]{ }^{0.25}$ $=[0.59]\left[1.033 \times 10^{3}\right]^{0.25}=3.345$
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$3.345=$ hL [0.01] / 0.02795
$\mathrm{h}_{\mathrm{L}_{1}}=9.349 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
[ii] $\quad \mathrm{T}_{\mathrm{mf}}[$ Local $]=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[56+38] / 2=47^{0} \mathrm{C}$.
Properties of air at $47^{\circ} \mathrm{C}$

$$
\rho=1.104 \mathrm{~kg} / \mathrm{m}^{3} ; \quad \operatorname{Pr}=0.689 ; \quad \mathrm{L}_{\mathrm{x}}=0.05 \mathrm{~m} ;
$$

$v=17.653 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} \quad \mathrm{K}=0.02805 \mathrm{~W} / \mathrm{mK} ; \beta=1 / 320=3.125 \times 10^{-3}$
$\mathrm{Gr}=\left[\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T}\right] / \nu^{2}$

$$
=9.81 \times 3.125 \times 10^{-3} \times 0.05^{3} \times[56-38] /\left[17.653 \times 10^{-6}\right]^{2}
$$

$\mathrm{Gr}=2.214 \times 10^{5}$.
$\mathrm{Gr} \operatorname{Pr}=\left[2.214 \times 10^{5}\right] \times[0.689]=1.525 \times 10^{5}$
$\mathrm{Nu}=[0.59]\left[1.525 \times 10^{5}\right]^{0.25}=11.66$.
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$11.66=[$ hL x 0.05$] / 0.02805$
$\mathrm{h}_{\mathrm{L}_{2}}=6.54 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
[iii] $\mathrm{T}_{\mathrm{mf}}=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[62+38] / 2=50^{\circ} \mathrm{C}$.
Properties of air at $50^{\circ} \mathrm{C}$

$$
\rho=1.093 \mathrm{~kg} / \mathrm{m}^{3} ; \quad \operatorname{Pr}=0.698 ; \quad \mathrm{L}_{\mathrm{x}}=0.1 \mathrm{~m} ;
$$

$v=17.95 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} \quad \mathrm{K}=0.02826 \mathrm{~W} / \mathrm{mK} ; \beta=3.0296 \times 10^{-3}$
$\mathrm{Gr}=\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T} / \nu^{2}$
$\mathrm{Gr}=9.81 \times 3.096 \times 10^{-3} \times 0.1^{3} \times[62-38] /\left[17.95 \times 10^{-6}\right]^{2}=2.262 \times 10^{6}$.
$\mathrm{Gr} \operatorname{Pr}=\left[2.262 \times 10^{6}\right] \times[0.698]=1.579 \times 10^{6}$.
$\mathrm{Nu}=[0.59]\left[1.579 \times 10^{6}\right]^{0.25}=20.91$.
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$20.91=[h L \times 0.1] / 0.02826$
$\mathrm{h}_{\mathrm{L} 3}=5.909 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
[iv] $\mathrm{T}_{\mathrm{mf}}=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[68+38] / 2=53^{\circ} \mathrm{C}$.
Properties of air at $53^{\circ} \mathrm{C}$

$$
\begin{array}{lll}
\rho=1.083 \mathrm{~kg} / \mathrm{m}^{3} ; & \mathrm{Pr}=0.697 ; & \mathrm{L}_{x}=0.2 \mathrm{~m} ; \\
v=18.26 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} & \mathrm{~K}=0.02847 \mathrm{~W} / \mathrm{mK} ; & \beta=3.067 \times 10^{-3}
\end{array}
$$

[4] Number of Thermocouples =7 and are shown as [1] - [7] and as marked on temperature indicator switch.
[5] Thermocouple number 8 reads the temperature of the air in the duct.
[6] Temperature indicator $0-300^{\circ} \mathrm{C}$. Multichannel type, calibrated for chromel alumel thermo couples.
[7] Ammeter

$$
\begin{aligned}
& =[0-2 \mathrm{~A}] \\
& =[0-100 / 200 \mathrm{~V}] \\
& =2 \mathrm{~A} / 230 \mathrm{Volt} . \\
& =400 \text { Watts }
\end{aligned}
$$

[8] Voltmeter
[9] Dimmer start
[10] Heater - cartridge type

## d] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Switch on the supply and adjust the dimmerstat to obtain the required heat <br> input. |
| 2. | Wait till the fairly steady state is reached, which is confirmed from <br> temperature readings $\left[\mathrm{T}_{1}\right.$ to $\left.\mathrm{T}_{7}\right]$. |
| 3. | Note down surface temperature at various points. |
| 4. | Note the Ambient Temperature $\left[\mathrm{T}_{8}\right]$. |
| 5. | Repeat the experiment at different heat inputs. |

## Precautions:

[1] Do not exceed 100 Watts.
[2] Operate the change over selector switch gently from position [1] to [8].

## Formula Used:

$[1] \mathrm{T}_{\mathrm{s}}=\left[\mathrm{T}_{1}+\mathrm{T}_{2}+\mathrm{T}_{3}+\mathrm{T}_{4}+\mathrm{T}_{5}+\mathrm{T}_{6}+\mathrm{T}_{7}\right] / 7{ }^{\circ} \mathrm{C}$
Where $\mathrm{T}_{1}, \mathrm{~T}_{2}, \mathrm{~T}_{3}$, $\qquad$ $\mathrm{T}_{7}$ are temperature at locations 1, 2 --7
Mean film temperature $\left[\mathrm{T}_{\mathrm{mf}}\right]=\left[\mathrm{T}_{\mathrm{s}}+\mathrm{T}_{\mathrm{a}}\right] / 2$.
Where $\mathrm{T}_{\mathrm{s}}=$ Average surface temperature in ${ }^{\circ} \mathrm{C}$.
$\mathrm{T}_{\mathrm{a}}=$ Ambient Temperature in ${ }^{\circ} \mathrm{C}$.
Experiment heat transfer co-efficient [Average]
$\mathrm{Q}=\mathrm{hA}[\Delta \mathrm{T}]$ Watts.
Where $\mathrm{h}=$ Experimental convective heat transfer co-efficient [Average] W/m²K.
A $=$ Area of heat transfer $\pi \mathrm{d} \mathrm{m}^{2}$.
$\Delta \mathrm{T}=\mathrm{T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}$ in ${ }^{\circ} \mathrm{C}$.
$\mathrm{T}_{\mathrm{s}}=$ Surface temperature in ${ }^{\circ} \mathrm{C}$.
$\mathrm{T}_{\mathrm{a}}=$ Ambient temperature in ${ }^{\circ} \mathrm{C}$.
$\mathrm{Q}=$ Average rate of heat transfer by convection in [Watts].
$h_{\exp }[$ average $]=\mathrm{Q} / \mathrm{As}\left[\mathrm{T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}\right] \mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{Gr}=\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T} / v^{2}$
$\mathrm{Gr}=9.81 \times 3.067 \times 10^{-3} \times 0.2^{3} \times[68-38] /\left[18.26 \times 10^{-6}\right]^{2}=21.659 \times 10^{6}$.
$\mathrm{Gr} \operatorname{Pr}=\left[21.659 \times 10^{6}\right] \times[0.697]=15.096 \times 10^{6}$.
$\mathrm{Nu}=[0.59]\left[15.096 \times 10^{6}\right]^{0.25}=36.776$.
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$36.776=\left[h_{L 4} \times 0.2\right] / 0.2847$
$\mathrm{h}_{\mathrm{L} 4}=5.235 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$[\mathrm{v}] \mathrm{T}_{\mathrm{mf}}=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[66+38] / 2=52^{\circ} \mathrm{C}$.
Properties of air at $52^{\circ} \mathrm{C}$

$$
\begin{array}{lll}
\rho=1.086 \mathrm{~kg} / \mathrm{m}^{3} ; & \operatorname{Pr}=0.698 ; & \mathrm{L}_{\mathrm{x}}=0.3 \mathrm{~m} ; \\
\nu=18.15 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} & \mathrm{~K}=0.02840 \mathrm{~W} / \mathrm{mK} ; & \beta=3.077 \times 10^{-3}
\end{array}
$$

$\mathrm{Gr}=\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T} / \nu^{2}$
$\mathrm{Gr}=9.81 \times 3.077 \times 10^{-3} \times 0.3^{3} \times[66-38] /\left[18.15 \times 10^{-6}\right]^{2}=69.28 \times 10^{6}$.
$\mathrm{Gr} \operatorname{Pr}=\left[69.28 \times 10^{6}\right][0.698]=48.36 \times 10^{6}$.
$\mathrm{Nu}=[0.59]\left[48.36 \times 10^{6}\right]^{0.25}=49.201$.
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$49.201=h_{L_{5}} \times 0.3 / 0.02840$
$\mathrm{h}_{\mathrm{L} 5}=4.658 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
$[\mathrm{vi}] \mathrm{T}_{\mathrm{mf}}=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[71+38] / 2=54.5^{\circ} \mathrm{C}$.
Properties of air at $54.5^{\circ} \mathrm{C}$

$$
\begin{array}{lcc}
\rho=1.078 \mathrm{~kg} / \mathrm{m}^{3} ; & \operatorname{Pr}=0.697 ; & \mathrm{L}_{\mathrm{x}}=0.4 \mathrm{~m} ; \\
\nu=18.41 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} & \mathrm{~K}=0.02858 \mathrm{~W} / \mathrm{mK} ; & \beta=3.053 \times 10^{-3}
\end{array}
$$

$\mathrm{Gr}=\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T} / \nu^{2}$
$\mathrm{Gr}=9.81 \times 3.053 \times 10^{-3} \times 0.4^{3} \times[71-38] /\left[18.41 \times 10^{-6}\right]^{2}=186.63 \times 10^{6}$.
$\mathrm{Gr} \operatorname{Pr}=\left[186.63 \times 10^{6}\right] \times[0.697]=130.08 \times 10^{6}$.
$\mathrm{Nu}=[0.59]\left[130.08 \times 10^{6}\right]^{0.25}=63.01$.
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$\mathrm{Nu}=\left[\mathrm{hL} \mathrm{x}_{\mathrm{x}} 0.4\right] / 0.02858$
$63.01=[$ hL x 0.4$] / 0.2858$
$\mathrm{h}_{\mathrm{L}_{6}}=4.502 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
[vii] $\mathrm{T}_{\mathrm{mf}}=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2=[64+38] / 2=51^{\circ} \mathrm{C}$.
Properties of air at $51^{\circ} \mathrm{C}$

$$
\rho=1.090 \mathrm{~kg} / \mathrm{m}^{3} ; \quad \operatorname{Pr}=0.698 ; \quad \mathrm{L}_{\mathrm{x}}=0.49 \mathrm{~m} ;
$$

$$
v=18.052 \times 10^{-6} \mathrm{~m}^{2} / \mathrm{s} \quad \mathrm{~K}=0.02833 \mathrm{~W} / \mathrm{mK} ; \beta=3.086 \times 10^{-3}
$$

$\mathrm{Gr}=\mathrm{g} \times \beta \times \mathrm{L}_{\mathrm{x}}{ }^{3} \times \Delta \mathrm{T} / \nu^{2}$
$\mathrm{Gr}=9.81 \times 3.086 \times 10^{-3} \times 0.49^{3} \times[64-38] /\left[18.052 \times 10^{-6}\right]^{2}=284.136 \times 10^{6}$.
$\mathrm{Gr} \operatorname{Pr}=\left[284.136 \times 10^{6}\right] \times 0.698=198.33 \times 10^{6}$.
$\mathrm{Nu}=[0.59] \times\left[198.33 \times 10^{6}\right]^{0.25}=70.16$
$\mathrm{Nu}=\mathrm{h}_{\mathrm{L}} \mathrm{L}_{\mathrm{x}} / \mathrm{K}$
$\mathrm{Nu}=[\mathrm{h} \mathbf{x} 0.49] / 0.02833$
$70.16=$ hL x $0.49 / 0.2833$
$\mathrm{h}_{\mathrm{L} 7}=4.048 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$.
[2] Local heat transfer co-efficient:

$$
\mathrm{h}_{\mathrm{exp}}[\text { local }]=\mathrm{Q} / \mathrm{A}\left[\mathrm{~T}_{\mathrm{x}}-\mathrm{T}_{\mathrm{a}}\right] .
$$

Where $\mathrm{T}_{\mathrm{x}}=$ Temperature at locations 1 to 7 in ${ }^{0} \mathrm{~K}$.
The local heat transfer co-efficient $h_{1}, h_{2}, h_{3}$, $\qquad$ $\mathrm{h}_{7}$ can be calculated from the above equation.
[3] Theoretical heat transfer co-efficient [Average].
Using free convection correlation for vertical cylinder.
$\mathrm{Nu}=0.59[\mathrm{Gr} \text {. } \mathrm{Pr}]^{0.25}$ for $10^{4} \leq \mathrm{Gr}$. $\operatorname{Pr} \leq 10^{9}$.
$=0.13[\mathrm{Gr} . \operatorname{Pr}]^{1 / 3}$ for $10^{9} \leq \mathrm{Gr} . \operatorname{Pr} \leq 10^{12}$.
$\mathrm{Nu}=$ Nusselt Number.
$\mathrm{Gr}=$ Grashof Number.
$\operatorname{Pr}=$ Prandtl Number.

Grash of Number $=\mathrm{g} \times \mathrm{L}^{3} \times \beta \times \Delta \mathrm{T} / \nu^{2}$
Where $\mathrm{g}=$ Acceleration due to gravity $=9.81 \mathrm{~m} / \mathrm{s}^{2}$.
$\mathrm{L}=$ Characteristics dimension in meters, Here $\mathrm{L}=0.5 \mathrm{~m}$.
$\beta=$ Co-efficient of thermal expansion for the fluid
$\beta=1 / T_{\mathrm{f}}$ in K .
$\Delta \mathrm{T}=$ Temperature difference in ${ }^{0} \mathrm{~K}=\left[\mathrm{T}_{\mathrm{s}}-\mathrm{T}_{\mathrm{a}}\right]$.
$v=$ Kinematic viscosity of the air at mean film temperature, $\mathrm{m}^{2} / \mathrm{s}$.
[from the HMT Data book].
$\mathrm{p}_{\mathrm{r}}=$ Prandtl Number of air at $\mathrm{T}_{\mathrm{mf}}$ [from the HMT Data book].
$\mathrm{Nu}=\mathrm{hL} / \mathrm{K}$.
Where $\mathrm{h}=$ Convective heat transfer co-efficient in $\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{L}=$ Characteristic dimension in $\mathrm{m} . \mathrm{L}=0.5 \mathrm{~m}$.
$\mathrm{K}=$ Thermal conductivity of air at $\mathrm{T}_{\mathrm{mf}}$ [from HMT Data book]
[4] Theoretical Local heat transfer co-efficient. $\mathrm{h}_{\text {theo }}$ [Local]
Gr [local] $=\mathrm{g} \mathrm{x}_{\mathrm{L}}{ }^{3} \times \beta \times \Delta \mathrm{T} / \nu^{2}$
Where $L_{x}=L_{1}, L_{2}, L_{3}, \ldots \ldots . . . . L_{7}$ distance from the bottom of the tube in ' m '
$\operatorname{Pr}=$ Prandtl Number [ $\left.\mathrm{T}_{\mathrm{mf}}\right]$
$\mathrm{T}_{\mathrm{mf}}[$ local $]=\left[\mathrm{T}_{\mathrm{x}}+\mathrm{T}_{\mathrm{a}}\right] / 2$
$\mathrm{T}_{\mathrm{x}}=$ Temperature at point 1 to 7 .
$v=$ Kinematic viscosity at $\mathrm{T}_{\mathrm{mf}}$ in $\mathrm{m}^{2} / \mathrm{s}$.
$\Delta \mathrm{T}=$ Temperature difference $-\left[\mathrm{T}_{\mathrm{x}}-\mathrm{T}_{\mathrm{a}}\right]$ in ${ }^{0} \mathrm{~K}$.
$\mathrm{Nu}\left[\right.$ Local $=\mathrm{h}_{1} \mathrm{~L}_{1} / \mathrm{K}$.
Where $\mathrm{h}_{1}=$ Local convective heat transfer co-efficient at point 1 to $7 \mathrm{in} \mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$.
$\mathrm{L}_{1}=$ Characterstics dimension in m .
$\mathrm{K}=$ Thermal conductivity of air at $\mathrm{T}_{\mathrm{mf}}$ in $\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$.

## Result:

Thus the average surface heat transfer co-efficient and local heat transfer co-efficient along the length of the tube are determined and also the graph $b / w$ local heat transfer coefficient and the distance along the height of the tube is drawn. The results are tabulated.

Experiment local heat transfer co-efficient:

| S. No. | Input <br> [watts] | Local heat transfer co-efficient [ $\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}$ ] [experiment] |  |  |  |  |  |  | Average <br> Surface <br> Heat <br> Transfer Coefficient $\left[\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}\right]$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\mathrm{h}_{1}$ | $\mathrm{h}_{2}$ | $\mathrm{h}_{3}$ | $\mathrm{h}_{4}$ | $\mathrm{h}_{5}$ | $\mathrm{h}_{6}$ | $\mathrm{h}_{7}$ |  |
| 1. | 50 | 68.45 | 57.045 | 42.784 | 34.227 | 36.672 | 31.17 | 39.493 | 41.309 |

Theortical Local heat transfer co-efficient:

| S. No. | Input <br> [watts] | [Theortical]Local heat transfer co-efficient experiment$\left[\mathrm{w} / \mathrm{m}^{2} \mathrm{~K}\right]$ |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\mathrm{h}_{1}$ | $\mathrm{h}_{2}$ | $\mathrm{h}_{3}$ | $\mathrm{h}_{4}$ | $\mathrm{h}_{5}$ | $\mathrm{h}_{6}$ | $\mathrm{h}_{7}$ |  |
| 1. | 50 | 9.349 | 6.54 | 5.909 | 5.235 | 4.658 | 4.502 | 4.048 | 3.983 |

## VIVA QUESTIONS

1. What is meant by free or natural convection?

It is fluid motion is produced due to change in density resulting from temperature gradients, the mode of heat transfer is said to be free or natural convection.
2. Define Grashof number [Gr].

It is defined as the ratio of product of inertia force and buoyancy force to the square of viscous force. $\mathrm{Gr}=$ Inertia force x Buoyancy force / [Viscous force] ${ }^{2}$
3. Define Stanton number [St].

It is the ratio of Nusselt number to the product of Reynolds number and
Prandtl number. $\mathrm{St}=\mathrm{Nu} / \mathrm{Re} \times \mathrm{Pr}$.
4. What is meant by Newtonion and non-newtonion fluids?

The fluids which obey the Newton's law of viscosity are called Newtonion fluids and those which do not obey are called no-newtonion fluids.
5. What is meant by laminar flow?

Laminar flow: Laminar flow is sometimes called stream line flow. In this type of flow, the fluid moves in layers and each fluid particle follows a smooth continuous path. The fluid particles in each layer remain in an orderly sequence without mixing with each other.

## Experiment Number: 4

## Title of the Experiment: Forced Convection inside Tube Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the heat transfer coefficient on the given Forced Convection inside tube Apparatus.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| S. No. | Facilities Required | Quantity |
| :---: | :---: | :---: |
| 1. | Forced Convection inside tube Apparatus | 1 |

## b] Description

The experimental setup consists of a tube through which air is sent in by a blower. The test section consists of a long electrical surface heater on the tube which serves as a constant heat flux source on the flowing medium. The inlet and outlet temperatures of the flowing air are measured by thermocouples and also the temperatures at several locations along the surface heater from which on average temperature can be obtained. An orifice meter in the tube is used to measure the air flow rate with a ' $U$ tube water manometer.

An ammeter and a voltmeter are provided to measure the power input to the heater. A power regulator is provided to vary the power input to heater.
A multipoint digital temperature indicator is provided to measure the above thermocouples input.

A valve is provided to regulate the flow rate of air.
c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Switch on the main. |
| 2. | Switch on the blower. |
| 3. | Adjust the regulator to any desired power into input to heater. |
| 4. | Adjust the position of the valve to any desired flow rate of air. |
| 5. | Wait till steady state temperature is reached. |
| 6. | Note manometer reading $\mathrm{h}_{1}$ and $\mathrm{h}_{2}$. |
| 7. | Note temperatures along the tube. Note air inlet and outlet temperature. |
| 8. | Note voltmeter and ammeter reading. |
| 9. | Adjust the position of the valve and vary the flow rate of air and repeat the <br> experiment. |
| 10. | For various valve openings and for various power inputs the readings may <br> be taken to repeat the experiments. |

TABULATION:

| S.No. | Voltage [V] <br> [Volts] | Current <br> [A] <br> [Amps] | Inlet <br> Temperature <br> of air $\left[\mathrm{T}_{1}\right]$ <br> $\left[{ }^{0} \mathrm{C}\right]$ | Outlet Temperature of air [ $\mathrm{T}_{6}$ ] $\left[{ }^{0} \mathrm{C}\right]$ | Temperature along the duct |  |  |  | Manometer reading |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  | $\begin{gathered} \mathrm{T}_{2} \\ {\left[{ }^{\circ} \mathrm{C}\right]} \end{gathered}$ | $\begin{gathered} \hline \mathrm{T}_{3} \\ {\left[{ }^{\circ} \mathrm{C}\right]} \end{gathered}$ | $\begin{gathered} \mathrm{T}_{4} \\ {\left[{ }^{\circ} \mathrm{C}\right]} \end{gathered}$ | $\begin{gathered} \mathrm{T}_{5} \\ {\left[{ }^{\circ} \mathrm{C}\right]} \end{gathered}$ | $\begin{aligned} & \mathrm{h}_{1} \\ & {[\mathrm{~cm}]} \end{aligned}$ | $\begin{aligned} & \mathrm{h}_{2} \\ & {[\mathrm{~cm}]} \end{aligned}$ |
| 1 | 50 | 1 | 35 | 38 | 42 | 45 | 46 | 47 | 9 | 19 |

## MODEL CALCULATIONS:

## EXPERIMENTAL METHOD:

$$
\begin{aligned}
\mathrm{PI}=\mathrm{V} \times \mathrm{I} & =50 \text { watts } \\
\mathrm{VI} & =\mathrm{h} \times \mathrm{A} \times \Delta \mathrm{t} \\
\Delta \mathrm{t} & =\text { Average temperature of heater }- \text { Average temperature of air } \\
\Delta \mathrm{t} & =45-36.5 \\
\Delta \mathrm{t} & =8.5^{\circ} \mathrm{C}
\end{aligned}
$$

Average temperature of heater $=T_{2}+T_{3}+T_{4}+T_{5} / 4=42+45+46+47 / 4=45^{\circ} \mathrm{C}$.
Average temperature of air $=\mathrm{T}_{1}+\mathrm{T}_{6} / 2=35+33 / 2=36.5^{\circ} \mathrm{C}$.

$$
A=\pi \times d \times 1 \quad \text { Diameter of tube } d=0.04 m
$$

$$
\text { Length of the tube } 1=0.5 \mathrm{~m}
$$

$$
\begin{aligned}
& \text { A = Area of heat transfer } \\
& \mathrm{A}=3.14 \times 0.04 \times 0.5 \\
& \mathrm{~A}=0.0634 \mathrm{~m}^{2} \text {. } \\
& \mathrm{VI}=\mathrm{hx} \mathrm{Ax} \Delta \mathrm{t} \\
& 50=\text { h X } 0.0634 \text { X } 8.5 \\
& \mathrm{~h}=92.782 \mathrm{~W} / \mathrm{m}^{2} \mathrm{C} \text {. }
\end{aligned}
$$

$h_{0}=\left[h_{1}-h_{2}\right] \times\left[\rho_{w} / \rho_{\mathrm{a}}\right] \mathrm{m}^{3} / \mathrm{sec}$
$\rho_{\mathrm{w}}=1000 \mathrm{~kg} / \mathrm{m}^{3}$
$\rho_{\mathrm{a}}=1.16 \mathrm{~kg} / \mathrm{m}^{3}$.

## [1] EXPERIMENTAL METHOD:

$\mathrm{VI}=\mathrm{hA} \Delta \mathrm{t}$
Where,
$\Delta t=$ Average temperature of heater - Average temperature of air $\left[{ }^{\circ} \mathrm{C}\right]$.
$\mathrm{A}=\pi \mathrm{dl}$
A = Area of heat transfer.
$\mathrm{d}=$ diameter of the tube $=0.04 \mathrm{~m}$
$\mathrm{l}=$ length of the tube $=0.5 \mathrm{~m}$.
$\mathrm{h}=$ heat transfer co-efficient [W/m $\left.{ }^{2} \mathrm{C}\right]$
$\mathrm{VI}=$ Power input to heater.
[2] THEORITICAL METHOD:
$\mathrm{Q}=\mathrm{C}_{\mathrm{d}} \times \mathrm{a}_{1} \times \mathrm{a}_{2} \sqrt{ } 2 \mathrm{gh}_{\mathrm{o}} / \sqrt{ } \mathrm{a}_{1}{ }^{2}-\mathrm{a}_{2}{ }^{2} \quad \mathrm{~m}^{3} / \mathrm{sec}$.
$\mathrm{h}_{\mathrm{o}}=$ head of air causing the flow.
$=\left[h_{1}-\mathrm{h}_{2}\right] \times\left[\rho_{\mathrm{w}} / \rho_{\mathrm{a}}\right]$
$\rho_{w}=$ Density of water $=1000 \mathrm{~kg} / \mathrm{m}^{3}$.
$\rho_{a}=$ Density of air $=1.16 \mathrm{~kg} / \mathrm{m}^{3}$.
$h_{1}, h_{2}=$ Manometer reading in $m$.
$\mathrm{a}_{1}=$ Area of the tube.
$\mathrm{a}_{2}=$ Area of the orifice.
$\mathrm{Q}=$ Volume of air flowing through the tube.
$\mathrm{C}_{\mathrm{d}}=0.6$
[3] VELOCITY OF AIR:
$\mathrm{V}=\mathrm{Q} / \mathrm{a} \mathrm{m} / \mathrm{sec}$.
[4] REYNOLDS NUMBER:
$\mathrm{Re}=\mathrm{VD} / \mathrm{v}$.
$\mathrm{V}=$ Velocity of air
$\mathrm{D}=\mathrm{Dia}$. of the pipe.
$v=$ Kinematic viscosity of air.
[5] $\mathrm{Nu}=\mathrm{hD} / \mathrm{K}$
$\mathrm{K}=$ Thermal conductivity of air.
$\mathrm{Nu}=0.023 \times \mathrm{Re}^{0.8} \times \operatorname{Pr}^{0.4}$
Re = Reynolds Number.
$\operatorname{Pr}=$ Prandtl Number.

## d] Result:

Thus the experiment of the forced convection is conducted and heat transfer coefficient are calculated.

## Heat Transfer Co-efficient:

Experimental value $=92.782 \mathrm{~W} / \mathrm{m}^{2} \mathrm{C}$.
Theoretical value $=31.2395763 \mathrm{~W} / \mathrm{m}^{2} \mathrm{C}$.

$$
\begin{aligned}
& \mathrm{h}_{1}=9 \\
& \mathrm{~h}_{2}=19 \text {. } \\
& \mathrm{h}_{\mathrm{o}}=[19-9] \times[1000 / 1.16] \\
& =10 \times 862.069=86.20689 \mathrm{~m} \text {. } \\
& \mathrm{C}_{\mathrm{d}}=0.6 \\
& \mathrm{a}_{1}=\pi / 4 \mathrm{xd}_{1}{ }^{2} \\
& \mathrm{~d}_{1}=\text { Dia of pipe }=40 \mathrm{~mm}=0.04 \mathrm{~m} \\
& =\pi / 4 \times[0.04]^{2} \\
& \mathrm{a}_{1}=0.00125664 \mathrm{~m}^{2} \text {. } \\
& \mathrm{a}_{2}=\pi / 4 \mathrm{xd}_{2}{ }^{2} \\
& \mathrm{~d}_{2}=\text { Dia of the orifice }=20 \mathrm{~mm}=0.02 \mathrm{~m} \text {. } \\
& =\pi / 4 \times[0.02]^{2} \\
& \mathrm{a}_{2}=0.00031416 \mathrm{~m}^{2} \text {. } \\
& \mathrm{Q}=0.6 \times 0.00125664 \times 0.00031416 \times \sqrt{ } 2 \times 9.81 \times 86.20689 / \sqrt{ }[0.00125664]^{2}- \\
& {[0.00031416]^{2}} \\
& \mathrm{Q}=2.3687 \times 10^{-7} \times 411.264 / 1.216 \times 10^{-3} \\
& \mathrm{Q}=0.008006376 \mathrm{~m}^{3} / \mathrm{sec} \text {. } \\
& \mathrm{V}=\mathrm{Q} / \mathrm{a}_{1} \\
& =0.008006376 / 0.00125664 \\
& =6.3713 \mathrm{~m} / \mathrm{sec} \text {. } \\
& \mathrm{Re}=\mathrm{VD} / \mathrm{v} \text {. } \\
& v \text { - Kinematic viscosity from HMT Data book } \\
& v=0.00001696 \\
& =6.3713 \times 0.04 / 0.00001696 \\
& R e=15027 \text {. } \\
& \text { NUSSELT NUMBER } \\
& \mathrm{Nu}=\mathrm{hD} / \mathrm{K} \\
& \mathrm{Nu}=0.023 \mathrm{x}[15027]^{0.8} \mathrm{x}[0.698]^{0.33} \\
& \mathrm{Nu}=43.75290799 \text {. } \\
& \mathrm{Nu}=\mathrm{hD} / \mathrm{K} \\
& \text { K - Thermal conductivity from HMT Data book } \\
& \mathrm{K}=0.02856 \text {. } \\
& 43.75290799=\mathrm{hx} 0.04 / 0.02856 \\
& \mathrm{~h}=31.2395763 \mathrm{~W} / \mathrm{m}^{2} \mathrm{C} \text {. }
\end{aligned}
$$

## VIVA QUESTIONS

## 1. Define Convection.

Convection is a process of heat transfer that will occur between a solid surface and a fluid medium when they are at different temperatures.
2. Define Reynolds number [Re].

It is defined as the ratio of inertia force to viscous force.
Re = Inertia force / Viscous force
3. Define Prandtl number [Pr].

It is the ratio of the momentum diffusivity to the thermal diffusivity.

## $\operatorname{Pr}=$ Momentum diffusivity / Thermal diffusivity

4. Define Nusselt Number [Nu].

It is defined as the ratio of the heat flow by convection process under an unit temperature gradient to the heat flow rate by conduction under an unit temperature gradient through a stationary thickness [L] of metre.

Nusselt Number $[\mathrm{Nu}]=\mathrm{q}_{\text {conv }} / \mathrm{q}_{\text {cond }}$
5. State Newton's law of convection.

Heat transfer from the moving fluid to solid surface is given by the equation.

$$
\mathrm{Q}=\mathrm{hA}=\left[\mathrm{T}_{\mathrm{w}}-\mathrm{T}_{\infty}\right]
$$

This equation is referred to as Newton's law of cooling.
Where $h=$ Local heat transfer coefficient in $W / m^{2} K$.
$\mathrm{A}=$ Surface area in $\mathrm{m}^{2}$.
$\mathrm{T}_{\mathrm{w}}=$ Surface [or] Wall temperature in K .
$\mathrm{T}_{\infty}=$ Temperature of fluid in K .
6. What is forced convection?

If the fluid motion is artificially created by means of an external force like a blower or fan, that type of heat transfer is known as forced convection.
7. What are the dimensionless parameters used in forced convection?

1. Reynolds number [Re].
2. Nusselt number $[\mathrm{Nu}]$.
3. Prandtl number $[\mathrm{Pr}]$.

## Experiment Number: 5

Title of the Experiment: Heat transfer from pin-fin(natural \& forced convection modes)
Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine Heat transfer from pin-fin(natural \& forced convection modes)

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Pin-fin Apparatus | 1 |

## b] Apparatus Description

The heat transfer from a heated surface to the ambient surrounding is given by the relation, $\mathrm{q}=\mathrm{h} A \Delta \mathrm{~T}$. In this relation $\mathrm{h}_{\mathrm{c}}$ is the convective heat transfer coefficient, $\Delta \mathrm{T}$ is the temperature difference \& A is the area of heat transfer. To increase q , h may be increased or surface area may by increased. In some cases it is not possible to increase the value of heat transfer coefficient \& the temperature difference $\Delta \mathrm{T} \&$ thus the only alternative is to increase the surface area of heat transfer. The surface area is increased by attaching extra material in the form of rod (circular or rectangular) on the surface where we have to increase the heat transfer rate. "This extra material attached is called the extended surface or fin."The fins may be attached on a plane surface, and then they are called plane surface fins. If the fins are attached on the cylindrical surface, they are called circumferential fins. The cross section of the fin may be circular, rectangular, triangular or parabolic.
c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Connect the equipment to electric power supply. |
| 2. | Keep the thermocouple selector switch to zero position. |
| 3. | Turn the Variac (dimmerstat) clockwise and adjust the power input to the heater to <br> the desired value and switch on the blower. |
| 4. | Set the air-flow rate to any desired value by adjusting the difference in mercury <br> levels in the manometer and allow the unit to stabilize. |
| 5. | Note down the temperatures, $T_{1}$ to T T from the thermocouple selector switch. $^{\text {N. }}$ (here |
| 6. | Note down the difference in level of the manometer and repeat the experiment for <br> different power inputs to the heater. |
| 7. | Connect the equipment to electric power supply. |
| 8. | Keep the thermocouple selector switch to zero position. |

## FORMULA:

Thermal Expansion ( $\beta$ ) $=\frac{d_{o}}{d_{p}}=$

## Where;

$d_{0}=$ Diameter of the Orifice; $d_{p}=$ Diameter of the pipe
Velocity of orifice $\left(v_{o}\right)=c_{d} \sqrt{\frac{2 g h\left(\rho_{m}-\rho_{a}\right)}{\rho_{a}}} \mathrm{X}(1 / 1-\beta)$
Where;
$\rho_{\mathrm{m}}=$ density of manometric fluid $=13.6 \times 10^{3} \mathrm{~kg} / \mathrm{m}^{3}$
$\rho_{a}=$ density of air $=1.17 \mathrm{~kg} / \mathrm{m}^{3}$

Velocity at orifice $x$ cross sectional area of orifice
$\mathrm{V}_{\mathrm{a}}=$ Velocity of air in the duct $=$
Cross sectional area of duct

$d_{p}=$ diameter of pipe
$d_{0}=$ diameter of orifice
$\mathrm{W}=$ Width of the duct
B = Breadth of duct
Average surface temperature of fin is given by

```
    \(T_{1}+T_{2}+T_{3}+T_{4}+T_{5}\)
\(T_{s}=\square \quad+273.15=\quad \mathrm{K}\)
\(\mathrm{T}_{\infty}=\mathrm{T}_{6}=\) Ambient temperature \(=\quad+273.15=\quad \mathrm{K}\)
```

$T_{m}=$ Mean temperature $=T s+T_{\infty}$
2

Properties of air at $\qquad$ $\mathbf{D}^{0} \mathrm{C}$
$\mathbf{v}=\quad, \operatorname{Pr}=\quad, \mathbf{K}=$

$$
\mathbf{V}_{\mathrm{a}} \mathrm{df}_{\mathrm{f}}
$$

$\operatorname{Re}=--------\quad=$
$v$
$\mathrm{Re}=$ Reynolds number
$\mathbf{P r}=\mathbf{P r a n d t l}$ number
$\mathbf{N u}=$ Nusselt number

The relationship for Nu is
$\mathrm{Nu}=\mathrm{C} \operatorname{Re}^{\mathrm{n}} \operatorname{Pr}^{1 / 3}$

| For | $\mathrm{Re}=0.4$ to 4.0 | $\mathrm{C}=0.989$ and $\mathrm{n}=0.33$ |  |
| :---: | :---: | :---: | :---: |
|  | $\mathrm{Re}=4$ to 40 | $C=0.911$ and | $\mathrm{n}=0.385$ |
|  | $\mathrm{Re}=40$ to 4000 | $\mathrm{C}=0.683$ and $\mathrm{n}=0$. |  |
|  | $\mathbf{R e}=4000$ to 40,000 | $\mathrm{C}=0.293$ and | $\mathrm{n}=0.618$ |
|  | $\mathbf{R e}=\mathbf{4 0 , 0 0 0}$ to 400,000 | $\mathrm{C}=0.27$ and $\mathrm{n}=0.805$ |  |
|  | Nuk |  |  |

Thermal conductivity of fin material, ' K ' $=110.7 \mathrm{~W} / \mathrm{m}-\mathrm{K}$
$m=\sqrt{\frac{h P}{-----}}$
Temperature distribution is given by

| $\mathrm{T}-\mathrm{T}_{\infty}$ | Cosh m (L-x) |
| :---: | :---: |
| $\mathrm{T}_{0}-\mathrm{T}_{\infty}$ | Cosh mL |

Therefore, $\mathrm{T}=\mathrm{T}_{\infty}+\left(\mathrm{T}_{0}-\mathrm{T}_{\infty}\right) \frac{\operatorname{Cosh} \mathrm{m}(\mathrm{L}-\mathrm{x})}{\operatorname{Cosh} \mathrm{mL}}$

| Distance | Temperature from | Temperature ${ }^{\circ} \mathrm{C}$ |  |  |
| :---: | :---: | :---: | :---: | :---: |
| $\mathrm{x}, \mathrm{m}$ | Experiment ${ }^{\circ} \mathrm{C}$ | from calculation | $\mathrm{x}_{1}=0.045$ | $\mathrm{~T}_{1}=$ |
|  |  |  | $\mathrm{x}_{2}=0.075$ | $\mathrm{~T}_{2}=$ |
| $\mathrm{x}_{3}=0.105$ | $\mathrm{~T}_{3}=$ |  |  |  |
| $\mathrm{x}_{4}=0.135$ | $\mathrm{~T}_{4}=$ |  |  |  |

Effectiveness of fin $=\frac{\mathrm{PK}}{\sqrt{\mathrm{hA}}} \mathrm{xtanhmL}$

TAfficioncyoffin $=\quad \underline{\tanh m L}$
mL

## TABULATION:

|  | Heat Input |  | Pressure drop, ' $h$ ' mm of mercury, | Temperatures, ${ }^{0} \mathrm{C}$ |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | V | A |  | T ${ }_{1}$ | T ${ }_{2}$ | T3 | T4 | $\mathrm{T}_{5}$ | T6 |
| 1 | 61 | 0.27 | 5 mm | 70 | 62 | 58 | 56 | 54 | 40 |

## CALCULATION:

1. $\beta=\frac{d_{6}}{d_{k}}=\frac{0.02}{0.05}=0.4$

Thermal Expansion $\beta=0.4$
2. Velocity of ofifice :

$$
\begin{aligned}
v_{o} & =c_{d d} \sqrt{\frac{2 g h(l m-l a)}{l a}} \mathrm{X}(1 / 1-\beta) \\
& =0.62 \sqrt{\frac{2 \times 9.81 \times\left(13.6 \times 10^{3}-1017\right) \times 5 \times 10^{3}}{1.17}} \times\left(\frac{1}{1}-0.4\right) \\
v_{o} & =27.02 \mathrm{~m} / \mathrm{s}
\end{aligned}
$$

## d] Result:

Thus the heat transfer coefficient under forced convection is found out the efficiency of fin.
(i).Theoretical value of temperature of fin $=327 \mathrm{~K}$
(ii).Effectiveness of fin=0.6
(iii).Efficiency of fin $=40 \%$

## VIVA QUESTIONS

## 1. What is fin?

Fins are extended surfaces used primarily to enhance the heat transfer rate between the solid fins and an adjoining fluid
2. Define Fin effectiveness

$$
\varepsilon_{4}=\frac{\text { Fin Heat Transfer Rate }}{\text { Heat Transfer Rate Without Fin }}
$$

## 3. List out the Fin types.

1. Straight fin of uniform cross section
2. Straight fin of non-uniform cross section
3. Annular fin
4. Pin fin
5. Sketch all types of fins


## Experiment Number: 6

Title of the Experiment: Determination of stefan-boltzmann constant Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the value of Stefan boltzman constant for radiation heat transfer.
FACILITIES REQUIRED AND PROCEDURE
a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :---: | :---: |
| 1. | Stefan-Boltzmann constant Apparatus | 1 |

## b] Apparatus Description

The apparatus consists of a flanged copper hemisphere fixed on a flat nonconducting plate. A test disc made of copper is fixed to the plate. Thus the test disc is completely enclosed by the hemisphere. The outer surface of the hemisphere is enclosed in a vertical water jacket used to heat the hemisphere to a suitable constant temperature. Three $\mathrm{Cr}-\mathrm{Al}$ thermocouples are attached at four strategic places on the surface of the hemisphere to obtain the temperatures. The disc is mounted on an ebonite rod which is fitted in a hole drilled at the center of the base plate. Another $\mathrm{Cr}-\mathrm{Al}$ thermocouple is fixed to the disc to record its temperature. Fill the water in the SS water container with immersion heater kept on top of the panel.
c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Remove the test disc before starting the experiment. |
| 2. | Heat the water in the ss containers to its boiling point . |
| 3. | Allow the boiling water into the container kept at the bottom containing copper <br> bemisphere units it is full .allow sufficient time to attain thermal equilibrium which <br> is indicated by the four thermocouple provided on the hemisphere. |
| 4. | Insert the test disc fixed on the ebonite rod sleeve completwly inside and lock it. <br> Start the stop clock simultaneously . |
| 5. | Note down the temperature of the test disc at an inter val of about 15 sec for about <br> 15 to 20 minutes. |
| 6. | Remove the test disc before starting the experiment. |
| 7. | Heat the water in the ss containers to its boiling point . |
| 8. | Allow the boiling water into the container kept at the bottom containing copper <br> bemisphere units it is full .allow sufficient time to attain thermal equilibrium which <br> is indicated by the four thermocouple provided on the hemisphere. |

## FORMULA:

$\mathrm{E}_{\mathrm{b}}=\sigma \mathrm{T}^{4}$
Where, $\sigma=$ Stefan Boltzman constant $=5.6697 \times 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} \mathrm{~K}^{4}\right)$

1. Temperature of disc $\mathrm{v} / \mathrm{s}$ time to obtain the slope $(\mathrm{dT} / \mathrm{dt})$ of the line, which passes through/nearer to all points.
$\mathrm{dT} / \mathrm{dt}=$
2. Average temperature of the hemisphere

$$
\mathrm{T}_{\text {avg }}=\underline{\left(\mathrm{T}_{1}+\mathrm{T}_{2}+\mathrm{T}_{3}\right)}+273.15=\mathrm{K}
$$

3. $\mathrm{T}_{\mathrm{d}}=$ Temperature of the disc before inserting to test chamber ${ }^{\circ} \mathrm{K}$ (ambient)=
4. Rate of change of heat capacity of the disc $=\mathrm{m} \mathrm{C}_{\mathrm{p}}(\mathrm{dT} / \mathrm{dt})$

Net energy radiated on the disc $=\sigma \mathrm{A}_{\mathrm{d}}\left(\mathrm{T}^{4}{ }_{\text {avg }}-\mathrm{T}^{4}{ }_{\mathrm{d}}\right)$
Where, $\quad \begin{aligned} & \mathrm{A}_{\mathrm{d}}=\text { area of the disc }=\pi \mathrm{d}^{2} \quad \mathrm{~m}^{2} \\ & \mathrm{~d}=20 \mathrm{~mm}\end{aligned}$

$$
C_{p}=\text { specific heat of copper }=0.38 \mathrm{~kJ} / \mathrm{kg}-\mathrm{K}
$$

Rate of change of heat capacity of the disc $=$ Net energy radiated on the disc

$$
\mathrm{mC}_{\mathrm{p}}(\mathrm{dT} / \mathrm{dt})=\sigma \mathrm{A}_{\mathrm{d}}\left(\mathrm{~T}^{4}{ }_{\text {avg }}-\mathrm{T}^{4}{ }_{\mathrm{d}}\right)
$$

Thus ' $\sigma$ ' can be evaluated as shown

$$
\sigma=\frac{\mathrm{m} \mathrm{C}_{\mathrm{p}}(\mathrm{dT} / \mathrm{dt})}{\mathrm{A}_{\mathrm{d}}\left(\mathrm{~T}_{\text {avg }}^{4}-\mathrm{T}_{\mathrm{d}}^{4}\right)}=
$$

## TABULATION:

| Thermocouple | Temperature of the copper <br> hemisphere |
| :---: | :---: |
| $T_{1}$ | 35 |
| $T_{2}$ | 35 |
| $T_{3}$ | 35 |
| $T_{4}$ | 36 |

## CALCULATION:

Rate of change of heat capacity of the disc $=$ Net energy radiated on the disc

$$
\mathrm{mC}_{\mathrm{p}}(\mathrm{dT} / \mathrm{dt})=\sigma \mathrm{A}_{\mathrm{d}}\left(\mathrm{~T}^{4}{ }_{\mathrm{avg}}-\mathrm{T}_{\mathrm{d}}^{4}\right)
$$

Thus ' $\sigma$ ' can be evaluated as shown

$$
\sigma=\frac{\mathrm{m} \mathrm{C}_{\mathrm{p}}(\mathrm{dT} / \mathrm{dt})}{\mathrm{A}_{\mathrm{d}}\left(\mathrm{~T}_{\text {avg }}^{4}-\mathrm{T}_{\mathrm{d}}^{4}\right)}=4.67 \times 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} \mathrm{~K}^{4}\right)
$$

## d] Result:

Thus the Stefan Boltzman constant for radiation heat transfer is found out and draw the graph. Stefan Boltzman constant $(\sigma)=4.67 \times 10^{-8} \mathrm{~W} / \mathrm{m}^{2} \mathrm{k}^{4}$

## VIVA QUESTIONS

1. Mention Stefan boltzman contant.
$\sigma=$ Stefan Boltzman constant $=5.6697 \times 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} \mathrm{~K}^{4}\right)$
2. Define Stefan boltzman contant.

Stefan Boltzman law states that the total emissive power of a perfect black body is proportional to
fourth power of the absolute temperature of black body surface
$\mathrm{E}_{\mathrm{b}}=\sigma \mathrm{T}^{4}$
$\sigma=$ Stefan Boltzman constant $=5.6697 \times 10^{-8} \mathrm{~W} /\left(\mathrm{m}^{2} \mathrm{~K}^{4}\right)$
3. Define Emissive power $\left[E_{b}\right]$.

The emissive power is defined as the total amount of radiation emitted by a body per unit time and unit area. It is expressed in $\mathrm{W} / \mathrm{m}^{2}$.
4. Define monochromatic emissive power. [Eb ${ }_{b}$ ]

The energy emitted by the surface at a given length per unit time per unit area in all directions is known as monochromatic emissive power.
5. What is meant by absorptivity?

Absorptivity is defined as the ratio between radiation absorbed and incident radiation.
Absorptivity, $\alpha=$ Radiation absorbed / Incident radiation.

## Experiment Number: 7

## Title of the Experiment: Determination of Emissivity of a Grey Surface

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the emissivity of the test plate at any desired temperature.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Emissivity Measurement Apparatus | 1 |

## b] Apparatus Description

The experimental setup consists of two circular Al plates identical in size and is provided with heating coils at the bottom. The plates are mounted on an asbestos cement sheet and are kept in an enclosure so as to provide undisturbed natural convection surroundings.

The heat input to the heaters is varied by separate dimmerstats and is measured by a wattmeter with the help of a double pole double throw switch. The temperatures of the plates are measured by separate thermocouples which are connected diametric opposite points to get the average temperature of the places. Other thermocouples are kept in the enclosure to read the ambient temperature.

Plate 1 is blackened by a thick layer of lamp black to form the idealized black surface where as the plates 2 is the test plate whose emissivity is to be determined.

## c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :---: |
| 1. | Switch on the power supply. |
| 2. | Keep the thermocouple selector switch in first position. |
| 3. | Adjust the position of the regulator to provide desired input to heater. |
| 4. | Allow the unit to stabilize. |
| 5. | Note down the temperature indicated by temperature indicator. |
| 6. | Tabulate the readings and calculate. |
| 7. | After the experiment is over turn both the energy regulators 1 and 2 to zero position. |
| 8. | For various power input repeat the experiment. |
| FORMULA: |  |
| Emissivity $\varepsilon p=\varepsilon b\left[T b^{4}-\mathrm{Ta}^{4}\right] /\left[\mathrm{T}^{4}-\mathrm{Ta}^{4}\right]$ |  |
| Where $\varepsilon b=$ Emissivity block body Temperature [ $\varepsilon 6=1$ ] |  |
| $\mathrm{Tb}=$ Block Body Temperature in K. |  |
| $\mathrm{T}_{\mathrm{p}}=$ Polished Body Temperature in K. |  |
|  | $\mathrm{Ta}=$ Chamber Temperature in K . |

TABULATION:

| S.No. | Voltage | Current | Black Body <br> Temperature <br> $\left[{ }^{\circ} \mathbf{C}\right]$ | Polished Body <br> Temperature <br> $\left[{ }^{\circ} \mathbf{C}\right]$ | Chamber <br> Temperature <br> $\left[{ }^{\circ} \mathbf{C}\right]$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 100 | 0.4 | 80 |  |  |

## CALCULATION:

[1] Black body Temperature $[\mathrm{Tb}]=80^{\circ}+273=353 \mathrm{~K}$.
[2] Polished body Temperature $[\mathrm{T} \mathrm{p}]=90^{\circ}+273=363 \mathrm{~K}$.
[3] Chamber Temperature [Ta] $\quad=40^{\circ}+273=313 \mathrm{~K}$.

## EMISSIVITY:

$$
\begin{array}{rlr}
\varepsilon p & =\varepsilon b\left[\mathrm{~Tb}^{4}-\mathrm{Ta}^{4}\right] /\left[\mathrm{Tp}^{4}-\mathrm{Ta}^{4}\right] \quad[\varepsilon b=1] \\
& =1 \times\left[353^{4}-313^{4}\right] /\left[363^{4}-313^{4}\right] \\
& =\left[1.55 \times 10^{10}-9.59 \times 10^{9}\right] /\left[1.736 \times 10^{10}-9.59 \times 10^{9}\right] \\
\varepsilon p & =0.7626 .
\end{array}
$$

## d] Result:

Thus the Emissivity of the test plate is determined.
Emissivity $\varepsilon p=0.7626$.

## VIVA QUESTIONS

## 6. Define Radiation

The heat transfer from one body to another without any transmitting medium is known as radiation. It is an electromagnetic wave phenomenon.
7. Define Emissivity.

It is defined as the ability of the surface of a body to radiate heat. It is also defined as the ratio of emissive power of any body to the emissive power of a black body of equal temperature.

Emissivity, $\varepsilon=\mathrm{E} / \mathrm{E}_{\mathrm{b}}$.
8. Define Emissive power $\left[\mathrm{E}_{\mathrm{b}}\right]$.

The emissive power is defined as the total amount of radiation emitted by a body per unit time and unit area. It is expressed in $\mathrm{W} / \mathrm{m}^{2}$.
9. Define monochromatic emissive power. [ $\mathrm{E}_{\mathrm{b} \lambda}$ ]

The energy emitted by the surface at a given length per unit time per unit area in all directions is known as monochromatic emissive power.
10. What is meant by absorptivity?

Absorptivity is defined as the ratio between radiation absorbed and incident radiation.

Absorptivity, $\alpha=$ Radiation absorbed / Incident radiation.
11. What is meant by reflectivity?

Reflectivity is defined as the ratio of radiation reflected to the incident radiation.

Reflectivity, $\rho=$ Radiation reflected / Incident radiation.

## 12. What is meant by transmissivity?

Transmissivity is defined as the ratio of radiation transmitted to the incident radiation.

Transmissivity, $\tau=$ Radiation transmitted / Incident radiation.

## 13. What is black body?

Black body is an ideal surface having the following properties.

1. A black body absorbs all incident radiation, regardless of wav e length and direction.
2. For a prescribed temperature and wave length, no surface can emit more energy than black body.
3. What is meant by gray body?

If a body absorbs a definite percentage of incident radiation irrespective of their wave length, the body is known as gray body. The emissive power of a gray body is always less than that of the black body.

PARALLEL FLOW SIDE

| HOT WATER SIDE |  |  | COLD WATER SIDE |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Flow rate [kg/s] | $\begin{gathered} \mathrm{T}_{\mathrm{hi}} \\ {\left[{ }^{0} \mathrm{C}\right]} \end{gathered}$ | $\begin{gathered} \hline \mathrm{T}_{\mathrm{ho}} \\ {\left[{ }^{0} \mathrm{C}\right]} \end{gathered}$ | Flow rate [kg/s] | $\begin{gathered} \mathrm{T}_{\mathrm{ci}} \\ {\left[{ }^{0} \mathrm{C}\right]} \end{gathered}$ | $\begin{gathered} \mathrm{T}_{\mathrm{co}} \\ {\left[{ }^{0} \mathrm{C}\right]} \end{gathered}$ |
| $400 \mathrm{ml} / 10 \mathrm{sec}$. | 56 | 45 | $400 \mathrm{ml} / 12 \mathrm{sec}$ | 34 | 39 |

## COUNTER FLOW SIDE

| HOT WATER SIDE |  |  | COLD WATER SIDE |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Flow rate | $\mathrm{T}_{\mathrm{hi}}$ | $\mathrm{T}_{\mathrm{ho}}$ | Flow rate | $\mathrm{T}_{\mathrm{ci}}$ | $\mathrm{T}_{\mathrm{c}}$ |
| $[\mathrm{kg} / \mathrm{s}]$ | $\left[{ }^{0} \mathrm{C}\right]$ | $\left[{ }^{0} \mathrm{C}\right]$ | $[\mathrm{kg} / \mathrm{s}]$ | $\left[{ }^{\circ} \mathrm{C}\right]$ | $\left[{ }^{0} \mathrm{C}\right]$ |
| $800 \mathrm{ml} / 10 \mathrm{sec}$. | 82 | 57 | $800 \mathrm{ml} / 15 \mathrm{sec}$ | 35 | 51 |

## CALCULATION:

## PARALLEL FLOW:

$$
\begin{aligned}
& \text { LMTD }= {\left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ci}}\right]-\left[\mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{co}}\right] / \ln \left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{c}} / \mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{co}}\right] } \\
& \mathrm{T}_{\mathrm{ci}}=\text { Entry temperature of cold fluid }\left[{ }^{0} \mathrm{C}\right] . \\
& \mathrm{T}_{\mathrm{co}}=\text { Exit temperature of cold fluid }\left[{ }^{0} \mathrm{C}\right] . \\
& \mathrm{T}_{\mathrm{hi}}=\text { Entry temperature of hot fluid }\left[{ }^{0} \mathrm{C}\right] . \\
& \mathrm{T}_{\mathrm{ho}}=\text { Exit temperature of hot fluid }\left[{ }^{0} \mathrm{C}\right] . \\
&=[329-307]-[318-312] / \ln [(329-307) /(318-312)] \\
&=12.31 \mathrm{~K} .
\end{aligned}
$$

Mass flow rate of hot water $\mathrm{m}_{\mathrm{h}}=400 / 10 \times 10^{-6} \times 1000=400 \times 10^{-4} \mathrm{Kg} / \mathrm{s}$.
Mass flow rate of cold water $\mathrm{m}_{\mathrm{c}}=400 / 12 \times 10^{-6} \times 1000=333.3 \times 10^{-4} \mathrm{Kg} / \mathrm{s}$.

$$
\begin{aligned}
\mathrm{Q}_{\mathrm{h}} & =\mathrm{m}_{\mathrm{h}} \times \mathrm{C}_{\mathrm{ph}}\left[\mathrm{~T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ho}}\right] \\
& =400 \times 10^{-4} \times 4.187 \times[329-318] \\
\mathrm{Q}_{\mathrm{h}} & =1.842 \mathrm{KJ} / \mathrm{sec} . \\
\mathrm{Q}_{\mathrm{c}} & =\mathrm{m}_{\mathrm{c}} \times \mathrm{cc}_{\mathrm{pc}}\left[\mathrm{~T}_{\mathrm{co}}-\mathrm{T}_{\mathrm{ci}}\right] \\
& =333.3 \times 10^{-4} \times 4.187[312-307] \\
\mathrm{Q}_{\mathrm{c}} & =0.698 \mathrm{KJ} / \mathrm{sec} . \\
\mathrm{Q}_{\mathrm{act}} & =\left[\mathrm{Q}_{\mathrm{h}}+\mathrm{Q}_{\mathrm{c}}\right] / 2=[1.842+0.698] / 2 \\
\mathrm{Q}_{\mathrm{act}} & =1.27 \mathrm{KJ} / \mathrm{sec} .
\end{aligned}
$$

## Overall heat transfer co-efficient

$$
\begin{aligned}
\mathrm{A} & =\pi \times \mathrm{D} \times \mathrm{L} \\
& =\pi \times 0.013 \times 1.5 \\
& =0.06123 \mathrm{~m}^{2} . \\
\mathrm{U} & =\mathrm{Q}_{\text {act }} / \mathrm{A} \times \mathrm{LMTD} \\
& =1.27 / 0.06123 \times 12.31 \\
& =1.685 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K} . \\
\mathrm{U} & =\text { Overall heat transfer co-efficient. }
\end{aligned}
$$

## Experiment Number: 8

## Title of the Experiment: Effectiveness Of Parallel / Counter Flow Heat Exchanger

Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the overall heat transfer co-efficient on the given double pipe parallel flow and counter flow heat exchanger.

FACILITIES REQUIRED AND PROCEDURE
a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Parallel/Counter flow heat exchanger <br> apparatus. | 1 |

b] Theory:
A heat exchanger is defined as equipment which transfers the heat from a hot fluid to a cold fluid.

## Types of Heat Exchanger

There are several types of heat exchangers which may be classified on the basis of
I. Nature of heat exchange process
II. Relative direction of fluid motion
III. Design and constructional features
IV. Physical state of fluids.

## I. Nature of heat exchange process

On the basis of the nature of heat exchange process, heat exchangers are classified as
a] Direct contact heat exchangers or Open heat exchangers
b] Indirect contact heat exchangers.

## a] Direct contact heat exchangers or Open heat exchangers

In direct contact heat exchanger, the heat exchange takes place by direct mixing of hot and cold fluids. This heat transfer is usually accompanied by mass transfer.

Examples: Cooling towers, direct contact feed heaters.

## b. Indirect contact heat exchangers

In this type of heat exchangers, the transfer of heat between two fluids could be carried out by transmission through a wall which separates the two fluids.

It may be classified as
[i] Regenerators
[ii] Recuperators [or] Surface heat exchangers.

## EFFICTIVENESS:

$$
\begin{aligned}
& \varepsilon= {\left[1-\exp \left(-U_{A} / \mathrm{C}_{\min }\left\{1+\mathrm{C}_{\min } / \mathrm{C}_{\max }\right)\right)\right] / 1+\left[\mathrm{C}_{\min } / \mathrm{C}_{\max }\right] } \\
& \mathrm{C}_{\mathrm{h}}=\mathrm{C}_{\mathrm{pn}} \times \mathrm{m}_{\mathrm{h}}=4.187 \times 400 \times 10^{-4}=0.167=\mathrm{C}_{\max } \\
& \quad \mathrm{C}_{\mathrm{c}}=\mathrm{C}_{\mathrm{pc}} \times \mathrm{m}_{\mathrm{c}}=4.187 \times 333.3 \times 10^{-4}=0.140=\mathrm{C}_{\min } \\
&=[1-\exp (-(1.685) \times(0.6123))] / 0.140 \times[1+0.663]] / 1+[0.839]
\end{aligned}
$$

Effictiveness $=\varepsilon=0.52$.

## COUNTER FLOW:

$$
\begin{aligned}
\mathrm{LMTD} & =\left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ci}}\right] \quad-\left[\mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{co}}\right] / \text { in }\left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ci}} / \mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{co}}\right] \\
& =[355-324]-[330-308] / \text { in }[355-324 / 330-308] \\
& =26.24 \mathrm{~K} . \\
\mathrm{Q}_{\mathrm{h}} & =\mathrm{m}_{\mathrm{ch}} \mathrm{C}_{\mathrm{ph}}\left[\mathrm{~T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ho}}\right] \\
& =0.08 \times 4.187[355-330] \\
\mathrm{Q}_{\mathrm{h}} & =8.374 \mathrm{KJ} / \mathrm{sec} . \\
\mathrm{Q}_{\mathrm{c}} & =\mathrm{M}_{\mathrm{c}} \mathrm{C}_{\mathrm{pc}}\left[\mathrm{~T}_{\mathrm{co}}-\mathrm{T}_{\mathrm{ci}}\right] \\
& =0.053 \times 4.187[324-308] \\
\mathrm{Q}_{\mathrm{c}} & =3.551 \mathrm{KJ} / \mathrm{sec} . \\
\text { Qact } & =8.374+3.551 / 2 \\
& =5.962 \mathrm{KJ} / \mathrm{sec} .
\end{aligned}
$$

## Overall heat transfer co-efficient

$$
\begin{aligned}
& \mathrm{U}=\mathrm{Q}_{\text {aee }} / \mathrm{A} \times \mathrm{LMTD} \\
& \mathrm{U}=\text { Overall heat transfer co-efficient }\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right] \\
& \mathrm{A}
\end{aligned}=\pi \mathrm{DL}=3.14 \times 0.013 \times 1.5 \mathrm{x} .
$$

## EFFICTIVENESS

$$
\begin{aligned}
& \varepsilon=1-\exp \left[-\mathrm{U}_{\mathrm{A}} / \mathrm{C}_{\min }\left[1+\mathrm{C}_{\min } / \mathrm{C}_{\max }\right]\right] / 1+\left[\mathrm{C}_{\min } / \mathrm{C}_{\max }\right] \\
& \varepsilon=[1-\exp [-3.543 \times 0.06123 / 0.222][1+0.663]] / 0.663+1 \\
& \mathrm{C}_{\mathrm{c}}=\mathrm{M}_{\mathrm{c}} \mathrm{C}_{\mathrm{pc}}=0.053 \times 4.187 \\
& =0.222 \mathrm{KJ} / \mathrm{sec} \text {. } \\
& \mathrm{C}_{\mathrm{h}}=\mathrm{M}_{\mathrm{h}} \mathrm{C}_{\mathrm{ph}}=0.68 \times 4.187 \\
& =0.333 \mathrm{KJ} / \mathrm{sec} \text {. } \\
& \mathrm{U}=\text { Overall heat transfer co-efficient }\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right] \\
& \mathrm{A}=\mathrm{Area}=\mathrm{M}^{2} \\
& \varepsilon=0.62 \%
\end{aligned}
$$

## [i] Regenerators

In this type of heat exchangers, hot and cold fluids flow alternately through the same space.

Examples: IC engines, gas turbines.
[ii] Recuperators [or] Surface heat exchangers
This is the most common type of heat exchanger in which the hot and cold fluid do not come into direct contact with each other but are separated by a tube wall or a surface.

Examples: Automobile radiators, Air pre heaters, Economisers etc.

## Advantages

1. Easy construction
2. More economical
3. More surface area for heat transfer.

## Disadvantages

1. Less heat transfer co-efficient
2. Less generating capacity.

## II. Relative direction of fluid motion

This type of heat exchangers are classified as follows
a] Parallel flow heat exchanger
b] Counter flow heat exchanger
c] Cross flow heat exchanger.
a] Parallel flow heat exchanger
In this type, hot and cold fluids move in the same direction.
b] Counter flow heat exchanger
In this type, hot and cold fluids move in parallel but opposite directions.

## c] Cross flow heat exchanger

In this type, the hot and cold fluids move at right angles to each other.
III Design and constructional features
On the basis of design and constructional features, the heat exchangers are classified as follows.
a] Concentric tubes
b] Shell and tube
c] Multiple shell and tube passes
d] Compact heat exchangers.

## a] Concentric tubes

In this type, two concentric pipes, each carrying one of the fluids are used as a heat exchanger. The direction of flow may be parallel or counter.
b] Shell and tube
In this type of heat exchanger, one or the fluids move through a bundle of tubes enclosed by a shell. The other fluid is forced through the shell and it moves over the outside surface of the tubes.

## c] Multiple shell and tube passes

In order to increase the over all heat transfer, multiple shell and tube passes are used. In this type, the two fluids traverse the exchanger more than one time. This type of exchanger is preferred due to its low cost of manufacture, and easy to repair.

## d] Compact heat exchangers

There are many special purpose heat exchangers called compact heat exchangers. They are generally employed when convective heat transfer co-efficient associated with one of the fluids is much smaller than that associated with the other fluid.

## IV Physical state of fluids

Based on the physical state of fluids inside the exchanger, heat exchangers are classified as
a] Condensers
b] Evaporators.

## a] Condensers

In a condenser, the condensing fluid remains at constant temperature throughout the exchanger while the temperature of the colder fluid gradually increased from inlet to outlet.

In other words, the hot fluid loses latent heat which is accepted by the cold fluid.

## b] Evaporators

In a evaporator, the cold fluid remains at constant temperature while the temperature of hot fluid gradually decreases from inlet to outlet.

## APPARATUS DESCRIPTION

Apparatus consists of the constrict tube heat exchanger. The hot fluid that is hot water is obtained from an electric geyser and it flows through the outer tube. The cold fluid that is cold water can be admitted at one of the ends enabling the heat exchanger to run as parallel flow apparatus [or] a counter flow apparatus. This can be done by operating the different valves can be provided.

Temperature of the fluid can be measured using thermometer. Flow rate can be measured using stopwatch and measuring clock. The outer tube is provided with adequate asbestos rope insulation to minimize the heat loss to the surroundings.

## FORMULA USED:

Heat Transfer rate ' $q$ ' is calculated
$\mathrm{Q}_{\mathrm{h}}=$ Heat transfer rate from hot water.
$\mathrm{Q}_{\mathrm{h}}=\mathrm{m}_{\mathrm{h}} \mathrm{X} \mathrm{C}_{\mathrm{ph}}\left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ho}}\right]$
Where
$\mathrm{m}_{\mathrm{h}}=$ Mass flow rate of hot water $[\mathrm{Kg} / \mathrm{s}]$
$\mathrm{C}_{\mathrm{ph}}=$ Specific heat of hot water $[\mathrm{KJ} / \mathrm{KgK}]$
$\mathrm{T}_{\mathrm{hi}}=$ Hot water inlet temperature $\left[{ }^{0} \mathrm{C}\right]$
$\mathrm{T}_{\mathrm{ho}}=$ Hot water outlet temperature $\left[{ }^{0} \mathrm{C}\right]$
$\mathrm{Q}_{\mathrm{c}}=$ Heat Transfer rate to the cold water
$\mathrm{Q}_{\mathrm{c}}=\mathrm{m}_{\mathrm{c}} \times \mathrm{C}_{\mathrm{pc}}\left[\mathrm{T}_{\mathrm{co}}-\mathrm{T}_{\mathrm{ci}}\right]$
Where
$\mathrm{m}_{\mathrm{c}}=$ Mass flow rate of cold water $[\mathrm{Kg} / \mathrm{s}]$
$\mathrm{C}_{\mathrm{pc}}=$ Specific heat of cold water $[\mathrm{KJ} / \mathrm{KgK}]$
$\mathrm{T}_{\mathrm{co}}=$ Cod water outlet temperature $\left[{ }^{0} \mathrm{C}\right]$
$\mathrm{T}_{\mathrm{ci}}=$ Cold water inlet temperature $\left[{ }^{0} \mathrm{C}\right]$
$\mathrm{Q}=[\mathrm{Qh}+\mathrm{Qc}] / 2$
Specific heat of cold water and heat water $=4.187 \mathrm{KJ} / \mathrm{KgK}$.

LMTD $=$ Logarithmic Mean Temperature Difference.

## FOR PARALLEL FLOW:

$$
\begin{aligned}
\mathrm{LMTD}=[\Delta \mathrm{T}]_{\mathrm{m}} & =\left[\Delta \mathrm{T}_{1}-\Delta \mathrm{T}_{\mathrm{o}}\right] / \ln \left[\Delta \mathrm{T}_{1} / \Delta \mathrm{T}_{\mathrm{o}}\right] \\
& =\left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{ci}}\right]-\left[\mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{co}}\right] / \ln \left[\mathrm{T}_{\mathrm{hi}}-\mathrm{Tc}_{\mathrm{i}}\right] /\left[\mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{co}}\right]
\end{aligned}
$$

## FOR COUNTER FLOW:

$[\Delta \mathrm{T}]_{\mathrm{m}}=\left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{co}}\right]-\left[\mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{ci}}\right] / \ln \left[\mathrm{T}_{\mathrm{hi}}-\mathrm{T}_{\mathrm{co}}\right] /\left[\mathrm{T}_{\mathrm{ho}}-\mathrm{T}_{\mathrm{ci}}\right]$

## OVERALL HEAT TRANSFER CO-EFFICIENT:

## $\mathrm{Q}=\mathrm{UA}[\Delta \mathrm{T}]_{\mathrm{M}}$

Where

$$
\mathrm{Q}=\text { Heat transfer rate } \mathrm{W}
$$

$\mathrm{U}=$ Overall Heat transfer co-efficient $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$

$$
[\Delta T]_{\mathrm{M}}=\mathrm{LMTD}
$$

$$
\mathrm{A}=\mathrm{Area}=\pi \mathrm{dl}
$$

$$
\mathrm{d}=0.013 \mathrm{~m}
$$

$$
1=1.5 \mathrm{~m}
$$

$\mathrm{U}=\mathrm{q} / \mathrm{Ax}[\Delta \mathrm{T}]_{\mathrm{M}} \quad \mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$
$\mathrm{U}=$ Overall heat transfer co-efficient $\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}$
$\mathrm{Q}=$ Heat transfer rate W
A = Area $\mathrm{m}^{2}$.

## EFFICITIVENESS

$$
\begin{aligned}
& \varepsilon=1-\exp \left[-\mathrm{U}_{\mathrm{A}} / \mathrm{C}_{\min }\left[1+\mathrm{C}_{\text {min }} / \mathrm{C}_{\max }\right]\right] / 1+\left[\mathrm{C}_{\min } / \mathrm{C}_{\max }\right] \\
& \mathrm{C}_{\mathrm{h}}=\mathrm{C}_{\mathrm{ph}} \mathrm{X} \mathrm{~m}_{\mathrm{h}} ; \mathrm{C}_{\mathrm{c}}=\mathrm{C}_{\mathrm{pc}} \mathrm{x} \mathrm{~m}_{\mathrm{c}} \\
& \mathrm{U}=\text { Overall heat transfer co-efficient }\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right] \\
& \mathrm{A}=\text { Area }\left[\mathrm{m}^{2}\right] . \\
& \mathrm{A}=\pi \mathrm{DL}=\mathrm{D}=0.013 \mathrm{~m} \\
& \mathrm{~L}=1.5 \mathrm{~m}
\end{aligned}
$$

c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Connect the water supply at the back of the unit. The inlet water flows <br> through the geyser and outer pipe of the heat exchanger and flows through <br> the inner pipe of the heat exchanger and flows out. |
| 2. | Switch on the Geyser allow the temperature to reach steady state. |
| 3. | The constant flow rate is maintained for both hot and cold fluid. |
| 4. | Note the inlet and outlet temperature [cold and hot water]. |
| 5. | The flow rate is measured. |
| 6. | The experiment is repeated for different flow. |

## d] Result:

Thus the heat transfer experiment was conducted in a double pipe parallel flow and counter flow heat exchanger.

## PARALLEL FLOW:

LMTD $=12.31 \mathrm{~K}$
Heat Transfer $\mathrm{Q}=1.27 \mathrm{KJ} / \mathrm{sec}$.
Overall heat transfer
co-efficient $\mathrm{U}=1.685 \mathrm{~W} / \mathrm{m}^{2} \mathrm{~K}$
Effictiveness $\varepsilon=0.52=52 \%$
VIVA QUESTIONS

1. What is heat exchanger?

A heat exchanger is defined as an equipment which transfers the heat from a hot fluid to a cold fluid.
2. What is meant by Direct heat exchanger [or] open heat exchanger?

In direct contact heat exchanger, the heat exchange takes place by direct mixing of hot and cold fluids.
3. What is meant by Indirect contact heat exchanger?

In this type of heat exchangers, the transfer of heat between two fluids could be carried out by transmission through a wall which separates the two fluids.
4. What is meant by parallel flow heat exchanger?

In this type of heat exchanger, hot and cold fluids move in the same direction.
5. What is meant by counter flow heat exchanger?

In this type of heat exchanger, hot and cold fluids move in parallel but opposite directions.
6. What is meant by cross flow heat exchanger?

In this type of heat exchanger, hot and cold fluids move at right angles to each other.
7. What is meant by Shell and tube heat exchanger?

In this type of heat exchanger, one of the fluids moves through a bundle of tubes enclosed by a shell. The other fluid is forced through the shell and it moves over the outside surface of the tubes.
8. What is meant by LMTD?

We know that the temperature difference between the hot and cold fluids in the heat exchanger varies from point to point. In addition various modes of heat transfer are involved. Therefore based on concept of appropriate mean temperature difference, also called logarithmic mean temperature difference, the total heat transfer rate in the heat exchanger is expressed as
$\mathrm{Q}=\mathrm{UA}[\Delta \mathrm{T}]_{\mathrm{m}} \quad$ Where, $\mathrm{U}=$ Overall heat transfer co-efficient $\left[\mathrm{W} / \mathrm{m}^{2} \mathrm{~K}\right], \mathrm{A}=$ Area, $\mathrm{m}^{2}$ $[\Delta \mathrm{T}]_{\mathrm{m}}=$ Logarithmic mean temperature difference.
9. What is meant by Effectiveness?

The heat exchanger effectiveness is defined as the ratio of actual heat transfer to the maximum possible heat transfer.
Effectiveness $\varepsilon=$ Actual heat transfer / Maximum possible heat transfer $=\mathrm{Q} / \mathrm{Q}_{\max }$

## Experiment Number: 9

## Title of the Experiment: Determination of COP of a Refrigeration System Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the [i] Theoretical COP, [ii] Experimental COP, [iii] Carnot COP, [iv] Relative COP on a refrigeration system.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Refrigeration test rig. | 1 |

## b] Description

Vapour compression cycle is widely used refrigeration cycle. The main object of the trainer is to demonstrate refrigeration system with basic components and necessary controls. The practical working is demonstrated in the system and considerable amount of theoretical analysis and performance can be studied.

The trainer consists of components of a refrigeration system viz. Hermetically sealed components, evaporator, condenser, capillary tube. The condenser is air cooled type for which a condenser fans and motor has been provided. Evaporator is water immersion type which is housed in a thermally insulated calorimeter. Calorimeter is provided with a electric heater which can be used for heating the water initially to be desired temperature.

In addition to capillary tube a thermostatic expansion valve is also provided. We have to select either a capillary tube or thermostatic expansion valve at a time. A toggle switch has been provided to facilitate this selection.

A temperature indicator with six point selection switch has been provided to get the various temperature of Freon - 12 viz. Compressor suction, compressor discharge after condenser and after expansion and water temperature.

Special gauges have been provides for indicating Freon - 12 pressure at above mentioned points except for colorimeter water.

An energy meter has been provided which indicates the consumption of energy of compressor. An additional energy meter has been provided to indicate the energy consumption of water heater.

The students are advised to find out the saturation temperature of $\mathrm{F}-12$ after knowing the pressures at various points and based on the saturation temperatures study the working of refrigeration considering the cycle based on
[a] Reversed Carnot cycle,
[b] Simple vapour compression cycle.

## TABULATION

| $\begin{gathered} \text { S. } \\ \text { No. } \end{gathered}$ | Time <br> [s] | Energy <br> Meter <br> Reading <br> For 10 <br> Rev. in <br> sev. | Pressure |  |  |  | Temperature [ ${ }^{\mathbf{C}} \mathrm{C}$ ] |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | $\mathrm{P}_{1}$ | $\mathbf{P}_{2}$ | $\mathbf{P}_{3}$ | $\mathbf{P}_{4}$ | T1 | T2 | T3 | T4 | T5 |
| $1 .$. | 2.15 | 176 | 25 | 195 | 150 | 20 | 20 | 52 | 22 | -12 | 29.5 |
| 2. | 2.25 | 186 | 22.5 | 195 | 150 | 20 | 20 | 53 | 24 | -14 | 29 |
| 3. | 2.35 | 191 | 24 | 195 | 160 | 22 | 22 | 54 | 23 | -13 | 28 |
| 4. | 2.45 | 201 | 24 | 200 | 160 | 22 | 22 | 55 | 24 | -15 | 27 |
| 5. | 2.55 | 206 | 25 | 200 | 160 | 25 | 22 | 53 | 24 | -17 | 26 |
| 6. | 3.05 | 209 | 24 | 200 | 165 | 25 | 24 | 51 | 26 | -19 | 22 |
| 7. | 3.15 | 212 | 24 | 200 | 170 | 24 | 26 | 49 | 23 | -21 | 19 |
| 8. | 3.25 | 208 | 24 | 200 | 170 | 22 | 20 | 46 | 27 | -20 | 16 |

Quantity of water in tank: 10 kg .
Initial temperature of water: $30^{\circ} \mathrm{C}$.]

## Pressure in bar:

Convert all the pressures in [PSIG] to bar [multiply the value in PSIG by 0.06894 and add 1.013 to convert to bar abs.]
$\mathrm{P}_{1}=25 \times 0.06894+1.013=2.736$ bar.
$P_{2}=195 \times 0.06894+1.013=14.456$ bar.
$P_{3}=150 \times 0.06894+1.013=11.354$ bar.
$\mathrm{P}_{4}=20 \times 0.06894+1.013=2.391$ bar.
[1] Total Refrigerant Effect:

$$
\begin{aligned}
& \mathrm{Q}=\mathrm{mC}_{\mathrm{p}} \Delta \mathrm{~T} / \Delta \mathrm{t} . \\
& \mathrm{Q}=10 \times 4.186 \times[30-16] / 60 \times 60 \\
& \mathrm{Q}=0.1627 \mathrm{KJ} / \mathrm{sec} .
\end{aligned}
$$

[2] Theoretical COP. $=\left[h_{1}-h_{3}\right] /\left[h_{2}-h_{1}\right]$
$h_{1}$ corresponding to $P_{1}$ and $T_{1}=370 \mathrm{KJ} / \mathrm{kg}$.
$\mathrm{h}_{2}$ corresponding to $\mathrm{P}_{2}$ and $\mathrm{T}_{2}=382 \mathrm{KJ} / \mathrm{kg}$.
$h_{3}=h_{4}$ corresponding to $P_{3}$ and $T_{3}=350 \mathrm{KJ} / \mathrm{kg}$.
Where $h_{1}, h_{2}, h_{3}$ are enthalpies of refrigerant taken from p-h chart.
Theoretical COP $=[370-350] /[382-370]$
Theoretical C.O.P. $=1.667$.

The interested students can also study the saturation temperature against the actual temperatures obtained during the experimentation and thus study the actual cycle of refrigeration system.

## Specification:

[1] Compressor: Hermetically sealed compressor.
[2] Air cooled condenser.
[3] Expansion valve
[a] Capillary tube.
[b] Thermostatic Expansion valve.
[4] Evaporator.
[5] Rota meter: For liquid refrigerant flow rate.
[6] Refrigerant: Freon - 12.
[7] Energy meters for power measurement of compressor and the fans and heater.
[8] Pressure gauges - 4 Nos. [Two for H.P. and Two for L.P.]
[9] Temperature indicator.
[10] Solenoid valves.
[11] H.P. / L.P. cut out.
[12] Ammeter.
[13] Voltmeter.
[14] Thermostat.
c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Switch on the main. |
| 2. | Switch on the fan motor and then compressor motor. |
| 3. | Allow the plant to run to reach steady conditions. Take readings for every <br> 10 minutes to know the steady state. |
| 4. | Observe the readings in compressor motor energy meter. Freon flow <br> meter, pressure gauges and thermometer and record it is a tubular form. |
| 5. | Switch off the plant after experiment is over by switching off the <br> compressor motor first. Allow the fan motors to run for 10 minutes and <br> then switch off. |

## Specimen Calculations:

$\mathrm{P}_{1}=$ Pressure of the Refrigerant before the compressor.
$\mathrm{P}_{2}=$ Pressure of the Refrigerant after the compressor.
$\mathrm{P}_{3}=$ Pressure of the Refrigerant before the expansion valve.
$\mathrm{P}_{4}=$ Pressure of the Refrigerant after the expansion valve.

## Sensor Meter Reading:

$\mathrm{T}_{1}=$ Temperature of Refrigerant before compression.
$\mathrm{T}_{2}=$ Temperature of Refrigerant after compression.
$\mathrm{T}_{3}=$ Temperature of Refrigerant before evaporation.
$\mathrm{T}_{4}=$ Temperature of Refrigerant after evaporation.
[3] Experimental COP
Time for 10 rev . of energy meter, $\mathrm{t}=208 \mathrm{sec}$.

$$
\mathrm{t}=208 \mathrm{sec} .
$$

Energy consumed by the compressor

$$
\begin{aligned}
\mathrm{P} & =10 / \mathrm{t} \times 1 / 1500 \times 3600 \times 0.9 \mathrm{KW} . \\
& =10 / 208 \times 1 / 1500 \times 3600 \times 0.9 \\
\mathrm{P} & =0.104 \mathrm{KW} .
\end{aligned}
$$

Experimental COP = Actual Refrigeration effect / workdone

$$
=\mathrm{Q} / \mathrm{p}=0.1627 / 0.104
$$

Experimental COP $=1.564$.
[4] Carnot COP $=\mathrm{T}_{\mathrm{L}} /\left[\mathrm{T}_{\mathrm{H}}-\mathrm{T}_{\mathrm{L}}\right]$

$$
\begin{aligned}
& \mathrm{T}_{\mathrm{L}}=\mathrm{P}_{\min }=\left[\mathrm{P}_{1}+\mathrm{P}_{4}\right] / 2=[2.736+2.391] / 2=2.5635 \mathrm{bar} . \\
& \mathrm{T}_{\mathrm{H}}=\mathrm{P}_{\max }=\left[\mathrm{P}_{2}+\mathrm{P}_{3}\right] / 2=[14.456+2.391] / 2=12.905 \mathrm{bar} .
\end{aligned}
$$

Lowest Temperature from table.
$\mathrm{T}_{\mathrm{L}}=-12^{\circ} \mathrm{C}=261 \mathrm{~K}$. Corresponding to $\mathrm{P}_{\text {min }}$
Highest Temperature from table.

$$
\mathrm{T}_{\mathrm{H}}=56^{\circ} \mathrm{C}=329 \mathrm{~K} \quad \text { Corresponding to } \mathrm{P}_{\max }
$$

Carnot COP $=\mathrm{T}_{\mathrm{L}} / \mathrm{T}_{\mathrm{H}}-\mathrm{T}_{\mathrm{L}}$

$$
=261 /[329-261]
$$

$$
=3.84 \text {. }
$$

[5] Relative COP = Actual COP / Carnot COP
= 1.564 / 3.84
$=0.407$.

## Formula:

[1] Total Refrigerating effect $\mathrm{Q}=\mathrm{mC}_{\mathrm{p}} \Delta \mathrm{T} / \Delta \mathrm{t}$.
Where, $m=$ Mass of water in kg .
$C_{p}=$ Specific heat of water $=4.186 \mathrm{KJ} / \mathrm{kg}$.
$\Delta \mathrm{T}=$ Temperature drop in the water.
[2] Theoretical COP $=\left[h_{1}-h_{3}\right] /\left[h_{2}-h_{1}\right]$
[Enthalpy is to be found out from the P-h diagram of R-12]
Where, $\mathrm{h}_{1}=$ Enthalpy corresponding to pressure $\mathrm{P}_{1}$ and refrigerant entering temperature at $\mathrm{T}_{1}{ }^{0} \mathrm{C}$.
$\mathrm{h}_{2}=$ Enthalpy corresponding to pressure $\mathrm{P}_{2}$ and refrigerant leveling temperature at $\mathrm{T}_{2}{ }^{0} \mathrm{C}$.
$h_{4}=h_{3}=$ Enthalpy corresponding to pressure $P_{3}$ and refrigerant temperature after condensing at $\mathrm{T}_{3}{ }^{0} \mathrm{C}$.
[3] Experimental COP $=$ [Actual Refrigeration Effect/time] / Workdone.
Actual Refrigeration effect/time $=\mathrm{mC}_{\mathrm{p}} \Delta \mathrm{T} / \Delta \mathrm{t}$.
Where, $\mathrm{m}_{\mathrm{w}}=$ mass of water in kg .
$C_{p}=$ Specific heat of water $=4.186 \mathrm{KJ} / \mathrm{kg}$.
$\Delta \mathrm{T}=$ Temperature drop in the water.
Workdone = Energy consumed by the compressor motor to be found out from the energy meter.
Workdone $=(10 / t) \times(3600 / x\} \times 0.9$.
Where, $\mathrm{x}=$ Energy meter constant $=1500 \mathrm{rev} . / \mathrm{Kw}-\mathrm{hr}$.
$t=$ Time taken in sec. for 10 revolutions of energy meter reading.
Experimental COP $=\mathrm{mC}_{\mathrm{p}} \Delta \mathrm{T} / \Delta \mathrm{t} /$ workdone.
[4] Carnot COP $=\mathrm{T}_{\mathrm{L}} /\left[\mathrm{T}_{\mathrm{H}}-\mathrm{T}_{\mathrm{L}}\right]$
$\mathrm{T}_{\mathrm{L}}=\mathrm{P}_{\text {min }}=\left[\mathrm{P}_{1}+\mathrm{P}_{4}\right] / 2 ; \quad \mathrm{T}_{\mathrm{H}}=\mathrm{P}_{\text {max }}=\left[\mathrm{P}_{2}+\mathrm{P}_{3}\right] / 2 ;$
Where, $T_{L}=$ Lower temperature to be maintained in the evaporator in absolute units [ $\left.{ }^{0} \mathrm{~K}\right]$
$\mathrm{T}_{\mathrm{H}}=$ Higher temperature to be maintained in the condenser in absolute units $\left[{ }^{0} \mathrm{~K}\right]$
[5] Relative COP. = Actual COP / Carnot COP

## d] Result:

The COP of the Refrigeration system were determined and tabulated.

| Theortical COP. | Experimental <br> [Actual] COP | Carnot COP | Relative COP |
| :---: | :---: | :---: | :---: |
| 1.667 | 1.564 | 3.34 | 0.407 |

## VIVA QUESTIONS

1. Power requirement of a refrigerator is $\qquad$ .
Inversely proportional to COP.
2. In SI units, one ton of refrigeration is equal to $\qquad$ .
$210 \mathrm{~kJ} / \mathrm{min}$.
3. Define tons of refrigeration and COP.

A tonne of refrigeration is defined as the quantity of heat required to be removed from one tonne of water $[1000 \mathrm{~kg}]$ at $0^{\circ} \mathrm{C}$ to convert that into ice at $0^{0} \mathrm{C}$ in 24 hours. In actual practice,

1 tonne of refrigeration $=210 \mathrm{~kJ} / \mathrm{min}=3.5 \mathrm{~kW}$.
4. The capacity of a domestic refrigerator is in the range of $\qquad$ .
1 to 3 tonne.
5. Name four important properties of a good refrigerant.

1. Low boiling point.
2. High critical temperature \& pressure.
3. Low specific heat of liquid.
4. What is the difference between air conditioning and refrigeration?

Refrigeration is the process of providing and maintaining the temperature in space below atmospheric temperature.

Air conditioning is the process of supplying sufficient volume of clean air containing a specific amount of water vapour and maintaining the predetermined atmospheric condition with in a selected enclosure.
7. Name any four commonly used refrigerants.

1. Ammonia $\left[\mathrm{NH}_{3}\right]$.
2. Carbon dioxide $\left[\mathrm{CO}_{2}\right]$.
3. Sulphur di oxide $\left[\mathrm{SO}_{2}\right]$.
4. Freon - 12.
5. What are the advantages and disadvantages of air refrigeration system? Advantages:
6. The refrigerant used namely air is cheap and easily available.
7. There is no danger of fire or toxic effects due to leakages.
8. The weight to tonne of refrigeration ratio is less as compared to other systems.

## Disadvantages:

1. The quantity of refrigerant used per tonne of refrigeration is high as compared to
other system.
2. The COP of the system is very low. Therefore running cost is high.
3. The danger of frosting at the expander valves is more as the air contains moisture content.
4. What is net refrigerating effect of the refrigerant?

Refrigerating effect is the total heat removed from the refrigerant in the evaporator.

COP $=$ Refrigeration effect / Work done.
Refrigeration effect $=$ COP x Work done.

## 10. Define refrigerant.

Any substance capable of absorbing heat from another required substance can be used as refrigerant.

TABULATION:

| $\begin{gathered} \text { S. } \\ \text { No. } \end{gathered}$ | Ammeter <br> Reading <br> In <br> [Amps] | Volt meter <br> Reading <br> In Volts | Pressure Reading in |  |  |  | Temperature <br> $\left[{ }^{0} \mathrm{C}\right]$ |  |  |  | Mano <br> Meter <br> Reading <br> In [mm] <br> [ $\left.\mathrm{h}_{2}-\mathrm{h}_{1}\right]$ | Before <br> Evapo- <br> -ration <br> Inlet |  | After Evapor--ation Outlet |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | $\mathrm{P}_{1}$ | $\mathrm{P}_{2}$ | $\mathrm{P}_{3}$ | $\mathrm{P}_{4}$ | $\mathrm{T}_{1}$ | $\mathrm{T}_{2}$ | $\mathrm{T}_{3}$ | $\mathrm{T}_{4}$ |  | $\begin{aligned} & \hline \mathrm{DBT} \\ & {\left[{ }^{\circ} \mathrm{C}\right]} \end{aligned}$ | $\begin{aligned} & \hline \text { WBT } \\ & {\left[{ }^{\circ} \mathrm{C}\right]} \end{aligned}$ | $\begin{aligned} & \hline \text { DBT } \\ & {\left[{ }^{\circ} \mathrm{C}\right]} \end{aligned}$ | $\begin{aligned} & \hline \text { WBT } \\ & {\left[{ }^{\circ} \mathrm{C}\right]} \\ & \hline \end{aligned}$ |
| 1. | 5 | 200 | 55 | 280 | 265 | 65 | 28 | 120 | 110 | 36 | 4 | 38 | 27 | 32 | 29 |
| 2. | 6 | 210 | 55 | 280 | 265 | 65 | 30 | 120 | 120 | 38 | 3.5 | 37 | 27 | 31 | 29 |
| 3. | 7 | 190 | 55 | 280 | 265 | 65 | 30 | 120 | 120 | 38 | 4 | 37 | 27 | 31 | 30 |
| 4. | 7 | 190 | 55 | 285 | 265 | 67.5 | 30 | 120 | 120 | 40 | 3.5 | 37 | 28 | 31 | 25 |

## EVAPORATOR:

Length of the grill $[\mathrm{L}]=14.8 \mathrm{Cm}=0.148 \mathrm{~m}$.
Height of the grill $[\mathrm{H}]=14.3 \mathrm{Cm}=0.143 \mathrm{~m}$.

$$
\begin{aligned}
\text { Area }[\mathrm{A}] & =\mathrm{L} \times \mathrm{H} \\
& =0.148 \times 0.143 \\
\mathrm{~A} & =0.0212 \mathrm{~m}^{2} .
\end{aligned}
$$

## MODEL CALCULATION:

AMBIENT AIR:
DBT of the ambient air $\left[\mathrm{t}_{\mathrm{d}}\right]=37^{\circ} \mathrm{C}$.
WBT of the ambient air $\left[\mathrm{t}_{\mathrm{w}}\right]=28^{\circ} \mathrm{C}$.
$\mathrm{h}_{1}=90 \mathrm{KJ} / \mathrm{kg} \quad$ [from psychometric chart]

## CONDITION AIR:

DBT of conditioned air $\left[\mathrm{t}_{\mathrm{d} 1}\right]=31^{\circ} \mathrm{C}$.
WBT of conditioned air $\left[\mathrm{t}_{\mathrm{w} 1}\right]=25^{\circ} \mathrm{C}$.

$$
\begin{aligned}
& \left.\mathrm{h}_{2}=76 \mathrm{KJ} / \mathrm{kg} \quad \text { [from psychometric chart }\right] \\
& \mathrm{V}_{\mathrm{s}_{2}}=0.89 \mathrm{~m}^{3} / \mathrm{kg} .
\end{aligned}
$$

[1] Pressure head in terms of air [ $h_{a}$ ].

$$
\begin{aligned}
\rho_{\mathrm{w}} \mathrm{~h}_{\mathrm{w}}=\rho_{\mathrm{a}} \mathrm{~h}_{\mathrm{a}}=\mathrm{h}_{\mathrm{a}} & =\rho_{\mathrm{w}} \mathrm{~h}_{\mathrm{w}} / \rho_{\mathrm{a}} . \\
\text { Where } \rho_{\mathrm{w}} & =\text { Density of water }\left[1000 \mathrm{~kg} / \mathrm{m}^{3}\right] \\
\mathrm{h}_{\mathrm{w}} & =\text { Manometer reading }\left[3.5 \mathrm{~mm}=3.5 \times 10^{-3} \mathrm{~m}\right] \\
\rho_{\mathrm{a}} & =\text { Density of air. } \\
\rho_{\mathrm{a}} & =1 / \mathrm{Vs}_{2} \\
\rho_{\mathrm{a}} & =1 / 0.89=1.123 \mathrm{~kg} / \mathrm{m}^{3} . \\
\mathrm{h}_{\mathrm{a}} & =1000 \times 3.510^{-3} / 1.123 \\
\mathrm{~h}_{\mathrm{a}} & =3.117 \mathrm{~m} .
\end{aligned}
$$

Experiment Number: 10
Title of the Experiment: Experiments on Air Conditioning System Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the carnot COP, theoretical COP and capacity of the refrigeration and air conditioning system.
FACILITIES REQUIRED AND PROCEDURE
a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Air-conditioning test rig. | 1 |

## b] Introduction:

Air Conditioning for human comfort or industrial process requires certain processes to be carried out on air to vary the psychometric properties of air to requirements. These processes may involve the mixing of air streams, heating of air, cooling of the air, humidifying air, and dehumidifying air and combination of the process. All such processes are studied with the given air-condition test rig.
c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Switch on the mains. |
| 2. | Switch on the condenser, fan and blower. |
| 3. | Switch on the compressor and allow the unit to stabilize. |
| 4. | Note down the following. <br> a] Pressure $\mathrm{P}_{1}, \mathrm{P}_{2}, \mathrm{P}_{3}$ and $\mathrm{P}_{4}$ from the respective pressure gauge. <br> b] Note the corresponding Temperatures $\mathrm{T}_{1}, \mathrm{~T}_{2}, \mathrm{~T}_{3}$ and $\mathrm{T}_{4}$ at the <br> respective state points. |
| c] Monometer readings. |  |
| d] Note DBT and WBT at the inlet of the duct [before evaporation]. |  |
| e] Note DBT and WBT at the outlet of the duct [after evaporation]. |  |

## FORMULA:

DBT = Dry Bulb Temperature [ $\mathrm{T}_{\mathrm{d}}$ ]
WBT $=$ Wet Bulb Temperature $\left[\mathrm{T}_{\mathrm{w}}\right.$ ]
[1] $h_{a}=\rho_{w} h_{w} / \rho_{a}$
$\rho_{\mathrm{w}}=$ Density of water $\left[1000 \mathrm{~kg} / \mathrm{m}^{3}\right]$.
$h_{w}=$ Manometer reading.
$\rho_{a}=$ Density of air [1.123 kg/m $\left.{ }^{3}\right]$.
[2]

$$
\begin{aligned}
& \mathrm{V}_{\mathrm{a}}=\text { Velocity of air } \\
& \mathrm{V}_{\mathrm{a}}=\sqrt{ } 2 \times \mathrm{g} \times \mathrm{h}_{\mathrm{a}} \\
& \quad \mathrm{~g}=\text { acceleration due to gravity } 9.81 \mathrm{~m} / \mathrm{s}^{2} .
\end{aligned}
$$

[2] Velocity of air [ $\mathrm{V}_{\mathrm{a}}$ ]

$$
\begin{aligned}
\mathrm{V}_{\mathrm{a}} & =\sqrt{ } 2 \times \mathrm{g} \mathrm{x} \mathrm{~h}_{\mathrm{a}} \\
& =\sqrt{ } 2 \times 9.81 \times 3.117 \\
\mathrm{~V}_{\mathrm{a}} & =7.82 \mathrm{~m} / \mathrm{s} .
\end{aligned}
$$

[3] Mass of air $\left[\mathrm{m}_{\mathrm{a}}\right]=\rho_{\mathrm{a}} \times \mathrm{Ax} \mathrm{V}_{\mathrm{a}}$

$$
=1.123 \times 0.0212 \times 7.82
$$

$$
\mathrm{m}_{\mathrm{a}}=0.186 \mathrm{~kg} / \mathrm{sec}
$$

[4] Refrigeration effect $=\mathrm{m}_{\mathrm{a}}\left[\mathrm{h}_{2}-\mathrm{h}_{1}\right]$.

$$
\begin{aligned}
& =0.186[90-76] \\
& =2.604 \mathrm{KJ} / \mathrm{sec} .[\mathrm{or}] \mathrm{KW} .
\end{aligned}
$$

[5] Capacity $=$ Refrigeration effect / 3.5

$$
\begin{aligned}
& =2.604 / 3.5 \\
& =0.744 \text { tonne of refrigeration. }
\end{aligned}
$$

$$
[1 \text { tonne of refrigeration }=210 \mathrm{KJ} / \mathrm{min} .=3.5 \mathrm{KW}]
$$

[6] Carnot COP $=\mathrm{T}_{\mathrm{L}} /\left[\mathrm{T}_{\mathrm{H}}-\mathrm{T}_{\mathrm{L}}\right]$
$\mathrm{T}_{\mathrm{L}}=$ Lower temperature to be maintained in the evaporator.
$\mathrm{P}_{1}=55 \mathrm{PSI}=55 \times 0.07+1.013=4.863 \mathrm{bar}$.
$\mathrm{P}_{4}=67.5 \mathrm{PSI}=67.5 \times 0.07+1.013=5.738$ bar.
$\mathrm{P}_{\text {min }}=\left[\mathrm{P}_{1}+\mathrm{P}_{4}\right] / 2$

$$
=[4.863+5.738] / 2
$$

$$
=5.3 \text { bar. }
$$

From Table $\mathrm{R}-22 \quad \mathrm{~T}_{\mathrm{L}}=2^{\circ} \mathrm{C}=275 \mathrm{~K}$
$\mathrm{T}_{\mathrm{H}}=$ Higher temperature to be maintained in the condenser.

$$
\mathrm{P}_{2}=285 \mathrm{PSI}=285 \times 0.07+1.013=20.963 \text { bar. }
$$

$$
\mathrm{P} 3=270 \mathrm{PSI}=270 \times 0.07+1.013=19.913 \text { bar. }
$$

$$
P_{\max }=\left[P_{2}+P_{3}\right] / 2=[20.963+19.913] / 2=20.438 \text { bar. }
$$

From Table Freon $-22, \mathrm{~T}_{\mathrm{H}}=52^{\circ} \mathrm{C}=325 \mathrm{~K}$.

$$
\begin{aligned}
\text { Carnot COP } & =\mathrm{T}_{\mathrm{L}} /\left[\mathrm{T}_{\mathrm{H}}-\mathrm{T}_{\mathrm{L}}\right] \\
& =275 /[325-275] \\
& =5.5
\end{aligned}
$$

Carnot COP $=5.5$.
[7] Theoretical COP
Theoretical COP $=\left[\mathrm{h}_{1}-\mathrm{h}_{3}\right] /\left[\mathrm{h}_{2}-\mathrm{h}_{1}\right]$
[Where $h_{1}, h_{2}, h_{3}$ are enthalpies of refrigerant taken from p-h chart.]
$\mathrm{P}_{1}=4.863 \mathrm{bar} ; \mathrm{T}_{1}=1.112^{\circ} \mathrm{C} ; \mathrm{h}_{1}=260 \mathrm{KJ} / \mathrm{kg}$.
$\mathrm{P}_{2}=5.738 \mathrm{bar} ; \mathrm{T}_{2}=48.88^{\circ} \mathrm{C} ; \mathrm{h}_{2}=300 \mathrm{KJ} / \mathrm{kg}$.
$\mathrm{P}_{3}=19.913 \mathrm{bar} ; \mathrm{T}_{3}=48.88^{\circ} \mathrm{C} ; \mathrm{h}_{3}=100 \mathrm{KJ} / \mathrm{kg}$.
Theoretical COP $=[260-100] /[300-260]$
Theoretical COP $=4$.
[3] Mass of air $\mathrm{m}_{\mathrm{a}}=\rho_{\mathrm{a}} \times \mathrm{Ax} \mathrm{V}$ a
$\rho_{\mathrm{a}}=$ Density of air $\left[\mathrm{kg} / \mathrm{m}^{3}\right]$
$\mathrm{V}_{\mathrm{a}}=$ Velocity of air [m/s]
$\mathrm{A}=\mathrm{H} x \mathrm{~L}$
[4] Refrigeration effect $=\mathrm{m}_{\mathrm{a}}\left[\mathrm{h}_{2}-\mathrm{h}_{1}\right]$.
$\mathrm{h}_{2}=$ Enthalpy of ambient air [KJ/kg.]
$\mathrm{h}_{1}=$ Enthalpy of condition air [KJ/kg.]
[5] Capacity $=$ Refrigeration effect $/ 3.5$
[6] Carnot COP $=\mathrm{T}_{\mathrm{L}} /\left[\mathrm{T}_{\mathrm{H}}-\mathrm{T}_{\mathrm{L}}\right]$
$\mathrm{T}_{\mathrm{L}}=$ Lower temperature to be maintained in the evaporator in absolute unit $\left[{ }^{0} \mathrm{~K}\right]$.
$\mathrm{T}_{\mathrm{H}}=$ Higher temperature to be maintained in the condenser in absolute unit $\left[{ }^{0} \mathrm{~K}\right]$.
[7] Theoretical COP $=\left[\mathrm{h}_{1}-\mathrm{h}_{3}\right] /\left[\mathrm{h}_{2}-\mathrm{h}_{1}\right]$
$h_{1}$ corresponding to $\mathrm{P}_{1}$ and $\mathrm{T}_{1}$.
$h_{2}$ corresponding to $\mathrm{P}_{2}$ and $\mathrm{T}_{2}$.
$h_{3}$ corresponding to $\mathrm{P}_{3}$ and $\mathrm{T}_{3}$.
[Enthalpy is to be found out from the $\mathrm{P}-\mathrm{h}$ diagram of $\mathrm{R}-22$ ]

## d] Result:

Thus the experiment on the air condition system was conducted and result were tabulated.

| Carnot COP | Theoretical COP | Capacity TR |
| :---: | :---: | :---: |
| 5.5 | 4 | 0.744 |

## VIVA QUESTIONS

1. What is psychrometry?

Psychrometry is a study of properties of moist air.

## 2. Define DPT and degree of saturation.

DPT [Dew point Temperature] is the temperature to which moist air is to be cooled before it starts condensing.

Degree of saturation is the ratio of specific humidity of moist air to the specific humidity of saturated air at temperature.
3. Degree Relative Humidity [RH] and Specific humidity.

RH is the ratio of the mass of water vapour $\left[\mathrm{m}_{\mathrm{v}}\right]$ in a certain volume of moist air at a given temperature to the mass of water vapour $\left[\mathrm{m}_{\mathrm{vs}}\right]$ in the same volume of saturated air at the same temperature.
i.e., $\mathrm{RH}[\mathrm{or}] \varphi=\mathrm{m}_{\mathrm{v}} / \mathrm{m}_{\mathrm{v}}$

Specific humidity $[\omega]$ is the ratio of mass of water vapour $\left[\mathrm{m}_{\mathrm{v}}\right]$ to the mass of dry air in the given volume of mixture.
i.e., $\omega=m_{v} / m_{a}$

## 4. How are air-conditioning systems classified?

I. Based on construction of components:
[i] Unitary system,
[ii] Central system,
[iii] Package system,
[iv] Split units.
II. Based on fluid flow methods:
[i] Direct expansion [DX] system,
[ii] Chilled water [DX] system,
[iii] Chilled water air washer system.

## 5. How does humidity affect human comfort?

If the humidity is above a certain level, water vapour from human body moisture cannot be absorbed by the atmospheric air. It results in discomfort because of sweating.
6. What are the various sources of heat gain of an air-conditioned space?
[i] Solar gain through glass panes.
[ii] Solar gain through roof and walls.
[iii] Heat gain from occupants.
[iv] Heat gain from appliances and lights.
[v] Duct leakage.
[vi] Infiltration.
[vii] Vapour transmission.
7. Define bypass factor [BPF] of a coil.

The ratio of the amount of air which does not contact the cooling coil [amount of bypassing air] to the amount of supply air is called BPF.
i.e., $\mathrm{BPF}=$ Amount of air bypassing the coil / Total amount of air passed.
8. What factors affect by pass factor?

1. Pitch of fins.
2. Number of coil tubes.
3. Air velocity over the coil.
4. Direction of air flow.
5. What are the requirements of comfort $\mathrm{a} / \mathrm{c}$ ?
6. Supply of $\mathrm{O}_{2}$ and removal of $\mathrm{CO}_{2}$.
7. Removal of heat of occupants.
8. Removal of moisture of occupants.
9. Good air distribution.
10. Maintaining air purity.
11. What factors affect effective temperature?
12. Climatic and seasonal differences.
13. Clothing.
14. Age and sex.
15. Activity.
16. Stay duration.
17. Air velocity.

TABULATION:

| $\begin{gathered} \text { S. } \\ \text { No. } \end{gathered}$ | Receiver <br> Pressure <br> $\mathrm{Kgf} / \mathrm{cm}^{2}$ | Speed of the compressor [RPM] | Manometer Reading |  | $\begin{gathered} \mathrm{h}_{1}-\mathrm{h}_{2} \\ \mathrm{~cm} \end{gathered}$ | $\begin{aligned} & \mathrm{V}_{\text {actual }} \\ & \mathrm{m}^{3} / \mathrm{s} \end{aligned}$ | $\begin{gathered} \mathrm{V}_{\text {Theoretical }} \\ \mathrm{m}^{3} / \mathrm{s} \end{gathered}$ | Volumetric Efficiency $\mathrm{V}_{\text {act }} / \mathrm{V}_{\text {theo }} \mathrm{x} 100 \%$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | $\begin{gathered} \hline \mathrm{h}_{1} \\ \mathrm{~cm} \end{gathered}$ | $\begin{aligned} & \mathrm{h}_{2} \\ & \mathrm{~cm} \end{aligned}$ |  |  |  |  |
| 1 | 2 | 880 | 10.3 | 8.3 | 2 | $3.58 \times 10^{-3}$ | $5.08 \times 10^{-3}$ | 70.47\% |
| 2 | 4 | 870 | 10.2 | 8.2 | 2 | $3.58 \times 10^{-3}$ | $5.02 \times 10^{-3}$ | 71.3\% |
| 3 | 6 | 850 | 9.9 | 8 | 1.9 | $3.49 \times 10^{-3}$ | $4.90 \times 10^{-3}$ | 71.22\% |
| 4 | 8 | 840 | 9.7 | 7.8 | 1.9 | $3.49 \times 10^{-3}$ | $4.85 \times 10^{-3}$ | 71.96\% |
| 5 | 10 | 830 | 9.5 | 7.7 | 1.8 | $3.39 \times 10^{-3}$ | $4.79 \times 10^{-3}$ | 70.77\% |

## MODEL CALCULATION

## [1] ACTUAL VOLUME OF AIR:

$$
\begin{aligned}
& V_{\text {act }}=C_{d} \times A \times \sqrt{ } 2 \mathrm{gH} \mathrm{~m}^{3} / \mathrm{sec} . \\
& C_{d}=0.62 \text {. }=\text { Coefficient of discharge } . \\
& \mathrm{d}=20 \mathrm{~mm} \text {. } \\
& \mathrm{A}=\pi / 4[20 / 1000]^{2}=0.000314 \mathrm{~m}^{2} \text {. } \\
& \mathrm{g}=9.81 \mathrm{~m} / \mathrm{sec}^{2} \\
& \mathrm{~h}=\mathrm{h}_{1}-\mathrm{h}_{2}=10.3-8.3=2 \mathrm{Cm} \text {. } \\
& \mathrm{H}=\left[\mathrm{hx} \rho_{\mathrm{w}}\right] / 100 \times \rho_{\text {air }} \quad \rho_{\text {air }}=\text { Density of air } \mathrm{kg} / \mathrm{m}^{3} \text {. } \\
& =[2 \times 1000] / 100 \times 1.162] \\
& \rho_{\mathrm{w}}=\text { Density of water } \mathrm{kg} / \mathrm{m}^{3} \text {. } \\
& \rho_{\mathrm{w}}=1000 \mathrm{~kg} / \mathrm{m}^{3} \text {. } \\
& \rho_{\text {air }}=1.162 \mathrm{~kg} / \mathrm{m}^{3} \text {. } \\
& \mathrm{H}=17.21 \text {. }
\end{aligned}
$$

$$
\begin{aligned}
\mathrm{V}_{\text {act }} & =\mathrm{C}_{\mathrm{d}} \times A \times \sqrt{ } 2 \mathrm{gH} \mathrm{~m}^{3} / \mathrm{sec} . \\
& =0.62 \times 0.000314 \times \sqrt{ } \times 9.81 \times 17.21 \\
\mathrm{~V}_{\text {act }} & =3.58 \times 10^{-3} \mathrm{~m} 3 / \mathrm{sec}
\end{aligned}
$$

## [2] THEORTICAL VOLUME OF AIR:

$$
\begin{array}{rlrl}
\mathrm{V}_{\text {theo }} & =3.14 \times \mathrm{D}^{2} \times \mathrm{L} \times \mathrm{N}_{\mathrm{c}} / 4 \times 60 & & \\
& =3.14 \times[0.07]^{2} \times 0.09 \times 880 / 4 \times 60 . & & \mathrm{D}=0.07 \mathrm{~m} . \\
& & \mathrm{L}=0.09 \mathrm{~m} . \\
\mathrm{V}_{\text {theo }} & =5.08 \times 10^{-3} \mathrm{~m}^{3} / \mathrm{sec} & & \mathrm{~N}_{\mathrm{c}}=880 \mathrm{rpm} .
\end{array}
$$

## [3] VOLUMETRIC EFFICIENCY:

Volumetric Efficiency $\eta_{v}=V_{\text {act }} / V_{\text {theo }} \mathrm{x} 100$

$$
\begin{aligned}
& =\left[3.58 \times 10^{-3}\right] /\left[5.08 \times 10^{-3}\right] \times 100 \\
& =70.47 \%
\end{aligned}
$$

## Experiment Number: 11

## Title of the Experiment: Performance test on Two stage reciprocating Air Compressor

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To conduct a performance test on a two stage air compressor and determine its volumetric efficiency.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Two stage reciprocating air compressor. | 1 |

## b] Description

The air compressor is a two stage reciprocating type. The air is sucked from atmosphere and compressed in the first cylinder. The compressed air then passes through an inter cooler into the second stage cylinder, where it is further compressed. The compressed air then goes to a reservoir through a safety valve. This valve operates an electrical switch that shuts off the motor when the pressure exceeds the set limit.

The test unit consists of an air chamber containing an orifice plate and a $U$ - tube manometer; the compressor and an induction motor.

## Compressor Specification:

Diameter of low pressure piston $=70 \mathrm{~mm}$.
Diameter of high pressure piston $=50 \mathrm{~mm}$.
Stroke $=90 \mathrm{~mm}$.

## KC Compressor Details:

Model : DPS
S. No. 317

RPM 900
Induction Motor Details:

| S. No. | 1970 |
| :--- | :--- |
| KW | $: 2.2$ |
| RPM | 1440 |

## Precautions:

1. Check oil level in the compressor crank case.
2. The orifice should never be closed, test the manometer liquid [water] will be sucked into the tank.
3. At the end of the experiment the outlet valve at the air reservoir should be opened as the compressor is to be started again at low pressure to prevent undue strain on the piston.

## c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Close the outlet valve. |
| 2. | Fill up the manometer with water up to the half level. |
| 3. | Start the compressor and observe the pressure developing slowly. |
| 4. | At the particular test pressure, the outlet valve is opened slowly and <br> adjusted so that the pressure in the tank is maintained constant. |
| 5. | Observe the following readings. <br> [i] Speed of the compressor -Nc R.P.M. <br> [ii] Manometer reading $\mathrm{h}_{1}$ and $\mathrm{h}_{2} \mathrm{~cm}$ of water. <br> [iii] Pressure gauge reading P Kg/cm |

FORMULA:
Volumetric Efficiency $\eta_{\mathrm{v}}=\mathrm{V}_{\text {act }} / \mathrm{V}_{\text {theo }} \mathrm{x} 100$
$\mathrm{V}_{\text {act }}=$ Actual volume of air compressed.
$V_{\text {act }}=C_{d} \times A \times \sqrt{ } 2 \mathrm{gH} \mathrm{m}^{3} / \mathrm{sec}$.
$C_{d}=$ Co-efficient of discharge of Orifice $=0.62$.
A = Orifice Area [Dia. $=20 \mathrm{~mm}$ ]
$\mathrm{g}=9.81 \mathrm{~m} / \mathrm{sec}^{2}$.
$\mathrm{h}=$ Water head causing flow.
Theoretical Volume of air

$$
\begin{aligned}
& \mathrm{V}_{\text {theo }}= {\left[3.14 \times \mathrm{D}^{2} \times \mathrm{L}^{2} \times \mathrm{N}_{\mathrm{c}}\right] / 4 \times 60 . } \\
& \mathrm{D}=\text { Dia. Of piston }=0.07 \mathrm{~m} . \\
& \mathrm{L}=\text { Stroke length }=0.09 \mathrm{~m} . \\
& \mathrm{N}_{\mathrm{c}}=\text { RPM of the compressor. } \\
& \rho_{\mathrm{w}}=\text { Density of water } \mathrm{Kg} / \mathrm{m}^{3} . \\
& \rho_{\text {air }}= \text { Density of air } \quad \mathrm{Kg} / \mathrm{m}^{3} .
\end{aligned}
$$

## d] Result:

Thus the volumetric efficiency of the reciprocating air compressor are determined.

| S. No. | Pressure [Kgf/cm ${ }^{\mathbf{2}}$ ] | Volumetric Efficiency [ $\boldsymbol{\tau}_{\mathbf{v o l}}$ ] |
| :---: | :---: | :---: |
| 1. | 2 | $70.47 \%$ |
| 2. | 4 | $71.3 \%$ |
| 3. | 8 | $71.22 \%$ |
| 4. | 10 | $71.96 \%$ |
| 5. | $70.7 \%$ |  |

## VIVA QUESTIONS

## 1. Classify the various types of air-compressors.

1] According to the design and principle of operation
a] Reciprocating compressors.
b] Rotary compressors.
2] According to the action
a] Single acting compressors.
b] Double acting compressors.
3] According to the number of stages
a] Single stage compressors.
b] Multistage compressors.
4] According to the pressure limit
a] Low pressure compressors.
b] Medium pressure compressors.
c] High pressure compressors.
5] According to the capacity
a] Low capacity compressors [Volume delivered $0.15 \mathrm{~m}^{3} / \mathrm{s}$ or less].
b] Medium capacity compressors [Volume delivered $0.15 \mathrm{~m}^{3} / \mathrm{s}$ to $\left.5 \mathrm{~m}^{2} / \mathrm{s}\right]$.
c] High capacity compressors [Volume delivered is above $5 \mathrm{~m}^{3} / \mathrm{s}$ ].
2. What is meant by single acting compressor?

In single acting compressor, the suction, compression and delivery of air takes place on one side of the piston.
3. What is meant by double acting compressor?

In double acting reciprocating compressor, the suction, compression and delivery of air takes place on both sides of the piston.
4. What is meant by single stage compressor?

In single stage compressor, the compression of air from the initial pressure to the final pressure is carried out in one cylinder only.
5. What is meant by multistage compressor?

In multistage compressor, the compression of the air from the initial pressure to the final pressure is carried out in more than one cylinder.
6. Indicate the applications of reciprocating compressors in industry.

The applications of compressed air are as follows:

1. Pneumatic brakes.
2. Pneumatic drills.
3. Pneumatic jacks.
4. Pneumatic lifts.
5. Spray painting.
6. Shop cleaning.
7. Injecting fuel in diesel engines.
8. Supercharging internal combustion engines.
9. Refrigeration and air conditioning systems.
10. Define the terms as applied to air compressors: Volumetric efficiency and isothermal compression efficiency.

## Volumetric efficiency:

Volumetric efficiency is defined as the ratio of volume of free air sucked into the compressor per cycle to the stroke volume of the cylinder.
$\eta_{\text {Vol }}=$ Volume of free air taken per cycle / Stroke volume of the cylinder.
Isothermal compression efficiency:
Isothermal efficiency is defined as the ratio between isothermal work to the actual work of the compressor.

Isothermal efficiency, $\eta_{\text {Isothermal }}=$ Isothermal work $/$ Actual work.

## Mechanical efficiency:

Mechanical efficiency is defined as the ratio between brake power to the indicated power.

Mechanical efficiency, $\eta_{\text {mech }}=$ Brake power $/$ Indicated power.
8. Define clearance ratio.

Clearance ratio is defined as the ratio of clearance volume to swept volume [or] stroke volume.

$$
\begin{aligned}
& \mathrm{C}=\mathrm{V}_{\mathrm{c}} / \mathrm{V}_{\mathrm{s}} \\
& \mathrm{~V}_{\mathrm{c}}=\text { clearance volume } . \\
& \mathrm{V}_{\mathrm{s}}=\text { swept volume } .
\end{aligned}
$$

9. Define isentropic efficiency.

It is the ratio of the isentropic power to the brake power required to drive the compressor.

Isentropic efficiency = Isentropic power / Actual brake power.
10. What is compression ratio?

Compression ratio is defined as the ratio between total volume and clearance volume.

Compression ratio $=$ Total volume $/$ Clearance volume.

## Experiment Number: 12

## Title of the Experiment: Thermal Conductivity of insulating powder

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the Thermal Conductivity of insulating powder

## [a] Description

The apparatus consists of two thin walled concentric copper spheres. The inner sphere houses the heating coil. The insulating powder [Asbestos powder - Lagging Material] is packed between the two shells. The powder supply to the heating coil is by using a dimmer stat and is measured by Voltmeter and Ammeter. Choromel Alumel thermocouples are use to measure the temperatures. Temperature readings in turn enable to find out the Thermal Conductivity of the insulating powder as an isotropic material and the value of Thermal Conductivity can be determined.

Consider the transfer of heat by conduction through the wall of a hollow sphere formed by the insulating powdered layer packed between two thin copper spheres.

```
Let, \(\quad r_{i}=\) Radius of inner sphere in meters.
    \(r_{0}=\) Radius of outer sphere in meters.
    \(\mathrm{T}_{\mathrm{i}}=\) Average Temperature of the inner sphere in \({ }^{0} \mathrm{C}\).
    \(\mathrm{T}_{\mathrm{o}}=\) Average Temperature of the outer sphere in \({ }^{\circ} \mathrm{C}\).
Where, \(\quad \mathrm{T}_{\mathrm{i}}=[\mathrm{T} 1+\mathrm{T} 2+\mathrm{T} 3+\mathrm{T} 4] / 4\)
and \(\quad \mathrm{T}_{\mathrm{o}}=[\mathrm{T} 5+\mathrm{T} 6+\mathrm{T} 7+\mathrm{T} 8+\mathrm{T} 9+\mathrm{T} 10] / 6\)
```

Note that T1 to T10 denote the temperature of thermocouples [1] to [10].
From the experimental values of $\mathrm{q}, \mathrm{T}_{\mathrm{i}}$ and $\mathrm{T}_{\mathrm{o}}$ the unknown thermal conductivity K cal be determined as $\qquad$

$$
\mathrm{K}=\mathrm{q}\left[\mathrm{r}_{\mathrm{o}}-\mathrm{r}_{\mathrm{i}}\right] / 4 \pi \mathrm{r}_{\mathrm{i}} \times \mathrm{r}_{\mathrm{o}}\left[\mathrm{~T}_{\mathrm{i}}+\mathrm{T}_{\mathrm{o}}\right]
$$

## Specifications:

1. Radius of the inner copper sphere, $r_{i}=50 \mathrm{~mm}$.
2. Radius of the outer copper sphere, $\mathrm{r}_{\mathrm{o}}=100 \mathrm{~mm}$.
3. Voltmeter [ $0-100-200 \mathrm{~V}]$.
4. Ammeter [0-2 Amps.]
5. Temperature Indicator $0-300^{\circ} \mathrm{C}$ calibrated for chromel alumel.
6. Dimmerstat $0-2 \mathrm{~A}, 0-230 \mathrm{~V}$.
7. Heater coil - Strip Heating Element sandwiched between mica sheets -200 watts.
8. Chromel Alumel Thermocouples - No. [1] to [4] embedded on inner sphere to measure $\mathrm{T}_{\mathrm{i}}$.
9. Chromel Alumel Thermocouples - No [5] to [10] embedded on outer sphere to measure $\mathrm{T}_{\mathrm{o}}$.
10. Insulating Powder - Asbestos magnesia commercially available powder and packed between the two spheres.

## Precautions:

1. Keep dimmerstat to zero volt position before and after the experiment. Check this before switching ON the supply.
2. Handle the changeover switch of temperature indicator gently.

## b] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Start main switch on control panel. |
| 2. | Increase slowly the input to heater by the dimmerstat starting from zero <br> volt position. |
| 3. | Adjust input equal to 40 Watts Max. by Voltmeter and Ammeter. <br> Wattage W = VI |
| 4. | See that this input remains constant throughout the experiment. |
| 5. | Wait till fairly steady state condition is reached. This can be checked by <br> reading temperatures of thermocouples [1] to [10] and note changes in their <br> readings with time. |
| 6. | Note down the readings in the observations table as given below: <br> Observation Table: <br> 1. Voltmeter reading [V] = Volts. <br> 2. Ammeter reading [I] = Amps. <br> 3. Heater input [VI] = Watts. |

## INNER SPHERE:

| Thermocouple <br> No. | 1 | 2 | 3 | 4 |  |
| :--- | :---: | :---: | :---: | :---: | :--- |
|  | T 1 | T 2 | T 3 | T 4 | Mean Temp. $\mathrm{T}_{\mathrm{i}}$ <br> $\mathrm{T}_{\mathrm{i}}=[\mathrm{T} 1+\mathrm{T} 2+\mathrm{T} 3+\mathrm{T} 4] / 4$ |
| Temp. ${ }^{0} \mathrm{C}$ |  |  |  |  |  |

## OUTER SPHERE:

| Thermocouple <br> No. | 5 | 6 | 7 | 8 | 9 | 10 |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :--- |
|  | T5 | T6 | T7 | T8 | T9 | T10 | Mean Temp. $\mathrm{T}_{\mathrm{i}}$ <br> $\mathrm{T}_{\mathrm{i}}=[\mathrm{T} 5+\mathrm{T} 6+\ldots . . \mathrm{T} 10] / 6$ |
| Temp. ${ }^{0} \mathrm{C}$ |  |  |  |  |  |  |  |

## CALCULATION:

W = V x I Watts.
$\mathrm{T}_{\mathrm{i}}=$ Inner sphere mean temp. ${ }^{\circ} \mathrm{C}$.
$\mathrm{T}_{0}=$ Outer sphere mean temp. ${ }^{\circ} \mathrm{C}$.
$r_{i}=$ Radius of inner copper sphere $=50 \mathrm{~mm}$.
$r_{0}=$ Radius of outer copper sphere $=100 \mathrm{~mm}$.

## Using Equation:

$\mathrm{q}=0.86 \mathrm{~W} \mathrm{Kcal} / \mathrm{hr}$ [In MKS units]
$\mathrm{K}=0.86 \mathrm{~W}\left[\mathrm{r}_{\mathrm{o}}-\mathrm{r}_{\mathrm{i}}\right] / 4 \pi \mathrm{r}_{\mathrm{i}} \times \mathrm{r}_{\mathrm{o}}\left[\mathrm{T}_{\mathrm{i}}+\mathrm{T}_{\mathrm{o}}\right]$
$\mathrm{q}=\mathrm{V}$ x Iw/m -k [In SI units]
$\mathrm{K}=\mathrm{q}\left[\mathrm{r}_{\mathrm{o}}-\mathrm{r}_{\mathrm{i}}\right] / 4 \pi \mathrm{r}_{\mathrm{i}} \mathrm{X} \mathrm{r}_{\mathrm{o}}\left[\mathrm{T}_{\mathrm{i}}+\mathrm{T}_{\mathrm{o}}\right]$

## d] Result:

Thus Thermal Conductivity of insulating powder is determined

## VIVA QUESTIONS

1. What are the modes of heat transfer?
2. Conduction
3. Convection
4. Radiation.
5. What is conduction?

Heat conduction is a mechanism of heat transfer from a region of high temperature to a region of low temperature within a medium [solid, liquid or gases] or different medium in direct physical contact.

## 3. State Fourier's law of conduction.

The rate of heat conduction is proportional to the area measured normal to the direction of heat flow and to the temperature gradient in that direction.

Q $\alpha$ - AdT/dx
$\mathrm{Q}=-\mathrm{kA} \mathrm{dT} / \mathrm{dx}$
Where, A - Area in $\mathrm{m}^{2}$.
$\mathrm{dT} / \mathrm{dx}$ - Temperature gradient, $\mathrm{K} / \mathrm{m}$
k - Thermal conductivity, W/mK.

## 4. Define Thermal conductivity.

Thermal conductivity is defined as the ability of a substance to conduct heat.

Observation Table:
$\left.\begin{array}{|c|c|c|}\hline \text { S. No. } & \text { Mass Flow Rate in } \mathrm{Kg} / \mathrm{Min} . & \begin{array}{c}\text { Temperature in Degree } \\ \text { Centigrade }\end{array} \\ \text { T1, T2, T3, T4...................13 }\end{array}\right]$

## Experiment Number: 13

## Title of the Experiment: Thermal Conductivity of Metal Rod

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the Thermal Conductivity of Metal Rod.

## [a] Introduction:

Thermal conductivity is the physical property of the material denoting the ease with a particular substance can accomplish the transmission of thermal energy by molecular motion.

Thermal conductivity of material is found to depend on the chemical composition of the substance or substance of which it is a composed, the phase [i.e. gas, liquid or solid] in which it exists, its crystalline structure if a solid, the temperature and pressure to which it is subjected, and whether or not it is a homogeneous material.

Table 1 lists the values of thermal conductivity for some common metal

| METAL | THERMAL CONDUCTIVITY <br> $\mathrm{Kcal} / \mathrm{hr}-\mathrm{m}-{ }^{\circ} \mathrm{C}$ | STATE |
| :--- | :---: | :---: |
| SOLID'S Pure Copper | 330 | 20 degree |
| Brass | 95 | -do- |
| Steel $[0.5 \% \mathrm{C}]$ | 46 | -do- |
| S. S. | 14 | -do- |

## Mechanism of Thermal Energy Conduction in Metals:

Thermal energy may be conducted in solids by two modes:

1. Lattice Vibration
2. Transport by free electrons.

In good electrical conductors a rather large number of free electrons move about in the lattice structure of the material. Just as these electrons may transport electric charge, they may also carry thermal energy from a high temperature region to a low temperature region. In fact, these electrons are frequently referred as the electron gas. Energy may also be transmitted as vibrational energy in the lattice structure of the material. In general, however, this latter mode of energy transfer s not as large as the electrons transport and it is for this reason that good electrical conductors are almost always good heat conductor viz. Copper, Aluminium and silver. With increase in the temperature, however the increased lattice vibrations come in the way of the transport
by free electrons for most of the pure metals the thermal conductivity decreases with increase in the temperature.

## Apparatus:

The experimental set up consists of the metal bar, one end of which is heated by an electric heater while the other end of the bar projects inside the cooling water jacket. The middle portion of the bar is surrounded by a cylindrical shell filled with the asbestos insulating powder. The temperature of the bar is measured at eight different sections. While the radial temperature distribution is measured by separate thermocouples at two different sections in the insulating shell.

The heater is provided with a dimmerstat for controlling the heat input. Water under constant heat condition is circulated through the jacket and its flow rate and temperature rise are noted.

## Specification:

1. Length of the metal bar [total] : 410 mm .
2. Size of the metal bar [diameter] $: 25 \mathrm{~mm}$.
3. Test length of the bar $: 200 \mathrm{~mm}$.
4. No. of thermocouple mounted on the bar :9
5. No. of thermocouples in the insulation shell :2
6. Heater coil [Bald type] : Nichrome
7. Water jacket diameter $: 80 \mathrm{~mm}$.
8. Temperature indicator, 13 channel :200 Degree
9. Dimmerstat for heater coil $: 2 \mathrm{~A} / 230 \mathrm{~V}$.
10. Voltmeter 0 to 300 Volts.
11. Ammeter 0 to 2 Amps .
12. Measuring flash for water flow rate.
13. Stop clock.

## Theory:

The heater will heat the bar at its end and heat will be conducted through the bar to other end.

After attaining the steady state Heat flowing out of bar.
Heat flowing out of bar $=$ Heat gained by water
$\mathrm{Q}_{\mathrm{w}}=\mathrm{m}_{\mathrm{w}} \times \mathrm{Cp}_{\mathrm{w}} \times\left[\mathrm{T}_{\text {out }}-\mathrm{T}_{\text {in }}\right]=\mathrm{m}_{\mathrm{w}} \operatorname{Cp}_{\mathrm{w}}\left[\Delta \mathrm{T}_{\mathrm{w}}\right]=\mathrm{m}_{\mathrm{w}}\left[\mathrm{Cp}_{\mathrm{w}}\left[\mathrm{T}_{\text {out }}-\mathrm{T}_{\text {in }}\right]\right.$
Where, $m_{w}=$ Mass flow rate of the cooling water in $\mathrm{Kg} / \mathrm{hr}$.
$\mathrm{C}_{\mathrm{p}}=$ Specific Heat of water [Given 1]
$T=\left[T_{\text {out }}-T_{\text {in }}\right]$ for water

Thermal Conductivity of Bar

1. Heat Conducted through the Bar [Q]

$$
\mathrm{Q}=\mathrm{Q}_{\mathrm{w}}+\left\{2 \pi \mathrm{KL}\left[\mathrm{~T}_{\mathrm{o}}-\mathrm{T}_{1}\right]\right\} /\left\{\operatorname{Loge} \mathrm{e}\left[\mathrm{r}_{\mathrm{o}} / \mathrm{r}_{\mathrm{i}}\right]\right\}
$$

Where, $\mathrm{Q}_{\mathrm{w}}=$ Heat conducted through water
$\mathrm{K}=$ Thermal conductivity of Asbestos powder is $0.3 \mathrm{Kcal} / \mathrm{hr}-\mathrm{m}-$ degree
$r_{o} \& r_{i}=$ Radial distance of thermocouple in insulating shell.
2. Thermal conductivity of Bar [K]
$\mathrm{Q}=\mathrm{K}[\mathrm{dt} / \mathrm{dx}] \mathrm{x} \mathrm{A}$
Where, $\mathrm{dt}=$ Change in temperature [ $\mathrm{T} 1-\mathrm{T} 9$ ]
$\mathrm{dx}=$ Length across temperature [0.2]
A = Area of the bar $[\pi / 4 \mathrm{xd} 2]$

$$
\pi / 4 \times[0.025] 2=4.9 \times 10^{-4} \mathrm{~m}^{2}
$$

b] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Start the electric supply. |
| 2. | Adjust the temperature in the temperature indicator by means of rotating <br> the knob for compensation of temperature equal to room temperature. <br> [Normally this is per adjusted] |
| 3. | Give input to the heater by slowly rotating the dimmer stat and adjust it to <br> voltage equal to $80 \mathrm{~V}, 120 \mathrm{~V}$ etc. |
| 4. | Start the cooling water supply through the jacket and adjust it about 350cc <br> per minute. |
| 5. | Go on checking the temperature at some specified time interval say 5 <br> minute and continue this till a satisfactory steady state condition is reached. |
| 6. | Note the temperature reading 1 to 13. |
| 7. | Note the mass flow rate of water in $\mathrm{Kg} /$ minute and temperature rise in it. |

## Observations:

Mass flow rate of water [m] : Kg/min
Water inlet temperature [T12] : Degree Centigrade
Water outlet temperature [T13] : Degree Centigrade
Rod Temperature [T1 to T9] : Degree Centigrade
Radial distance of Thermocouples [ $\mathrm{r}_{\mathrm{o}}$ ]:40 mm.
Insulating shell $\quad\left[\mathrm{r}_{\mathrm{i}}\right]: 25 \mathrm{~mm}$.
Specific heat of water $[\mathrm{Cp}]: 1 \mathrm{Kcal} / \mathrm{Kg}^{0} \mathrm{~K}=4.186 \mathrm{KJ} / \mathrm{Kg} \mathrm{K}$
Thermal conductivity of Asbestos powder [K]:0.3 Kcal/hr-m- ${ }^{0} \mathrm{C}$

$$
0.3 \times 4.18 \mathrm{KJ} / \mathrm{Kg} \mathrm{~K}
$$

Length of bar [L] : 200 mm .
Demeter of bar [d] : 50 mm
Area of the bar [A] : $4.9 \times 10^{-4} \mathrm{~m}^{2}$.

Plot the temperature distribution along the length of the bar using observed values.

## CALCULATIONS:

1. Heat flowing out of bar. Q bar $=\mathrm{Q}_{\mathrm{w}}$
$\mathrm{Q}_{\mathrm{w}}=\mathrm{m} \times \mathrm{Cp} \times[\Delta \mathrm{Tw}][\mathrm{Kcal} / \mathrm{hr}]$
Where, $\mathrm{m}=$ Mass flow rate of the cooling water in $\mathrm{Kg} / \mathrm{hr}$.
$\mathrm{Cp}=$ Specific Heat of water [Given 1]
$\Delta \mathrm{Tw}=\left[\mathrm{T}_{\text {out }}-\mathrm{T}_{\text {in }}\right]$ for water
2. Heat conducted through the bar [Q]
$\mathrm{Q}=\mathrm{Q}_{\mathrm{w}}+\{2 \pi \mathrm{KL}[\mathrm{T} 10-\mathrm{T} 11]\} /\left\{\log \mathrm{e}\left[\mathrm{r}_{\mathrm{o}} / \mathrm{r}_{\mathrm{i}}\right]\right\}[\mathrm{Kcal} / \mathrm{Hr}]$
Where, $\mathrm{Qw}=$ Heat conductivity of bar [K]
$\mathrm{K}=$ Thermal conductivity of Asbestos powder is $0.3 \mathrm{Kcal} / \mathrm{hr}-\mathrm{m}$-degree $r_{0} \& r_{i}=$ Radial distance of thermocouple in insulating shell.
3. Thermal conductivity of Bar $[\mathrm{K}]$
$\mathrm{Q}=\mathrm{K}[\mathrm{dt} / \mathrm{dx}] \times \mathrm{A}\left[\mathrm{Kcal} / \mathrm{Hr}-\mathrm{m}-{ }^{0} \mathrm{C}\right]$
Where, $\mathrm{dt}=$ Change in temperature [ $\mathrm{T} 1-\mathrm{T} 9$ ]
$\mathrm{dx}=$ Length Across temperature [0.2]
A = Area of the bar [ $\mathrm{n} / 4 \mathrm{x}$ d2].
$\mathrm{n} / 4 \times[0.025] 2=4.9 \times 10^{-4} \mathrm{~m}^{2}$

## c] Result:

Thus Thermal Conductivity of Metal Rod is determined.

## VIVA QUESTIONS

1. Define heat transfer.

Heat transfer can be defined as the transmission of energy from one region to another due to temperature difference.
2. What are the modes of heat transfer?

1. Conduction
2. Convection
3. Radiation.

## 3. What is conduction?

Heat conduction is a mechanism of heat transfer from a region of high temperature to a region of low temperature within a medium [solid, liquid or gases] or different medium in direct physical contact.
4. State Fourier's law of conduction.

The rate of heat conduction is proportional to the area measured normal to the direction of heat flow and to the temperature gradient in that direction.
$\mathrm{Q} \alpha-\mathrm{AdT} / \mathrm{dx}$
$\mathrm{Q}=-\mathrm{kA} \mathrm{dT} / \mathrm{dx}$
Where, A - Area in $\mathrm{m}^{2}$.
$\mathrm{dT} / \mathrm{dx}$ - Temperature gradient, $\mathrm{K} / \mathrm{m}$
k - Thermal conductivity, W/mK.

## 5. Define Thermal conductivity.

Thermal conductivity is defined as the ability of a substance to conduct heat.

## Observation Table:

| S. No. | Description | Set - I | Set - II |
| :---: | :--- | :---: | :---: |
| 1. | Volume of water collected during test period $\left[\mathrm{m}^{3}\right]$ |  |  |
| 2. | Inlet temperature of water $\left[\mathrm{T}_{1}{ }^{\circ} \mathrm{C}\right]$ |  |  |
| 3. | Outlet temperature $\left[\mathrm{T}_{2}{ }^{\circ} \mathrm{C}\right]$ |  |  |
| 4. | Gas inlet temperature $\left[\mathrm{T}_{\text {act }}{ }^{\circ} \mathrm{C}\right]$ |  |  |
| 5. | Duration of test period $[\mathrm{t} \mathrm{sec}]$. |  |  |
| 6. | Volume of gas burnt during test period $\left[\mathrm{V}_{\mathrm{g}}\right.$ lit] |  |  |
| 7. | Barometer reading $[\mathrm{mm} \mathrm{Hg}]$ |  |  |

Experiment Number: 14
Title of the Experiment: Calorific Value Determination by Junker's Gas Calorimeter

## Date of the Experiment:

## OBJECTIVE [AIM] OF THE EXPERIMENT

To determine the higher calorific value of given gaseous fuel using gas calorimeter.

## FACILITIES REQUIRED AND PROCEDURE

a] Facilities required to do the experiment:

| Sl. No. | Facilities required | Quantity |
| :---: | :--- | :---: |
| 1. | Junker's gas calorimeter | 1 |
| 2. | LPG cylinder with pressure regulator | 1 |
| 3. | Gas flow meter | 1 |
| 4. | Measuring jar of two lit capacity | 1 |
| 5. | Thermometers $\left[0-100^{\circ} \mathrm{C}\right]$ | 3 |
| 6. | Stop watch | 1 |

## b] Theory:

The calorific value of a gaseous fuel is the total amount of energy liberated in the form of heat due to the combustion of unit volume of fuel under standard conditions. The unit of calorific value is $\mathrm{KJ} / \mathrm{m}^{3}$. The heat energy liberated by the controlled combustion of LPG fuel is transferred to the circulating cooling water in the calorimeter. Under steady state conditions, the heat energy liberated by burning the fuel is equal to the energy gained by the water. By calculating this, the calorific value of the fuel can be found out.

## c] Description:

The instrument consists of a gas-meter, a gas pressure regulator, vertical cylindrical chamber and a burner to maintain perfect combustion. The gas first passes through the gas flow meter and then through a regulator, finally the gas is burned by the Bunsen burner. The gas is burned in a burner placed underneath of the chamber and then downward through the tubes. The tubes are surrounded by continuously flowing water which enters at the bottom and leaves at constant head to ensure uniform circulation. The flowing water absorbs the heat produced by the burning gas whose calorific value is to be determined. Water which is flowing through the calorimeter is collected and weighed.

## Precaution:

1. Check the rubber hose joints at the regulator valve, gas flow meter for leak proof.
2. Keep the LPG cylinder in vertical position.
3. Ensure no gas leak.
4. Ensure proper flame in the burner.
5. Pour water in the pressure regulator till it over flows.
6. Pour water to the level marked in the flow meter.
7. Before inserting the lighted burner, ensure proper circulation of water through the calorimeter.
8. After the completion of the experiment, the water supply should be closed only after the gas supply is closed.
c] Procedure for doing the experiment:

| Step No. | Details of the Step |
| :---: | :--- |
| 1. | Level the calorimeter by adjusting the legs. |
| 2. | Insert thermometers and check all the connections to ensure that there is no <br> leakage of gas. |
| 3. | Open the water tap and circulate water through the calorimeter and ensure <br> that the water is flowing through the flow pipe uniform and continuous. |
| 4. | Open the regulator on the LPG cylinder and measure the gas flow using gas <br> flow meter. The time taken for a required volume of gas to pass the flow <br> meter is noted. |
| 5. | The air regulator on the Bunsen burner is adjusted to get a blue non- <br> luminous flame. |
| 6. | The water flow to the calorimeter is adjusted to get a temperature <br> difference of $10^{\circ} \mathrm{C}$ to $15^{\circ} \mathrm{C}$ between the inlet and outlet. |
| 7. | The water coming out of the calorimeter is collected using a measuring jar <br> and the time taken is noted. |
| 8. | Take all the readings after steady state conditions are reached. |
| 9. | Repeat the experiment by altering the quantity of water and the tabulate the <br> readings. |

## Specimen Calculation:

1. Volume of gas burnt per sec at STP is obtained from the following relation:
$\left\{\left[\mathrm{P}_{\text {act }} \times \mathrm{V}_{\text {act }}\right] / \mathrm{T}_{\text {act }}\right\}-\left\{\left[\mathrm{P}_{\text {STP }} \times \mathrm{V}_{\text {STP }}\right] / \mathrm{T}_{\text {STP }}\right.$
Where,
$\mathrm{V}_{\text {STP }}=$ Volume of gas burnt as STP $\left[\mathrm{m}^{3} / \mathrm{s}\right]$
$\mathrm{P}_{\text {act }}=$ Actual pressure of gas [mm of Hg ]
$\mathrm{P}_{\mathrm{STP}}=$ Standard atmospheric pressure $[760 \mathrm{~mm} \mathrm{Hg}]$
$\mathrm{T}_{\text {act }}=$ Actual room temperature $[\mathrm{K}]$
$\mathrm{V}_{\mathrm{act}}=\mathrm{V}_{\mathrm{g}} \times 10^{-3} / \mathrm{t}\left[\mathrm{m}^{3} / \mathrm{sec}\right]$
$\mathrm{T}_{\text {STP }}=$ Standard atmospheric temperature $\left[25^{\circ} \mathrm{C}\right]$
2. Water flow rate, $\left[m_{w}\right]=\left[\right.$ Volume of water collected in $m^{3} x$ density $] x$ Time taken $\mathrm{kg} / \mathrm{s}$
3. Higher calorific value $[\mathrm{HCV}]=\left\{\mathrm{m}_{\mathrm{w}} \times \mathrm{C}_{\mathrm{pw}} \times\left[\mathrm{T}_{2}-\mathrm{T}_{1}\right]\right\} / \mathrm{V}_{\mathrm{STP}} \mathrm{kJ} / \mathrm{m}^{3}$

Where,
$\mathrm{C}_{\mathrm{pw}}=$ Specific heat of water $=4.187 \mathrm{~kJ} / \mathrm{kg} \mathrm{K}$.
$\mathrm{T}_{2}-\mathrm{T}_{1}=$ difference in temperature between inlet and outlet, ${ }^{\circ} \mathrm{C}$.

## d] Result:

The average higher calorific value of give LPG gas at standard conditions $=---------M J / \mathrm{m}^{3}$

## VIVA QUESTIONS

1. Define Calorific Value.

The amount of heat released in one kg of fuel in one hour is called as calorific value. It is denoted by CV. It's unit is $\mathrm{kJ} / \mathrm{hr}$.
2. State the types of fuels

1. Solid fuels
2. Liquid fuels.
3. Gaseous fuels.
4. State advantages of Liquid fuels.
5. It has higher calorific value.
6. It requires lesser space in use.
7. It keeps cleanliness surroundings.
8. It eliminates wear and tear of grate.
9. It is easy to control of combustion.
10. It is easy to handle and supply.

## QUESTION BANK

## THERMAL ENGINEERING LAB II

## HEAT TRANSFER

1. Determine Thermal conductivity measurement by guarded plate method.
2. Determine Thermal conductivity of pipe insulation using lagged pipe apparatus.
3. Determine Natural convection heat transfer from a vertical cylinder.
4. Determine Forced convection inside tube.
5. Determine Heat transfer from pin-fin(natural \& forced convection modes)
6. Determination of Stefan-Boltzmann constant.
7. Determination of emissivity of a gray surface.
8. Find Effectiveness of Parallel/counter flow heat exchanger.

## REFRIGERATION AND AIR CONDITIONING

9. Determination of COP of a refrigeration system.
10. Determine COP on air-conditioning system.
11. Conduct Performance test on single/two stage reciprocating air compressor.

## LIST OF OUESTIONS BEYOND THE SYLLUBUS

12. Determine the Thermal Conductivity of insulating powder.
13. Determine the Thermal Conductivity of metal rod.
14. Determine the higher calorific value of given gaseous fuel using gas calorimeter.
